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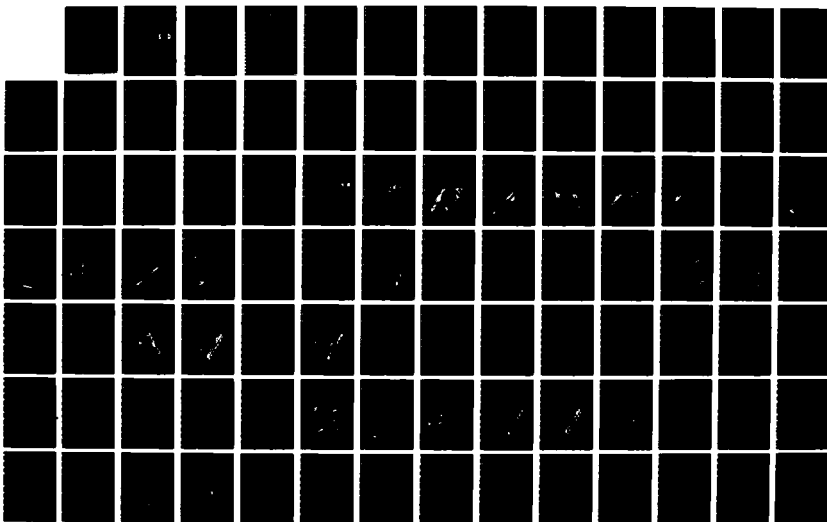
LIGHTWEIGHT TOWED HOWITZER DEMONSTRATOR PHASE 1 AND
PARTIAL PHASE 2 VOLUM (U) FMC CORP MINNEAPOLIS MINN
NORTHERN ORDNANCE DIV R RATHE ET AL APR 87
FMC-E-3041-VOL-D3-PT-2 DAAA21-86-C-0047

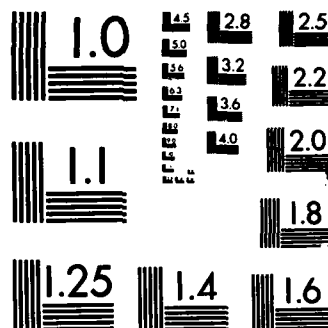
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Lightweight Towed Howitzer Demonstrator

Final Report

Volume D3 - Part II

Structural Analysis of System

April 1987

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Contract Number DAAA21-86-C-0047

FMC CORPORATION
Northern Ordnance Division
4800 East River Road
Minneapolis, Minnesota 55421

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The LTHD (Lightweight Towed Howitzer Demonstrator) was to be a 9,000 lb equivalent to the M198, transportable via Blackhawk helicopter, with reduced emplacement time using fewer personnel. The FMC design achieved weight reduction via a mortar-like configuration, composites structure, and hydraulic actuators. Recovery of power from the recoil system, in turn, facilitated crew reduction via hydraulic emplacement, four-way joystick tube lay, and power ramming. FMC completed Concept Development (Ph I) and two-thirds of Detailed Design (Ph II) prior to funds running out. <i>Keywords:</i>		

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D3/200

CEL MEMO: JANUARY 26, 1987

Interoffice

To R. Rathe* Date January 26, 1987

From C. R. Ortloff cc L. Libhardt
J. Ries
L. Langerud
B. Anderson
B. Zierwick
E. Thuse
R. Kazares
A. Amberg

Subject STRESS & STABILITY RESULTS FOR THE 22.5° GIMBAL
ROTATION, 72° GUN ELEVATION SYSTEM MODEL LWHD
UNDER PROOF IMPULSE LOADS
*(figures 497-601 attached)

This stress report summarizes results of the system FE model load case (22.5°-72°) under the assumption of "hard soil" spade emplacement, free trail ends, and proof dynamic loads. Previously, reports have been forwarded for the 0°-0° case (CRO to L. Libhardt, 17 Dec 86), the 0°-72° case (CRO to L. Libhardt, 29 Dec 86) and the 22.5°-0° case (CRO to L. Libhardt, 8 Jan 87). This report completes the cycle of specified load cases for the system model with free trail ends and hard soil spade emplacement for dynamic impulse loading. To date, 601 computer hard copy figures have been delivered on system stress results.

SUMMARY

The present load case (22.5°-72°) appears to be the limiting case of the 4 cases thus far computed. In addition to previously observed gimbal failure zones (viz., the lower cradle attachment arms and shaft attachment zones on both upper and lower box beams), this load case indicates a plastic hinge failure of the gimbal upper arms also. This load case also confirms that plastic deformation occurs on the horizontal spade plate attached to the lower box beam of the platform. The details of stress pass results at times equal to 0.084, 0.129, 0.237 and 0.304 sec. are summarized below as are stability results. Stresses shown in the accompanying plots are von Mises equivalent stress values.

DETAILS OF STRESS PASS RUNS

Shown in figures 498 to 515 are dynamic displacement-time histories for Master Degree of Freedom nodes (MDOF) located on the platform. The locations of these nodes has been shown in earlier memos of this sequence. Peak displacement times are selected from these plots for use in stress pass calculations (Figures 521 to 595). A damping value of 0.2% is incorporated into the runs; in that actual system damping values (ALPHAD, BETAD) are unknown, the decay envelope of the oscillatory displacement behavior (exhibited in Figures 498-515) past c.310 ms may vary depending on the damping constant chosen. Use of the 0.2% damping value is conservative past c.310ms in that

system stability is not dependent on an energy dissipation mechanism (which may not exist). Below c.310 ms, where loads are applied, damping plays a minor role in stress and deflection levels; note that the stress pass times are all below c.310 ms and that this time range contains all the displacement peaks.

Reference to Figures 519 and 520 shows the MDOF node locations for Figures 516-518. Node 3009 is located on the cradle forward manifold. Figure 516 indicates about a 5 inch lateral (UX) deflection peak at c.750 ms as well as a 4 inch vertical excursion at about the same time. In that the torque vector can be decomposed into components in the (global) x,y,z directions (Figure 520), then moments exist (acting on the cradle arm) to twist the cradle about the x,y, and z axes simultaneously. The beam model used for the cradle is constructed to roughly match the bending and rotational stiffness of the 10-86 cradle design so that UX, UY deflections (Figures 516, 517) roughly model the dynamic deflection effect of both cradle and cable deflection. It appears that for the 22.5°, 72° load case then that the cradle experiences reasonably large UX and UY deflections. Depending on cradle and cable damping values, it is unclear if the duration of the cradle oscillations will interfere with reloading/reaiming procedures under "rapid fire" conditions. Later cradle designs with higher stiffness components will undoubtedly reduce these deflections and vibration levels.

Figures 521-539 summarize results for the stress pass at 0.304 sec. The conclusions are summarized below: ←

- o The upper gimbal arms (Figures 527, 528, 529, 532 and 537) contain stresses far in excess of the 80 ksi yield stress. A plastic hinge appears to develop and cause arm failure. The attachment zone of upper arms to gimbal (Figure 530) also shows zones of plastic failure.
- o Upper and lower platform tabs (Figures 533-534) appear to be adequately designed.
- o The lower gimbal arms contain tip zones (Figure 535) at the 80 ksi yield stress; although the actual gimbal mount zone has not been modeled exactly, nevertheless high stresses are encountered close to the mount points.

Figures 540-558 summarize stress results for the 0.084 sec. stress pass. The conclusions are summarized below.

- o The gimbal lower box beam shaft mount openings (upper and lower) exhibit stresses in excess of yield stress (Figures 540, 541, 542). The upper gimbal box beam shaft openings are in the elastic range.

- o The platform (Figures 545-548) horizontal spade plate has zones between vertical reinforcing spacers that are in the plastic region. The moment (about the x-axis) and compressive force put on the lower platform box beam is substantial (considering that the spade horizontal plate is constrained by ground contact). This leads to stress concentration areas in the vicinity of the spade horizontal plate-lower box beam welded connection seams. As recommended in earlier memos, a possible fix to this recurring problem is to extend the triangular reinforcing tabs (front and rear) to the edge of the horizontal spade plate and/or use of several more of the extended reinforcing tabs in the open bays between the current tabs.
- o Upper (Figure 549) and lower (Figure 550) platform to gimbal shaft mounting tabs appear to be adequately designed as stresses do not exceed 30 ksi.
- o The lower gimbal arms (Figure 552) appear to have localized failure zones in the region of the cradle attachment points (here taken to be the central nodes denoted C_1 , C_2). Although the actual connecting bearing region has not been modeled, nevertheless stresses are over 80 ksi in regions away from the C_1 , C_2 nodes. Based on St. Venant's principle, the vertical plate part of the I-Beam lower arms will need to be thickened to reduce stress levels near the C_1 , C_2 nodes.
- o The gimbal lower arm reinforcing plates (Figure 554) contain stresses in the plastic regime and need to be thickened to reduce stress to acceptable levels.
- o The platform spade reinforcing spacers (Figures 555-557) appear adequate to absorb the side thrust loads as stresses shown are in the elastic range.

Figures 560-573 summarize stress results for the 0.237 sec. stress pass. The conclusions are summarized below.

- o Again, this stress pass case indicates stresses over yield on the top gimbal arms (Figure 561).
- o Stress on the lower gimbal box beam shaft opening (Figure 565) is at the yield value. This type of local failure has been observed for all the load cases and may be remedied by local reinforcement of the gimbal at all the box beam openings.
- o Stresses on the platform horizontal spade plate exceed yield stress (Figure 567-569) in the plate bays between triangular reinforcing plates by as much as 40 ksi. The overstressed zones occur both ahead (Figure 569) and behind (Figure 567) the lower platform box beam.

Figures 575-595 summarize stress results for the 0.129 sec. stress pass. The conclusions are summarized below.

- o Local yielding again occurs on the platform vertical spade plate (Figures 575-577). Stresses in the outer panel bays exceed yield by about 4 ksi.
- o Stresses in the lower gimbal box beam shaft opening (or both upper and lower surfaces) exceed yield stress (Figures 578-580) by about 6 ksi.
- o Stresses in the lower gimbal arms (Figure 583) have localized stress zones over yield stress in the web area near the gimbal-cradle attachment zone. This type of failure has been previously observed for all other stress pass times.

The values given below summarize maximum stress results for the upper and lower shafts (connecting the gimbal to the platform). These results proceed from the printed output associated with each stress pass case together with interpretative descriptive material from the ANSYS 4.2B, V. I Manual, pp. 4.4.1-4.2.5. Reference to Figure 596 (lower shaft nodes), Figure 597 (lower shaft elements), Figure 598 (upper shaft nodes) and Figure 599 (upper shaft elements) is used with printout results to obtain the maximum shaft stress values.

- o The maximum upper shaft stress (for all times considered) is 37 ksi; the maximum lower shaft stress is 32.4 ksi. These stresses are the maximum of the outer fiber stresses.
- o The maximum force in the connecting rod (element 3262) between gimbal and platform is 13,000 lbf. This rod is necessary to prevent gimbal rotation within the platform frame caused by the Y components of torque on the gimbal. Recall that the STIF4 shafts have zero torsional stiffness so that the connecting rod is necessary for gimbal stability.
- o A survey of forces on connecting nodes from trails to platform (4 connecting nodes per trail) indicates a maximum force of about 4000 lbf. The forces are obtained by observing the stress in the beam elements of the trails joining connecting nodes on the platform and multiplying by the beam area.

SUMMARY OF THE FOUR LOAD CASES

Since all four load cases have now been computed for the system model, some general observations can be made with respect to gimbal and platform stress levels. The 22.5°-72° case contains all of the previously noted structural failure mechanisms; however, the details of prior load cases should be reviewed for stress levels.

- The upper gimbal arms appear to have stresses in the plastic regime in the time period $0 < t < 400$ ms (Figure 529, 537, 561) and most likely form a plastic hinge. No allowance has been made for the apparent yield stress increase under large load rates; i.e., the 80 ksi yield stress value is used for dynamic load conditions. Note that stresses can be as high as $2.65 \times$ yield stress for the 0.304 sec. stress pass case.
- The lower gimbal arms likewise possess regions near the cradle mount zone that are over yield stress (Figures 540, 543, 552, 578). For many cases, stress levels are close to or exceed yield stress in the I-Beam web area as well as the triangular reinforcement plates connecting the arms to the gimbal box beams. Details of the maximum stress levels are obtainable from review of the totality of $0^\circ-0^\circ$, $0^\circ-72^\circ$, $22.5^\circ-0^\circ$ and $22.5^\circ-72^\circ$ load cases; this failure occurs for most of these load cases.
- The horizontal plate forming the top part of the spade is highly stressed. Since this plate lies on the ground and is constrained to move parallel to the ground, the lower platform box beam puts considerable moment and compressive force on the spade plate. The bay regions between vertical triangular reinforcement plates are in the plastic regime for most load cases. Several reinforcement ideas have been presented to strengthen this region.
- The upper and lower shaft opening zones on the gimbal, for both upper and lower box beams, are in the plastic regime. This result appears consistently in all the load cases and calls for local reinforcement of these gimbal areas.
- o The upper and lower gimbal-to-platform shafts appear to be adequate for all load cases. Stresses between 20-50 ksi are encountered for these shafts for all load cases.
- o The upper and lower platform to gimbal shaft mounting tabs appear to be adequate for all load cases as no stresses outside the elastic regime are encountered.
- o The vertical platform and gimbal box beams can be reduced in thickness somewhat (about 30%) to raise their dynamic stress levels. The weight saved can be put to use in providing local reinforcement plates or thickness increases of elements elsewhere in the structure.
- o Dynamic stability of the system exists for all load cases. The $22.5^\circ-72^\circ$ case appears to contain somewhat large horizontal and vertical amplitude displacements (about 4 inches) at low frequency (1-5 Hz) for the cradle. The high damping

characteristics of the composite slide tube and Kevlar cable may damp this motion out before the next firing event. More work needs to be done on the effect of material and geometric (structural) damping constants on system stability.

- o The spade appears adequate to absorb loads for the "hard soil" model for all load cases computed.



C. R. Ortloff

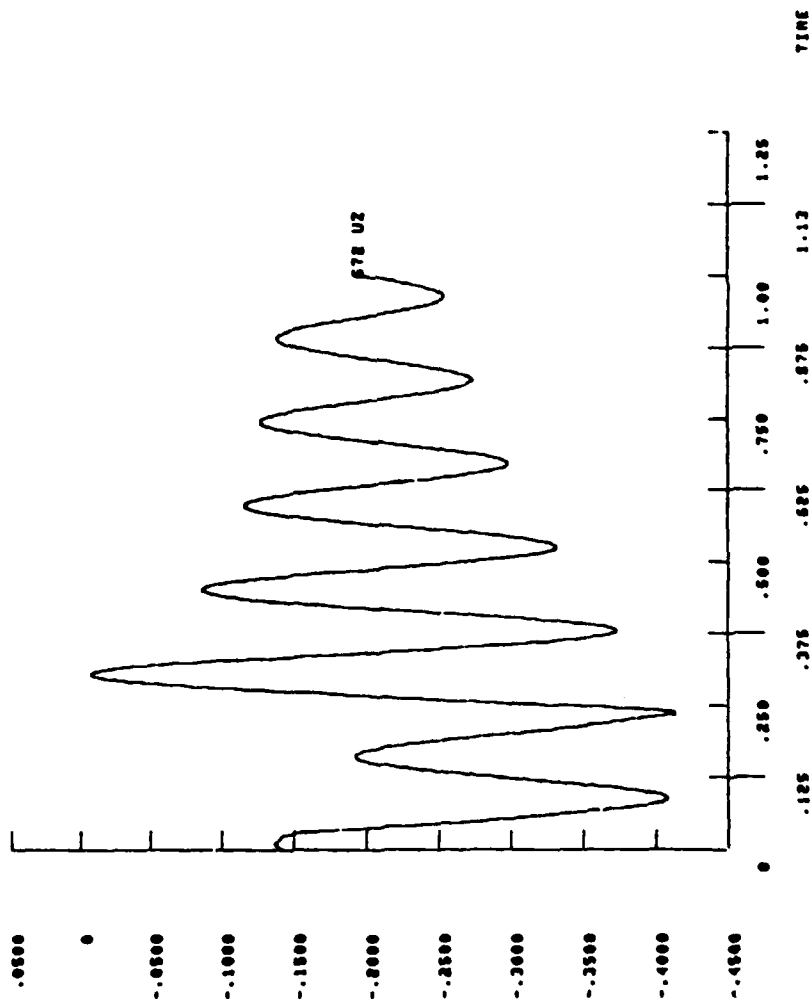
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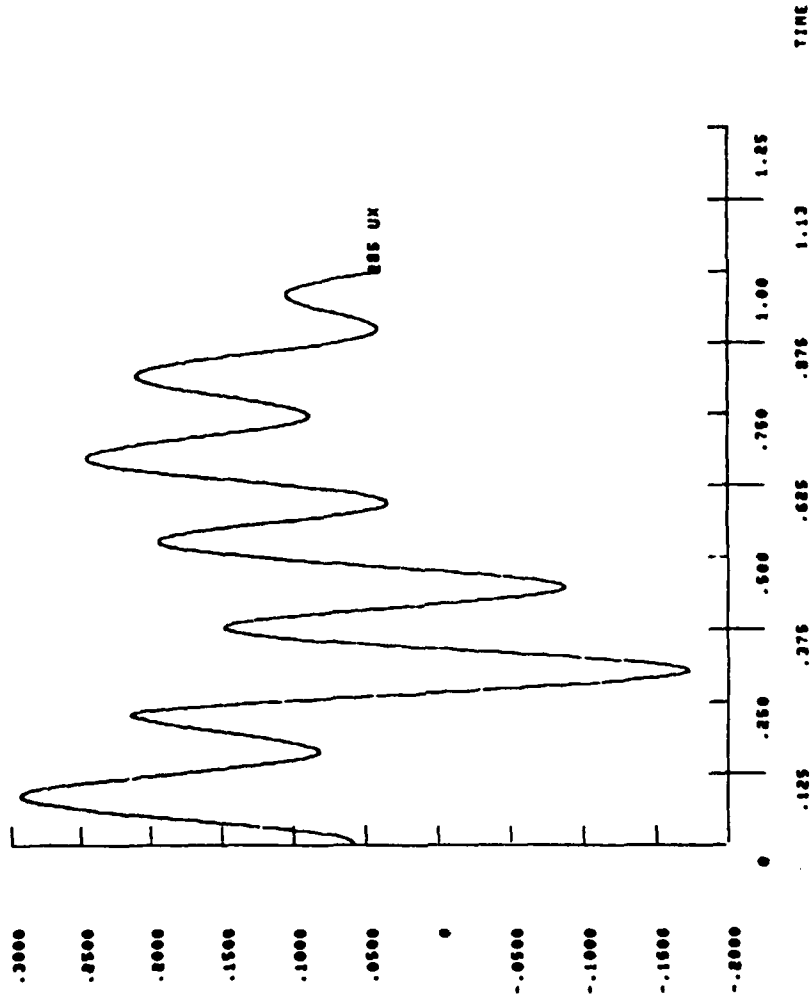
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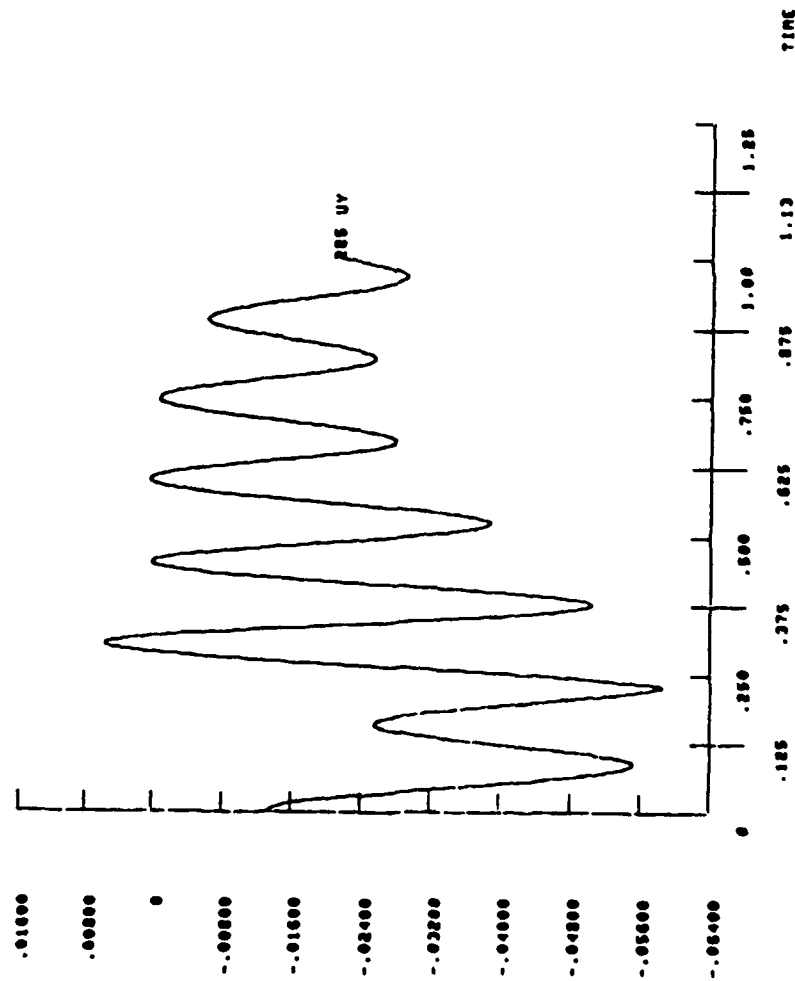
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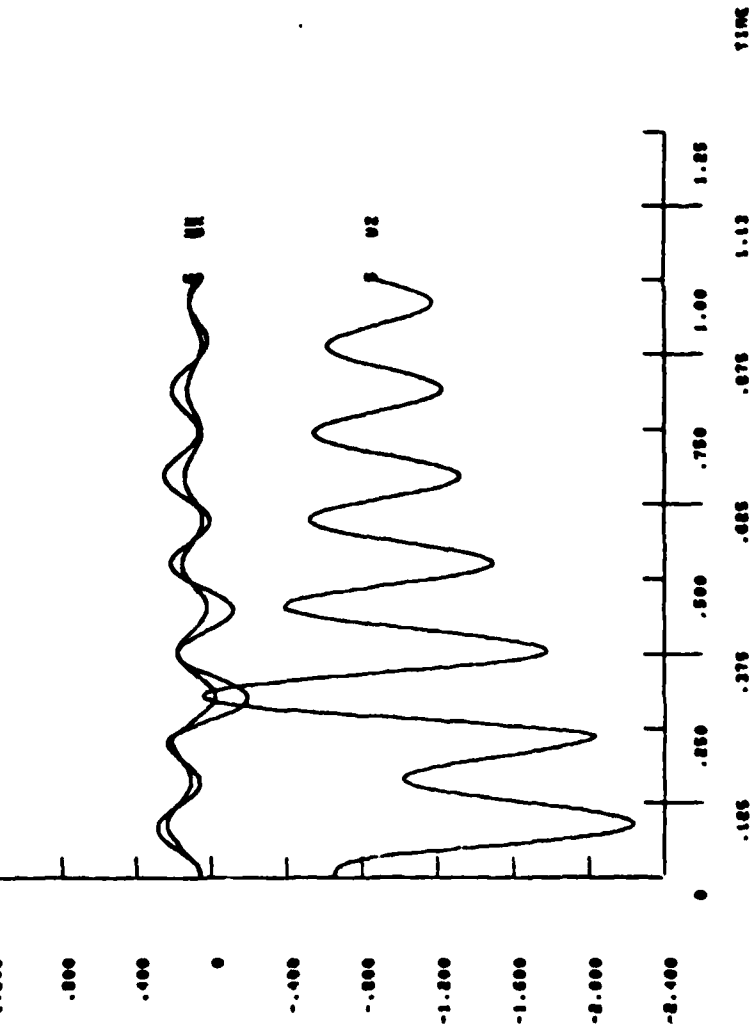
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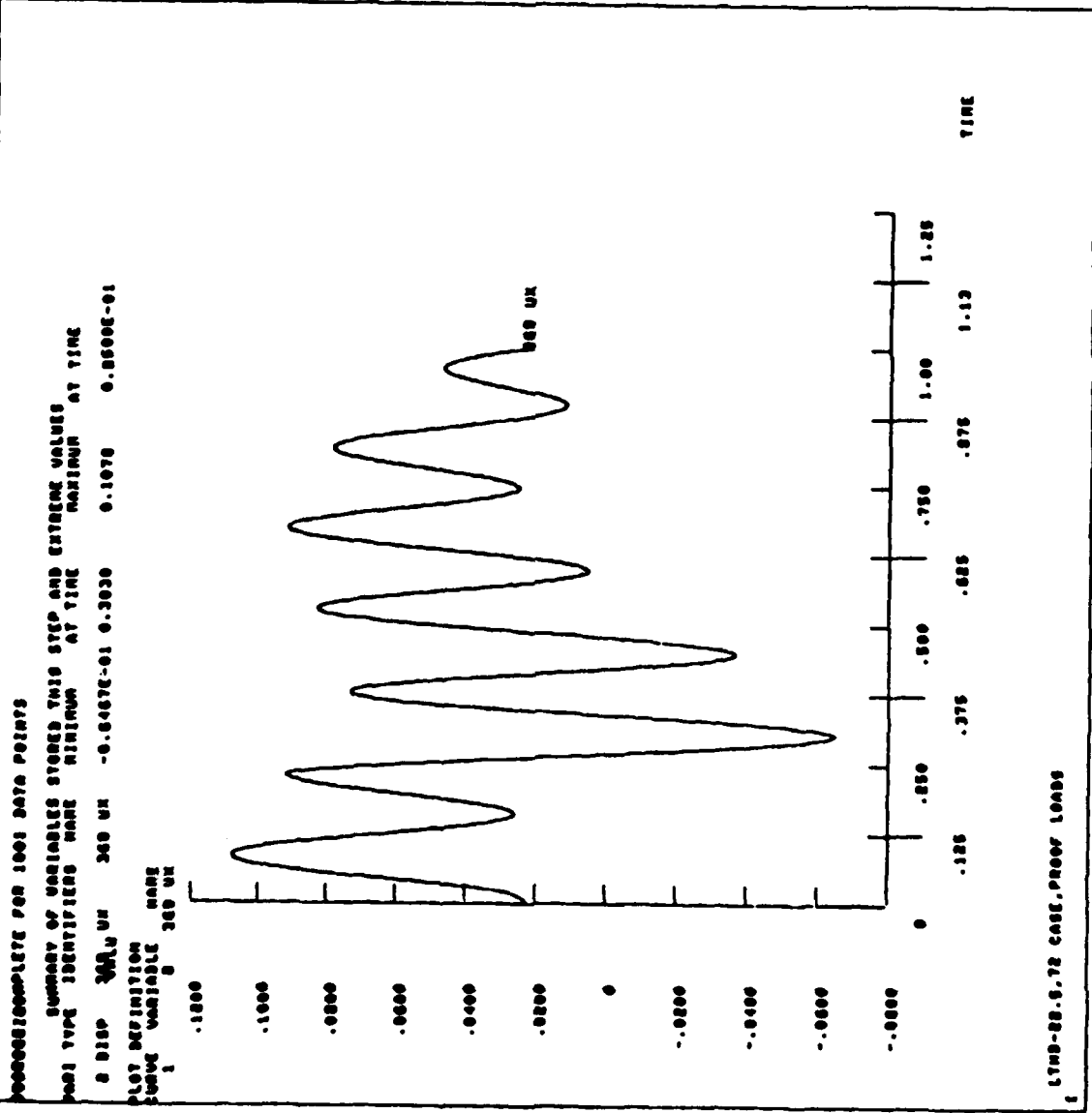
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LYND-28.0.72 CASE, PROOF LOADS



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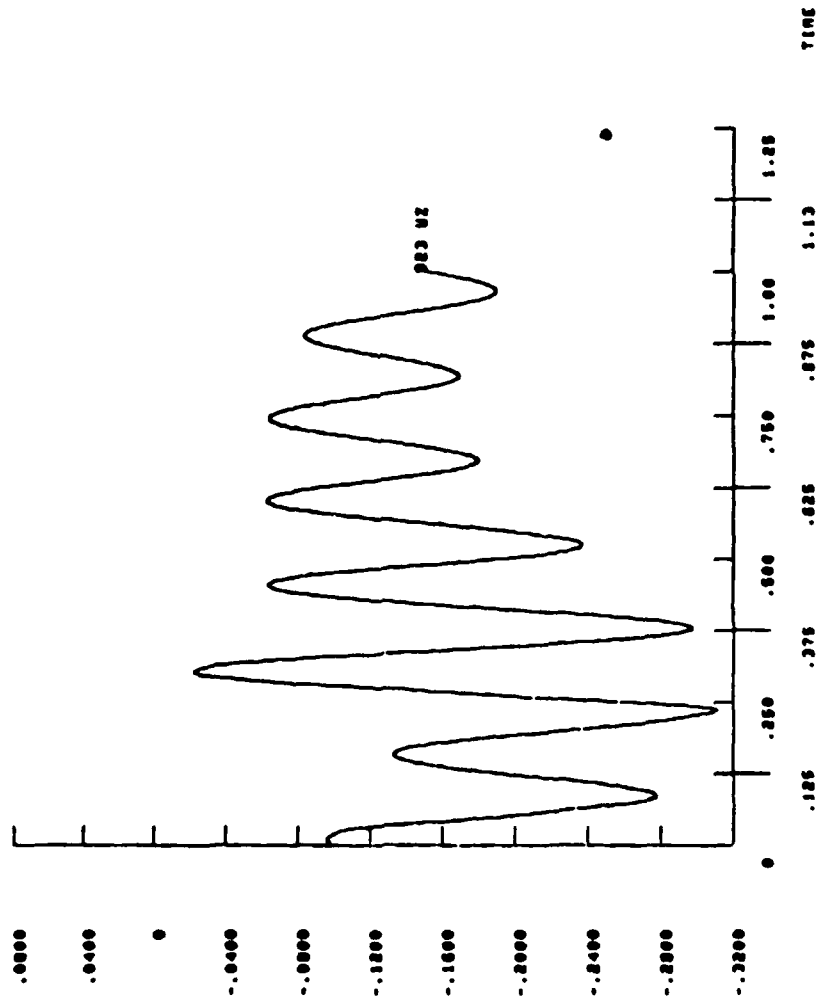
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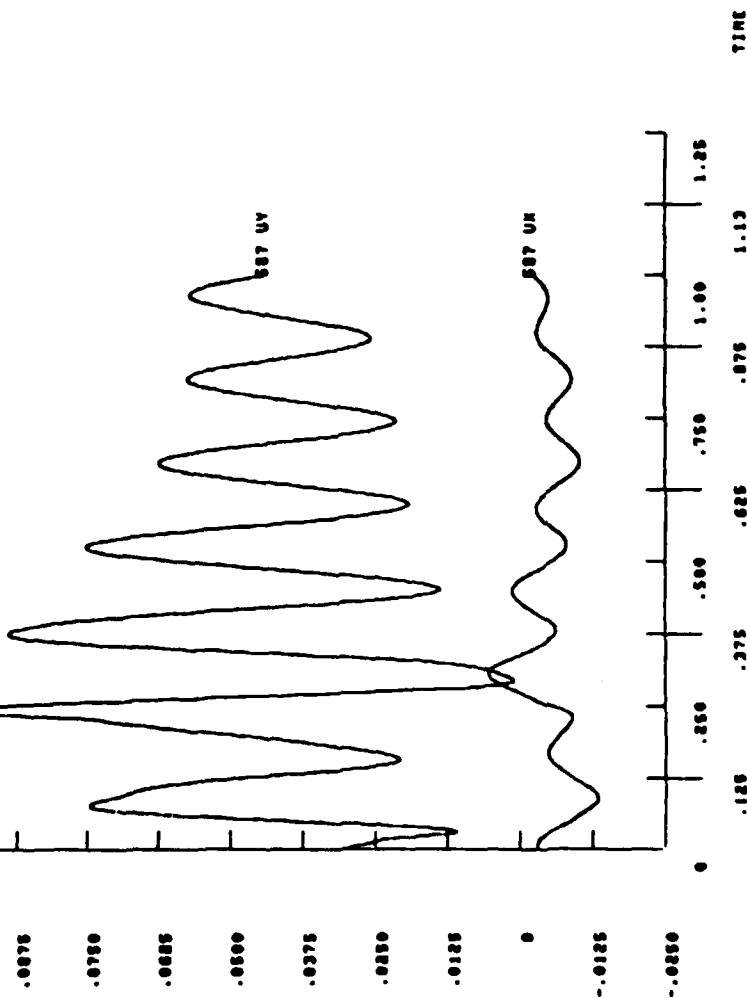


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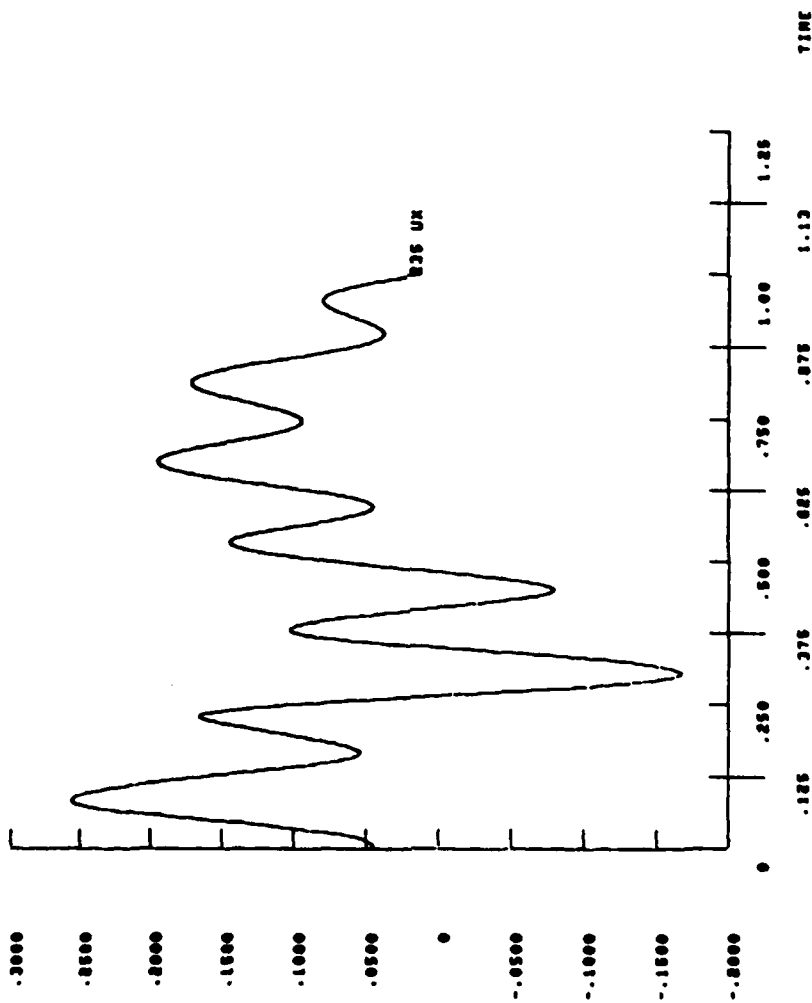
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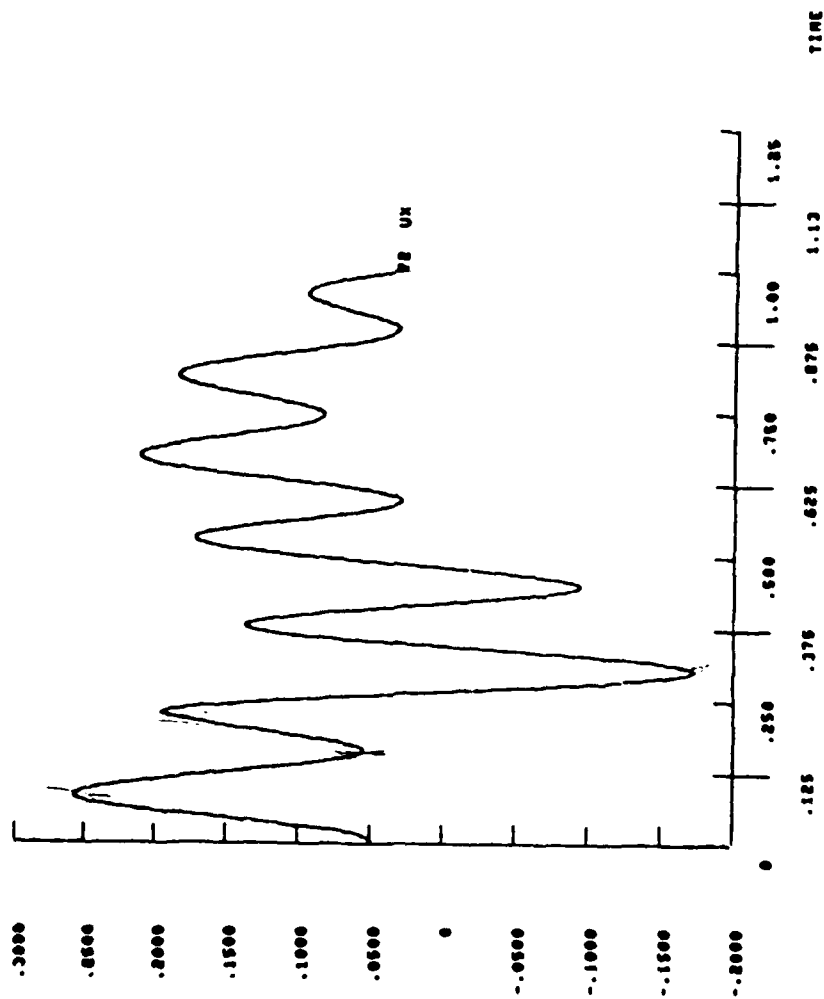
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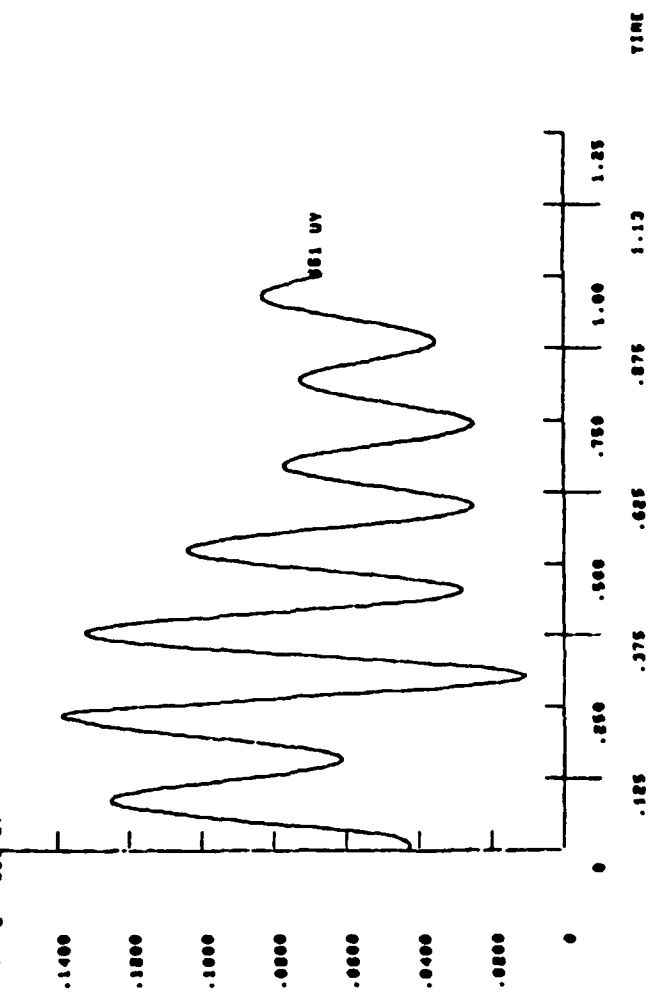
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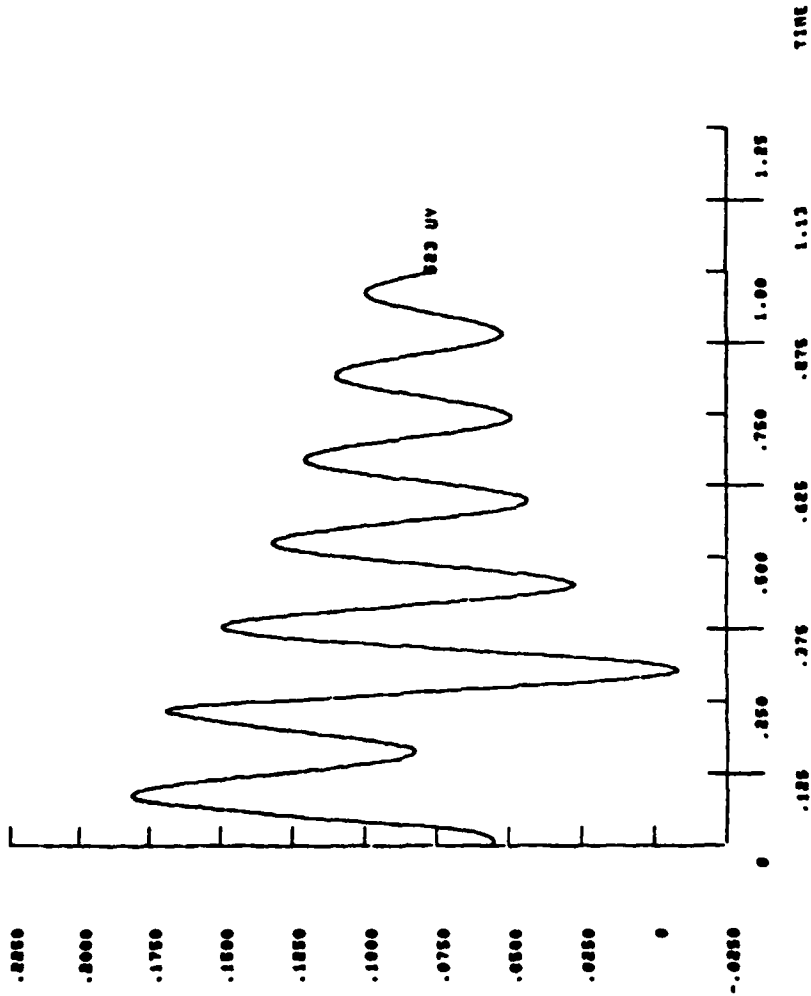


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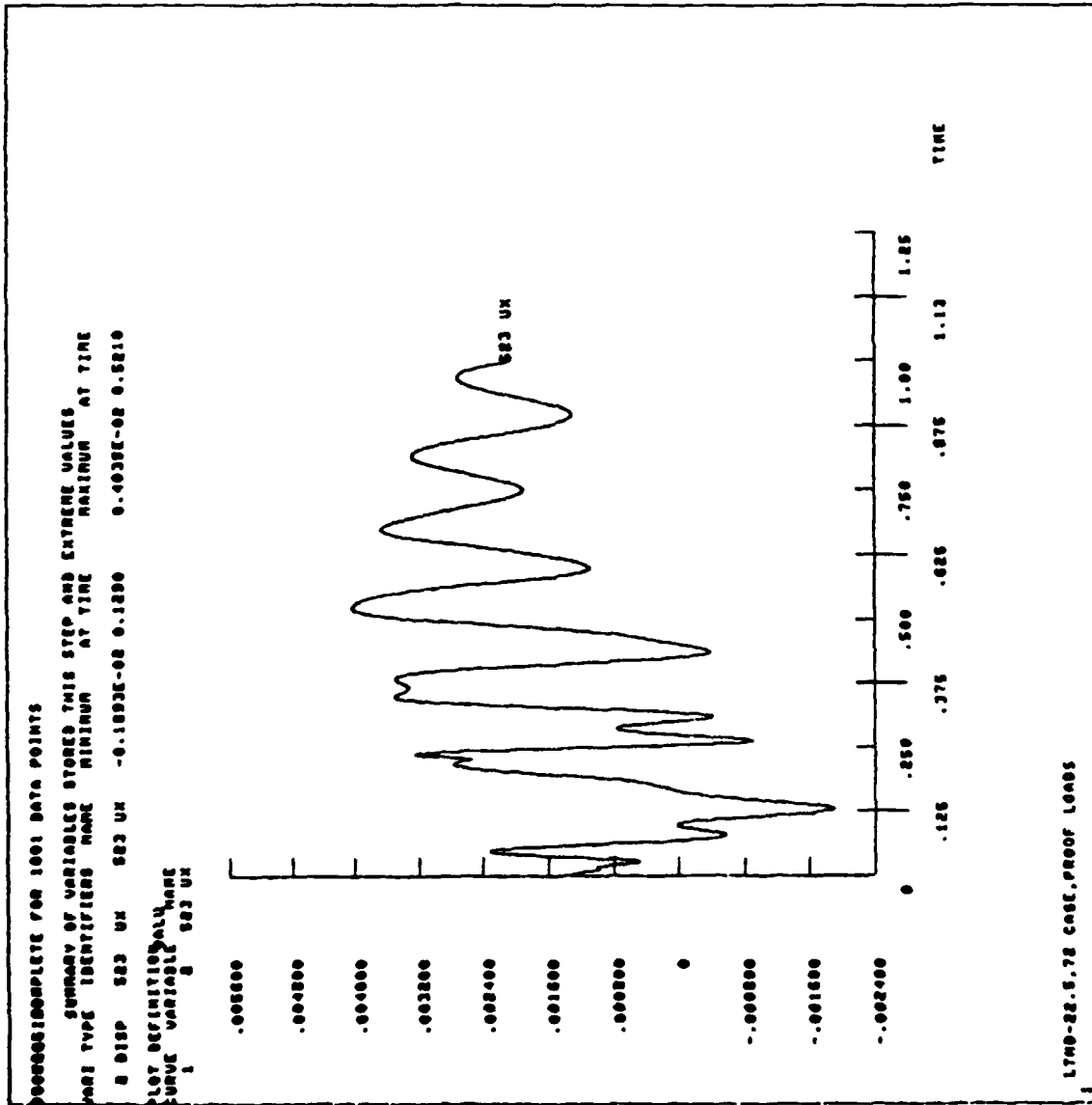
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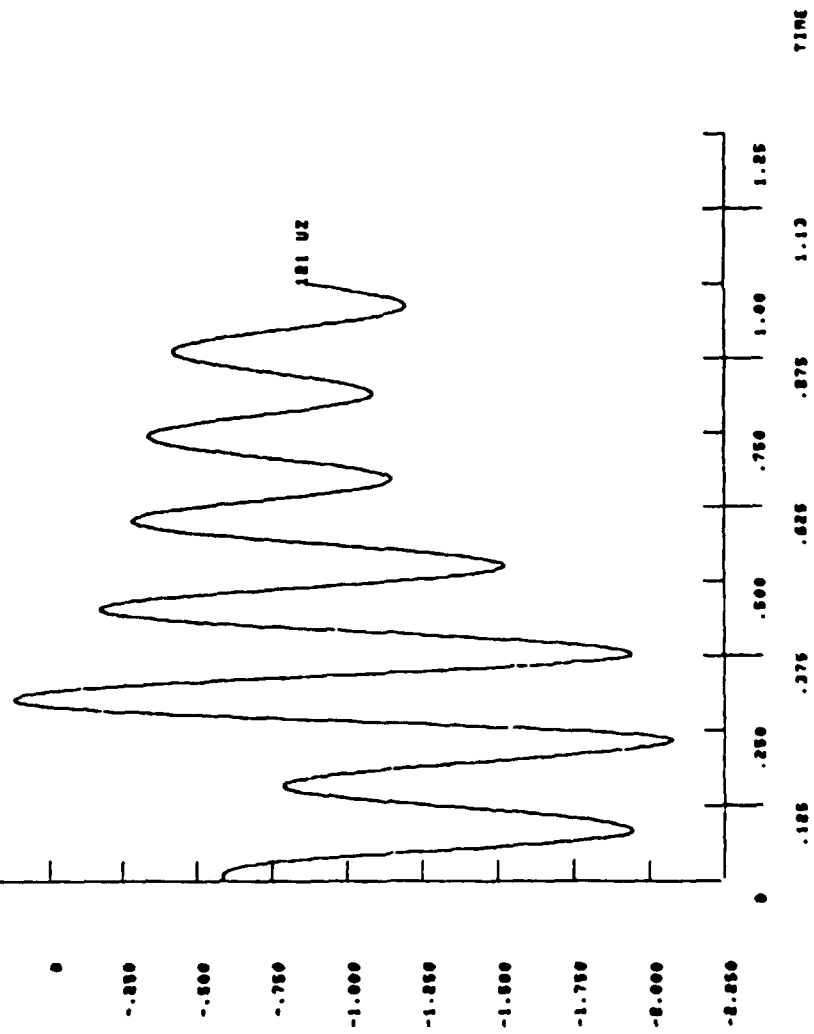
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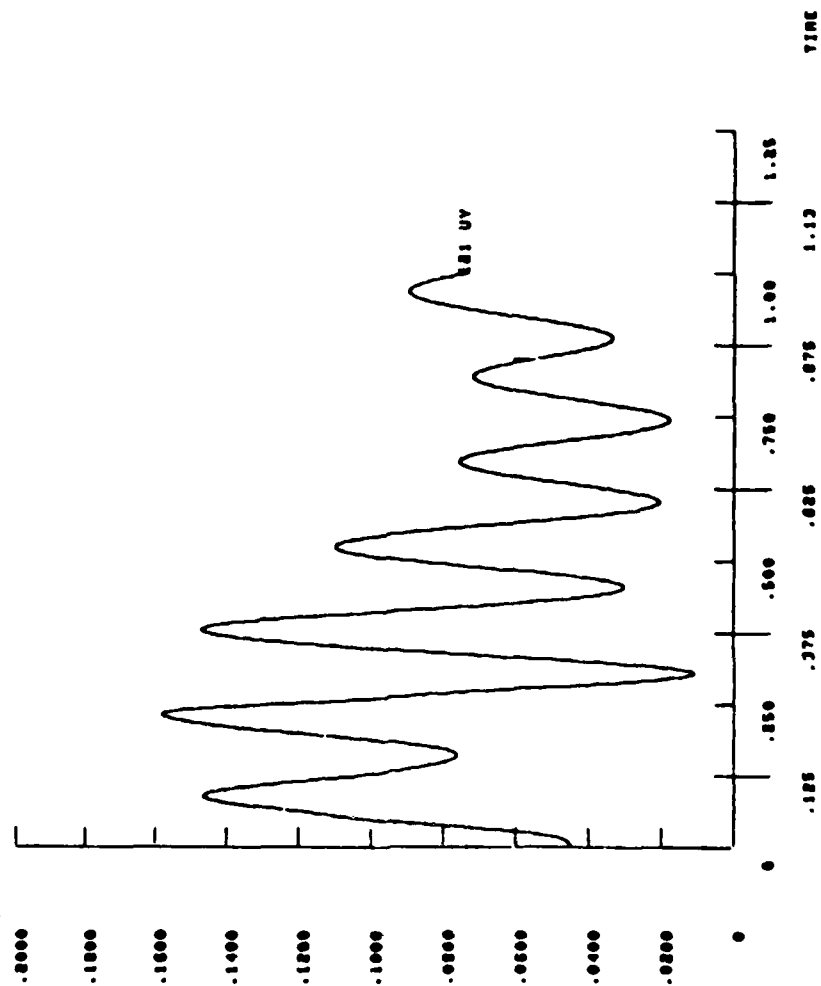
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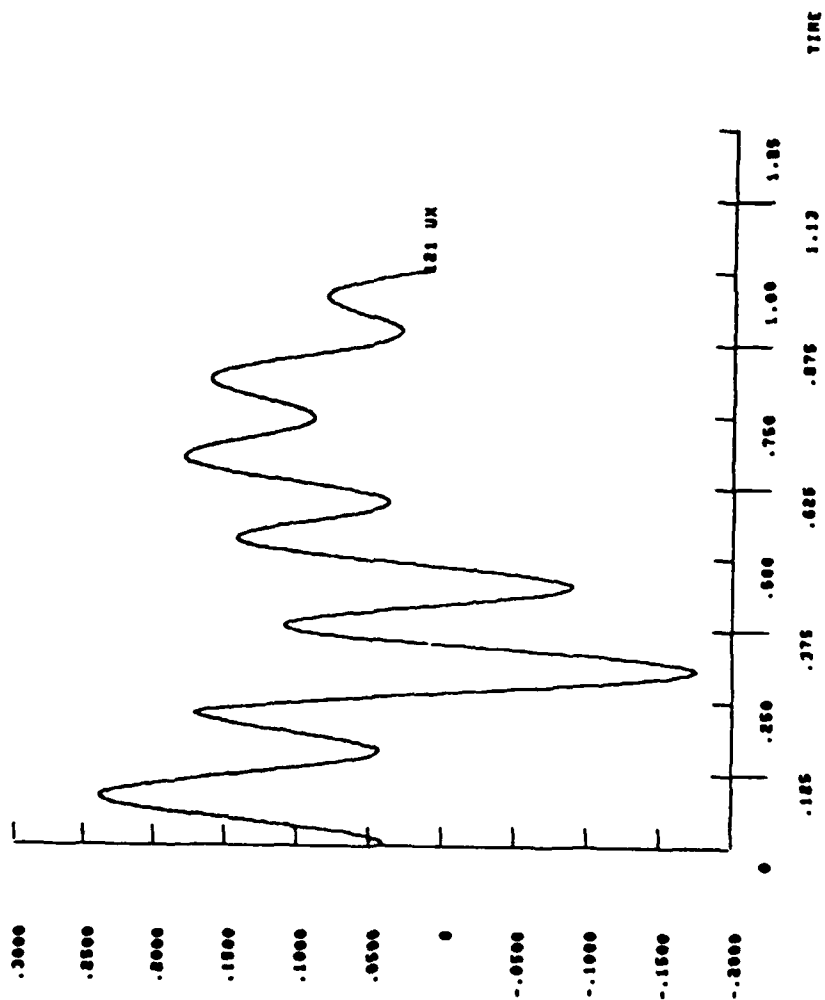
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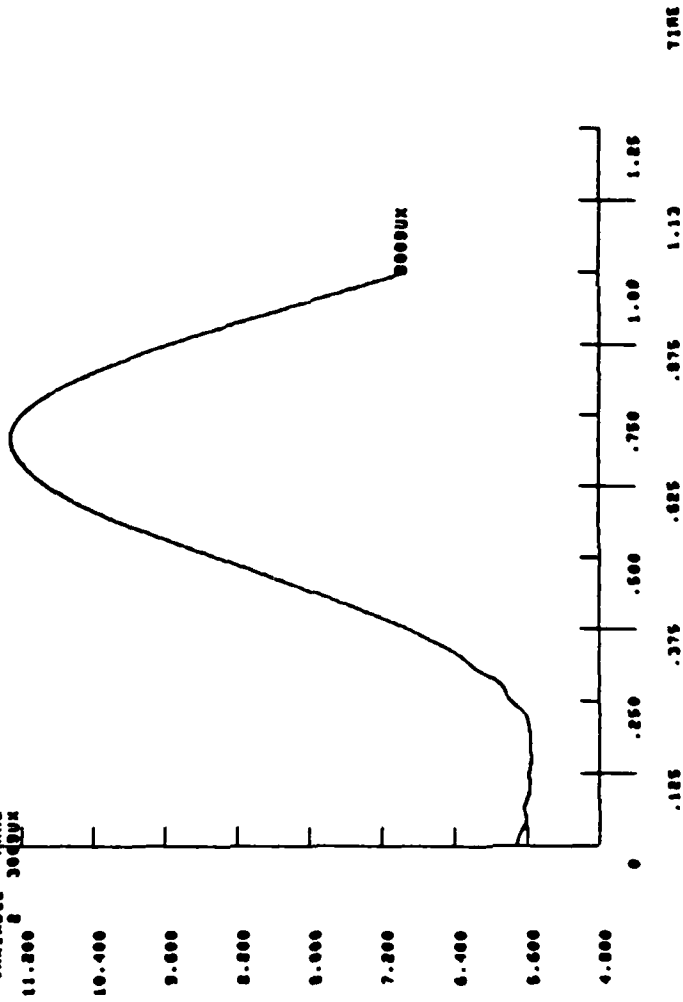
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 NAME TYPE IDENTIFIERS NAME MINIMUM AT TIME MAXIMUM AT TIME

NAME	TYPE	IDENTIFIERS	NAME	MINIMUM	AT TIME	MAXIMUM	AT TIME
2	DISP	3000 UX	3000UX	5.500	0.1550	11.31	0.7000

PLOT DEFINITION
 CURVE VARIABLE NAME
 1 11.000 2 3000UX



1 LTWD-22.9.72 CASE, PROOF LOADS

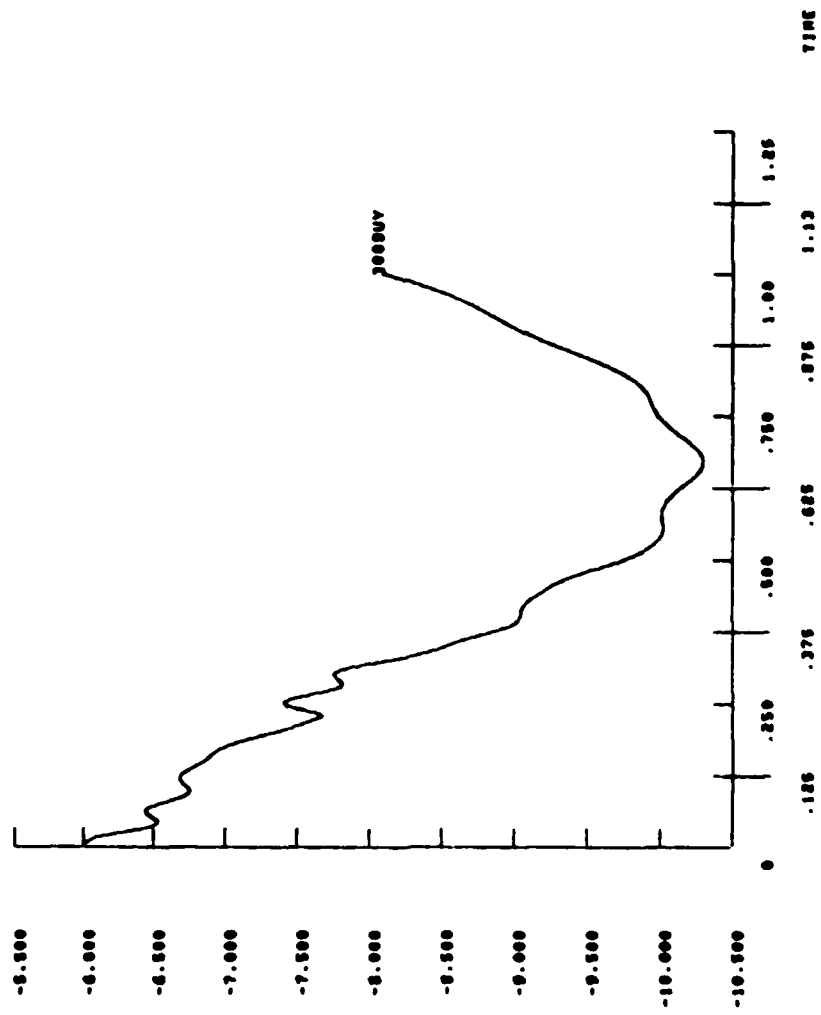
CRADLE FORWARD
 END MDOF NODE
 (SIDE-TO-SIDE MOTION)

*****COMPLETE FOR 1001 DATA POINTS

SUMMARY OF VARIABLES STORED THIS STEP AND EXTREME VALUES

NAME	TYPE	IDENTIFIERS	NAME	MINIMUM	AT TIME	MAXIMUM	AT TIME
U	DISP	3000UV	U	-10.30	0.6700	-0.005	0.00000000

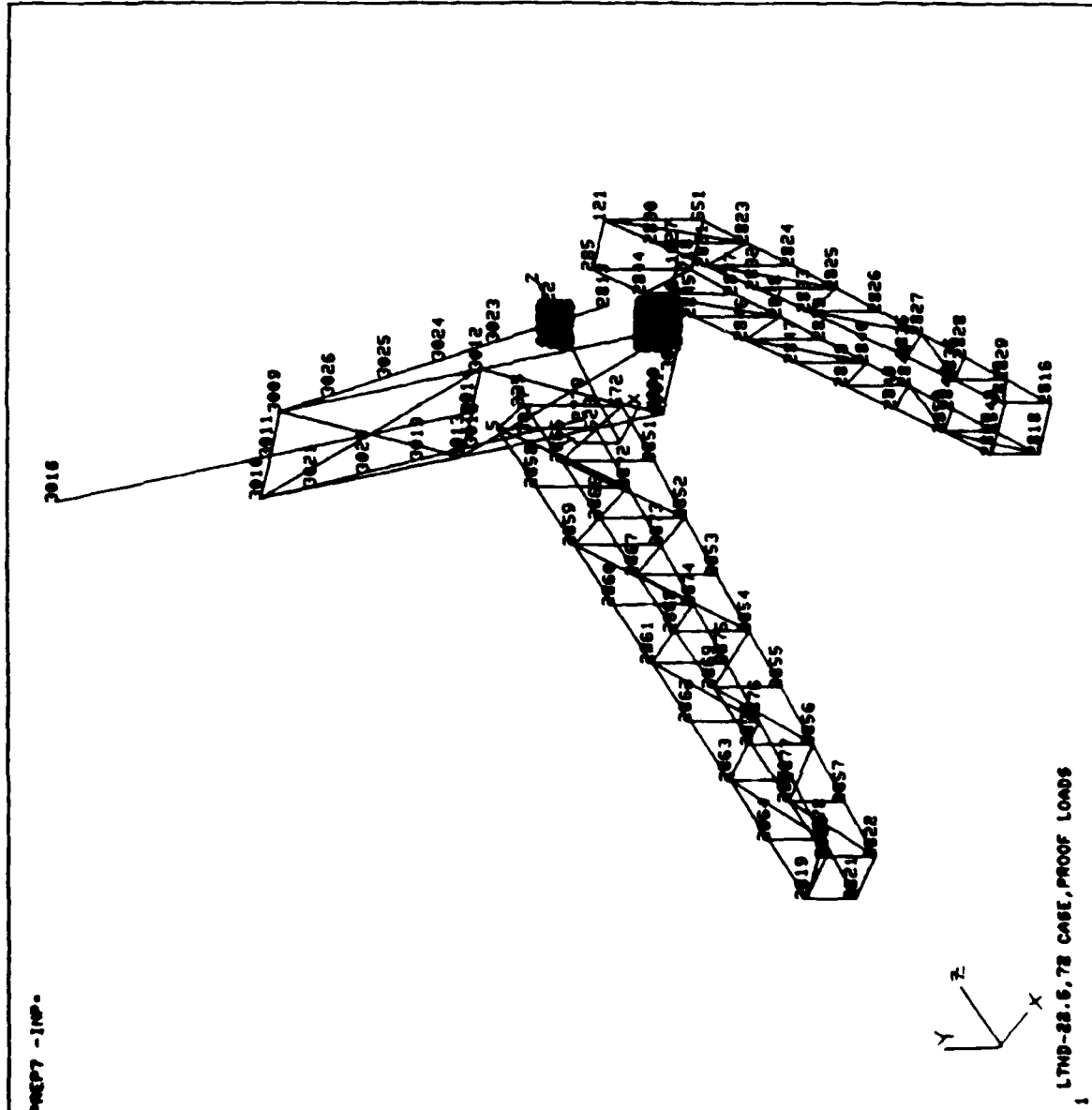
PLOT DEFINITION
 CURVE VARIABLE NAME
 1 3000UV

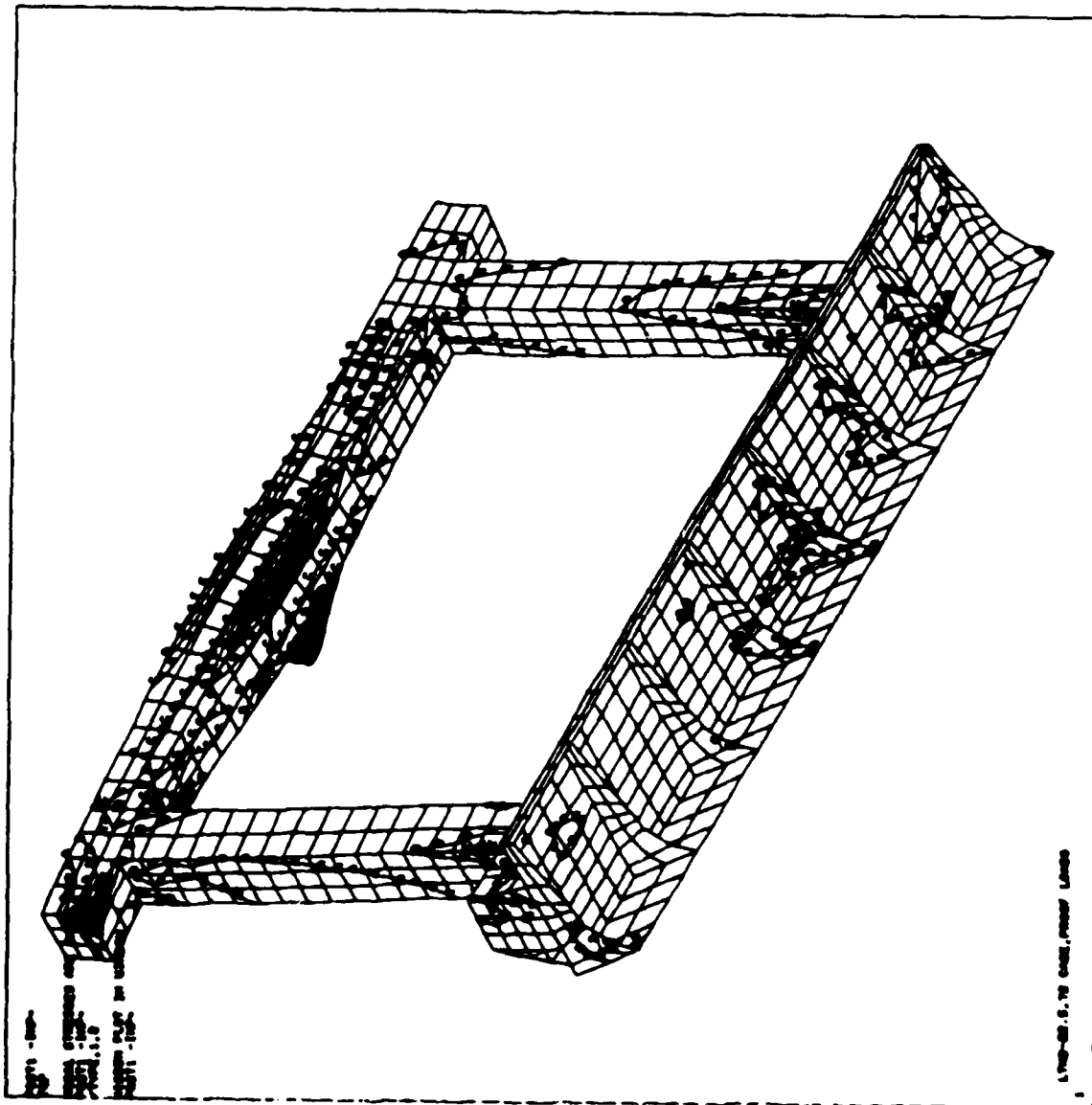


1 LTMS-22.5.72 CASE, PROOF LOADS

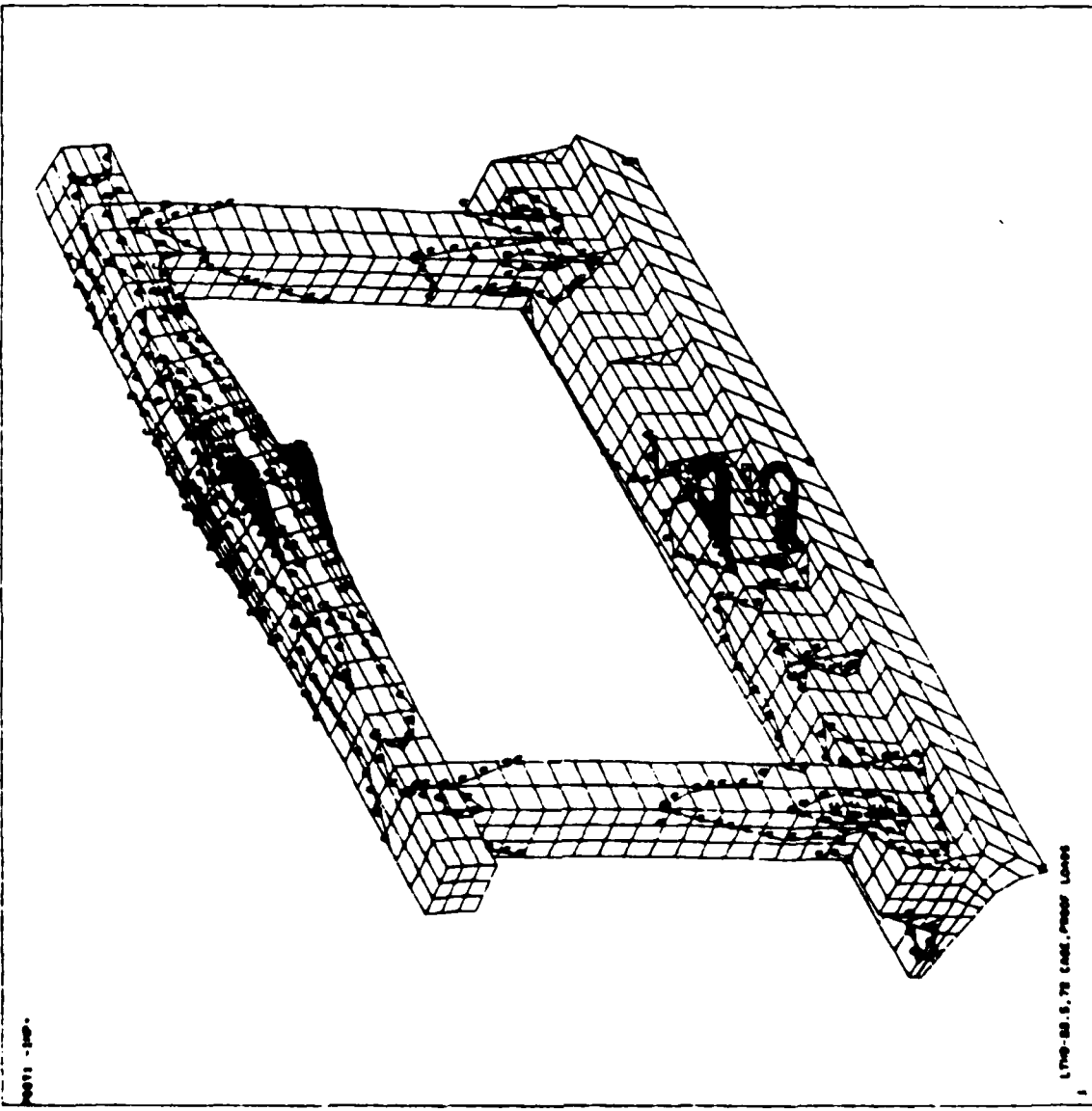
CRADLE FORWARD
 END MDOF NODE

ANDVS 4.28
 JAN 19 1987
 13130128
 PREP7 ELEMENTS
 MHUM-1
 XU--1
 VU--1
 ZU-1
 DIST=811
 XF=69.4
 YF=132
 ZF--129
 XRT0=4.06
 YRT0=3.64



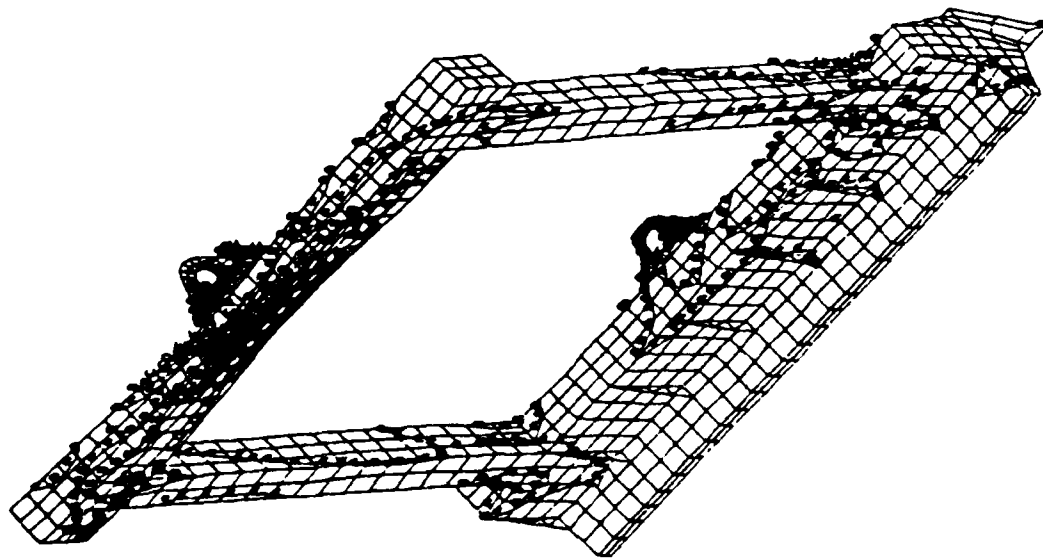
[illegible]

00071 0.00
 Jan 10 1987
 16:22:10
 00071 070000
 0700-1
 0700-1
 1100-1 204
 0700
 1100
 0700-1
 0700-1
 2000-1
 0700-100.8
 0700-100.2
 0700-100.4
 0700-100.6
 0700-100.8
 0700-101.0
 0700-101.2
 0700-101.4
 0700-101.6
 0700-101.8
 0700-102.0
 0700-102.2
 0700-102.4



PLATFORM

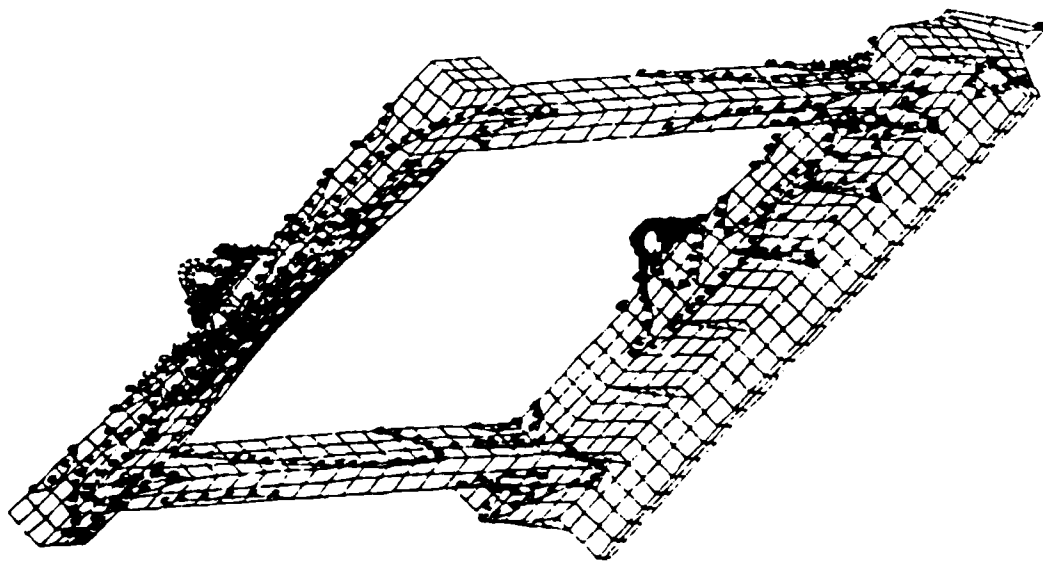
ANSYS 4.20
 JAN 10 1987
 15:28:46
 POST1 BYSTRESS
 STEP=1
 170-1
 TIME=.304
 SIDE
 TOP
 ZOOM
 ZV=1
 ZU=1
 ZW=1
 0.157-93.3
 XZ=51.2
 YZ=27.7
 ZF=9.37
 V870=1.45
 M180EM
 MX=29163
 MY=121
 A=4960
 B=8801
 C=14642
 D=19483
 E=24324



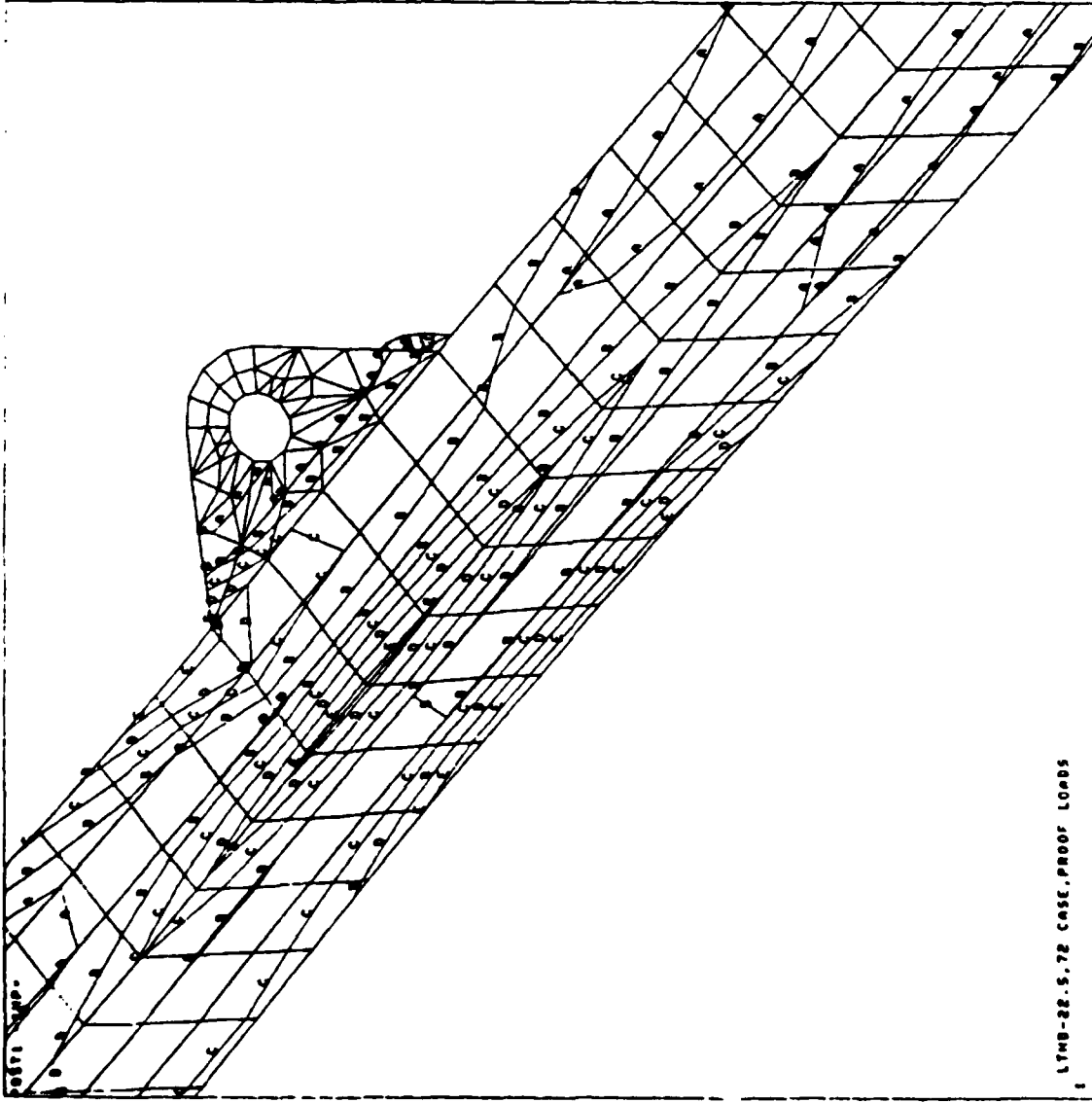
POST1 -IMP.

1749-22.5, 72 CASE, PROOF LOADS

ANSYS 4.20
 JAN 18 1987
 15:28:46
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=-.304
 SIZE
 TOP
 ZOOM
 KU=1
 YU=1
 ZU=1
 DIST=93.3
 WF=51.2
 VF=27.7
 ZF=4.17
 VATO=1.65
 MID8CM
 RM=20163
 RM=121
 A=4960
 B=9801
 C=14642
 D=19483
 E=24324

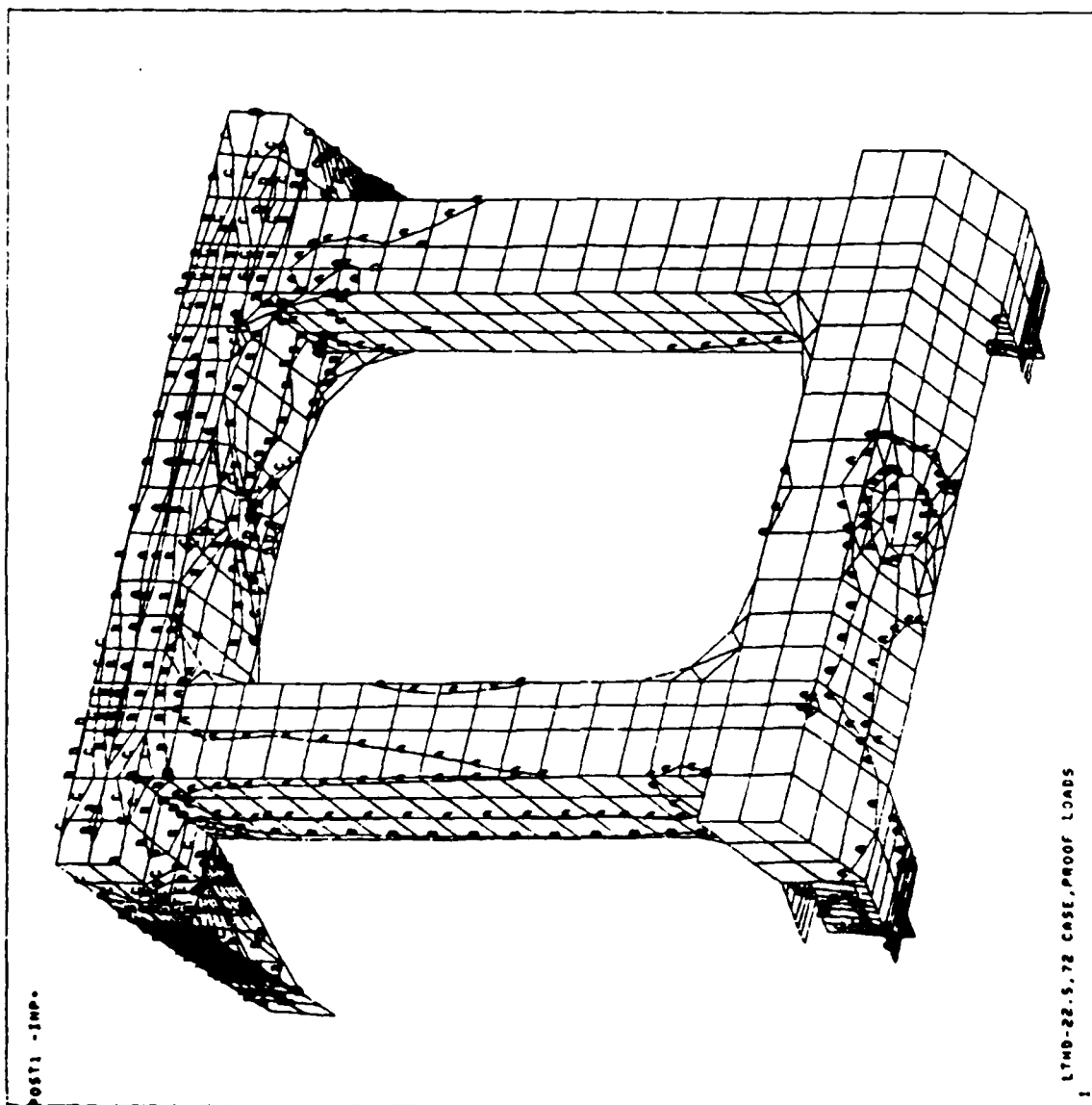


ANSYS 4.20
 JAN 19 1987
 15:28:46
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.304
 SICE
 TOP
 ZOOM
 RU=1
 VU=1
 ZU=1
 2 DIST=24.2
 3 XP=41.9
 4 YP=47.1
 5 ZP=-5.86
 6 XSTO=1.12
 7 YSTO=1.65
 8 HIDDEN
 9 HX=28180
 10 HX=0
 11 A=4960
 12 B=8801
 13 C=14642
 14 D=19483
 15 E=24324

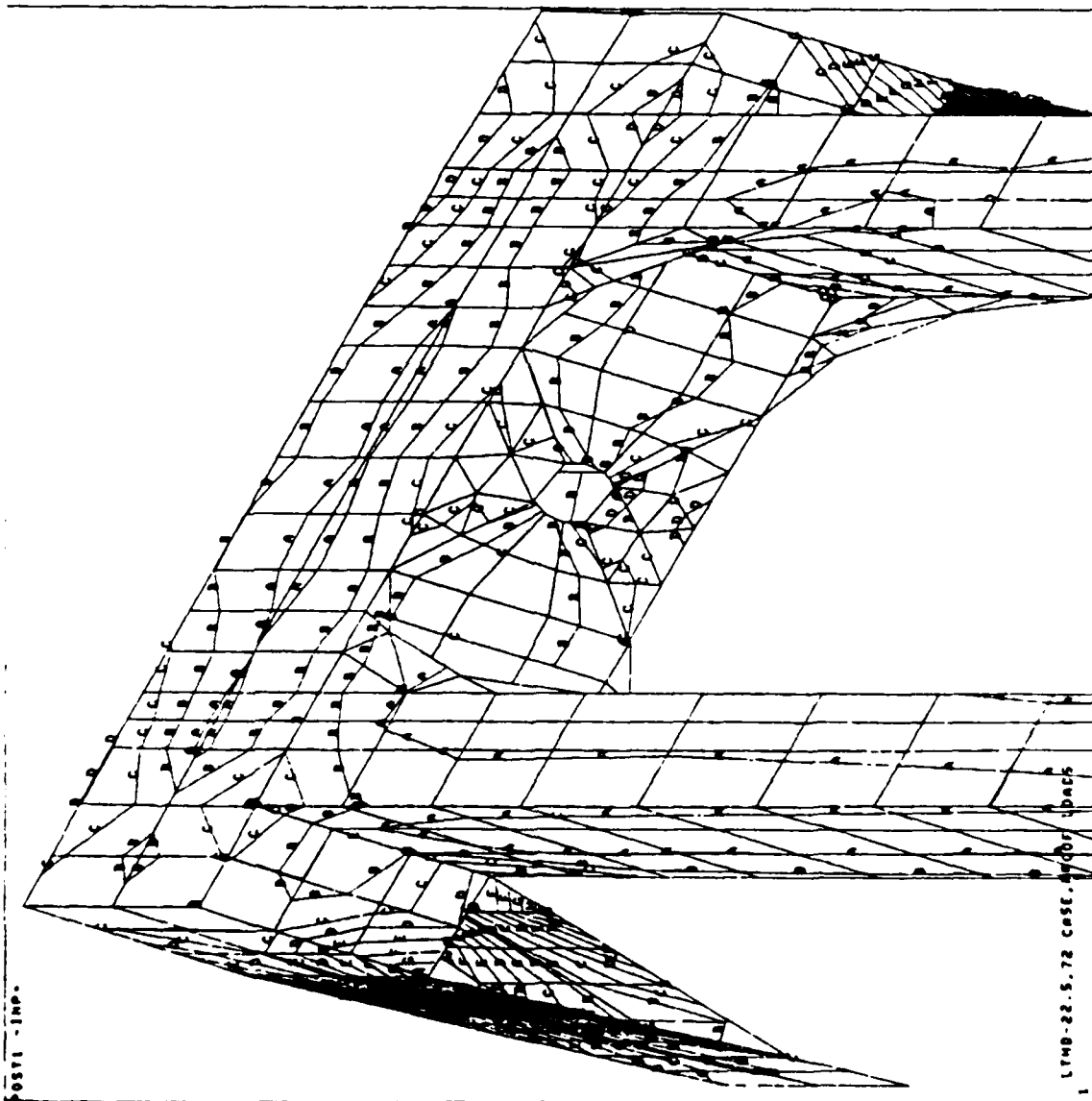


PLATFORM
 UPPER BOX
 BEAM

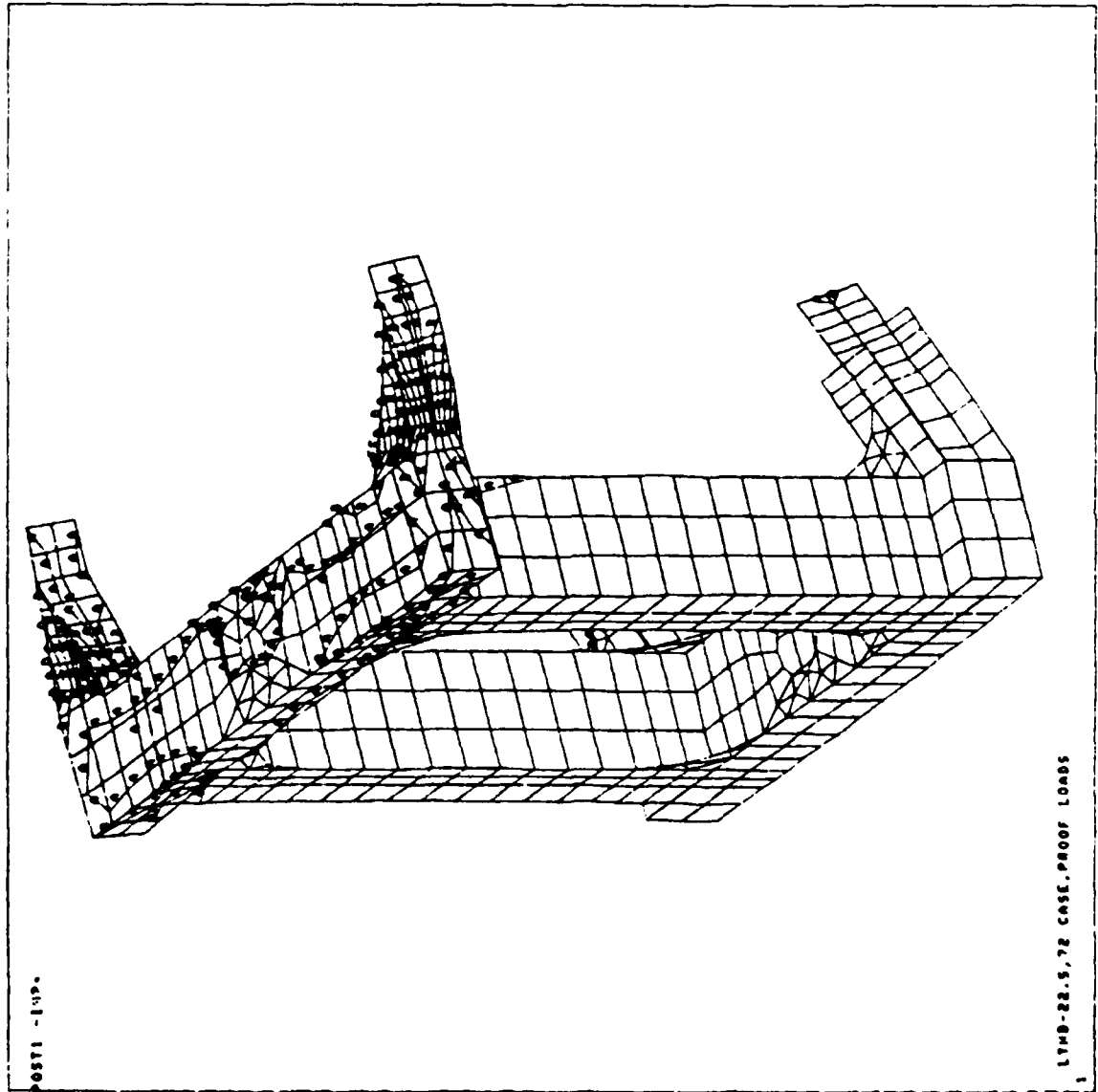
AMSYS 4.20
 JAN 19 1987
 15140153
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.304
 SIZE
 TOP
 KU=-1
 VU=-1
 ZU=1
 DIST=88.7
 KF=53.8
 VF=35
 ZF=-7.83
 MIDDLE
 RM=235105
 RM=660
 A=15311
 B=29964
 C=44617
 D=59278
 E=73923
 F=88576
 G=103229
 H=117882
 I=132535
 J=147188
 K=161841
 L=176494
 M=191147
 N=205800
 O=220453



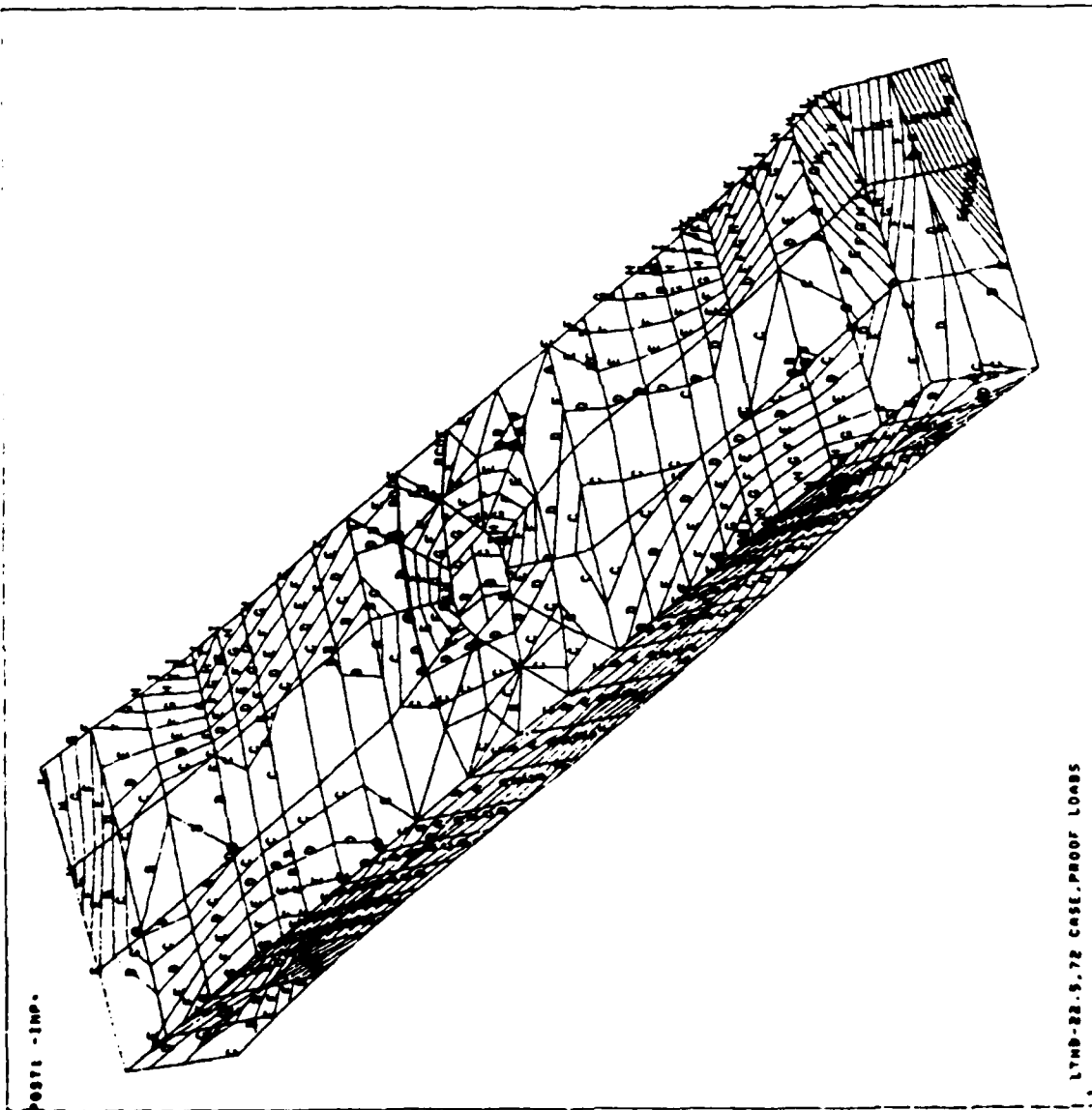
ANSYS 4.20
 JAN 19 1987
 15140158
 POST1 STRESS
 STEP=1
 LAYER=1
 TIME=.204
 SLOC
 TOP
 ZOOM
 XU=-1
 VU=-1
 ZU=1
 # DIST=24.8
 # XF=46.6
 # VF=48
 # ZF=-1.81
 VRTO=2.45
 MIDDEN
 RK=235105
 RN=6
 A=15311
 B=20964
 C=44617
 D=59270
 E=73923
 F=98576
 G=103229
 H=117882
 I=132535
 J=147100
 K=161841
 L=176494
 M=191147
 N=205800
 O=220453



ANSYS 4.20
 JAN 10 1987
 15:50:54
 POST1 STRESS
 STEP=1
 L100=1
 TIME=.304
 SICE
 TOP
 NU=1
 VU=1
 ZU=1
 D167-34.0
 XF-57.3
 YF-32.9
 ZF--7.62
 MID=1
 MM-235105
 MM-660
 A-39732
 B-78807
 C-117802
 D-156857
 E-106032



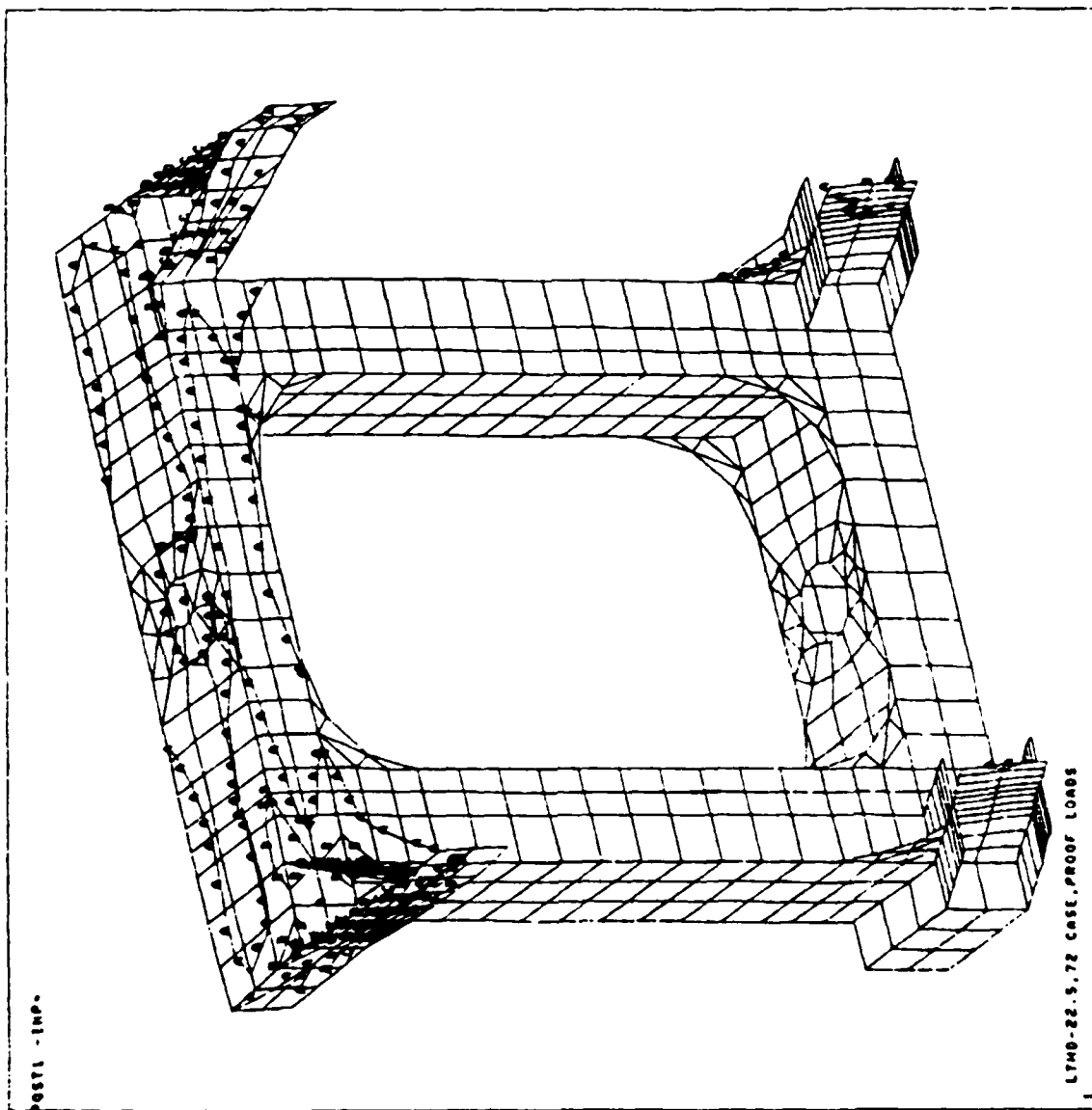
ANSYS 4.20
 JAN 19 1987
 16:07:06
 POST1 STRESS
 STEP=1
 LREQ=1
 TIME=.304
 SICE
 TOP
 ZOOM
 XU=1
 YU=1
 ZU=1
 BIST=16.8
 XF=53
 VF=54.7
 ZF=-3.5
 XROT=1.23
 YROT=1.08
 MIBDEM
 MN=121964
 MN=9161
 A=16205
 B=23256
 C=30307
 D=37358
 E=44400
 F=51460
 G=58511
 H=65562
 I=72613
 J=79664
 K=86715
 L=93766
 M=100817
 N=107868
 O=114919



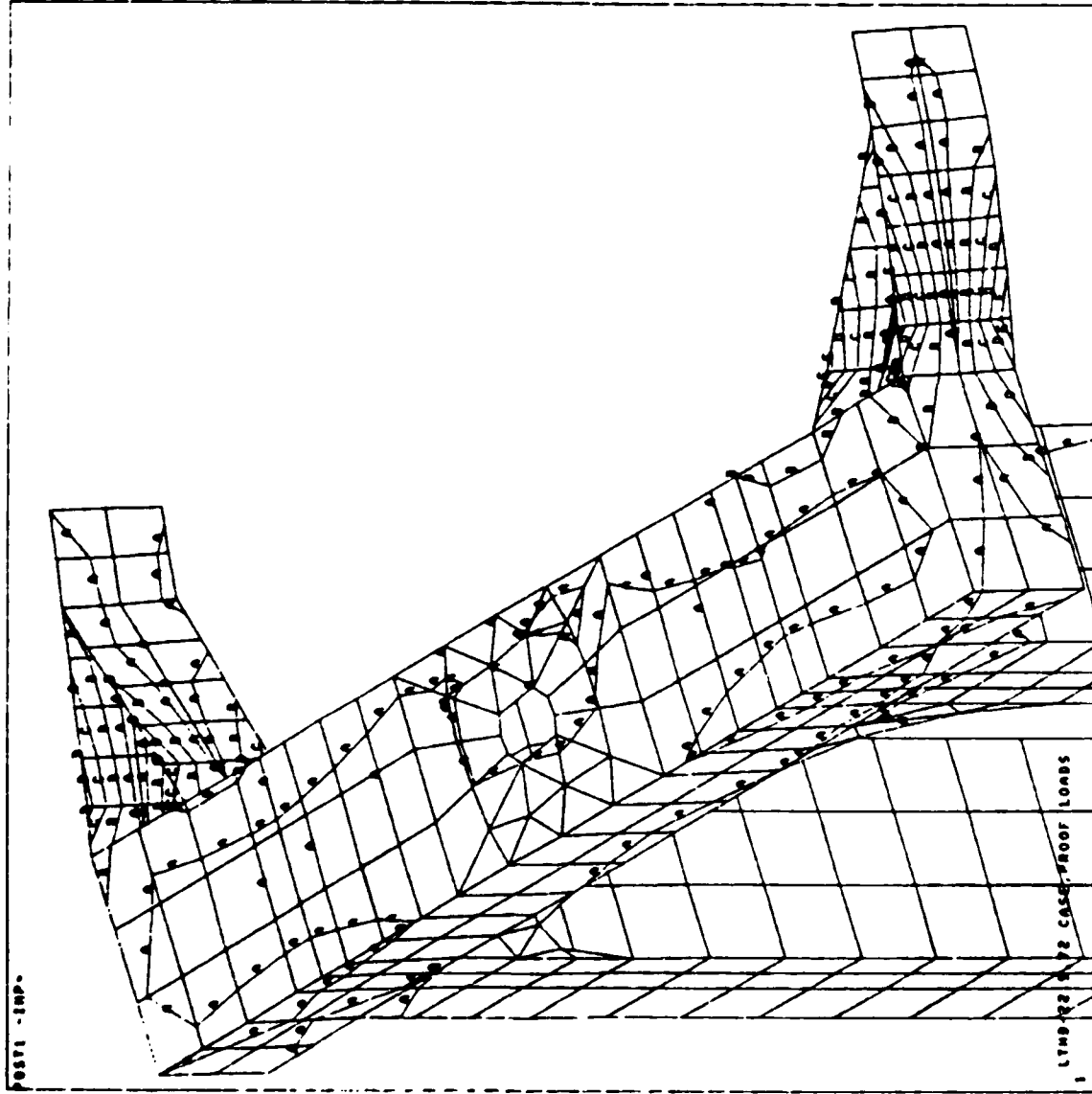
UPPER G-MIBAL
 BOX BEAM

530 (12)

ANSYS 4.28
 JAN 19 1987
 16:34:24
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.304
 SICE
 TOP
 NU=1
 VU=1
 ZU=-1
 DIST=28.7
 WF=52.9
 VF=35
 ZF=-7.93
 MIDDLEM
 RN=235105
 RN=659
 A=38732
 B=78807
 C=117882
 D=156957
 E=196932

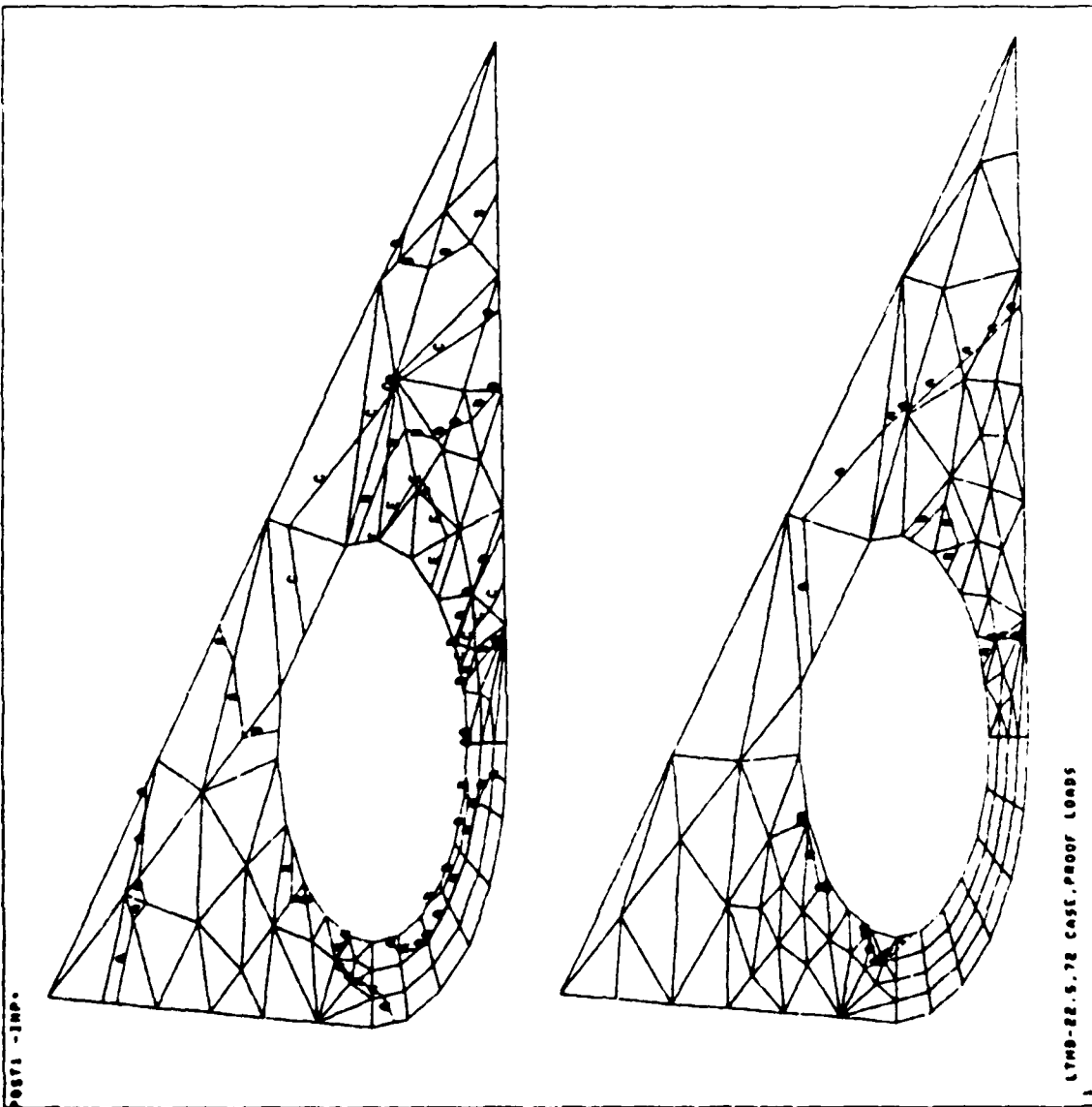


ANSYS 4.20
 JAN 19 1987
 15:50:54
 POST1 STRESS
 STEP=1
 ITEM=1
 TIME=...304
 SLC
 TOP
 ZOOM
 KU=1
 VU=1
 ZU=1
 8 DIST=10
 8 KF=48.0
 8 VF=48.0
 8 ZF=15.1
 VRT0=1.21
 MIDDLE
 RK=235105
 RM=0
 A=20732
 B=70807
 C=117002
 D=156957
 E=106032

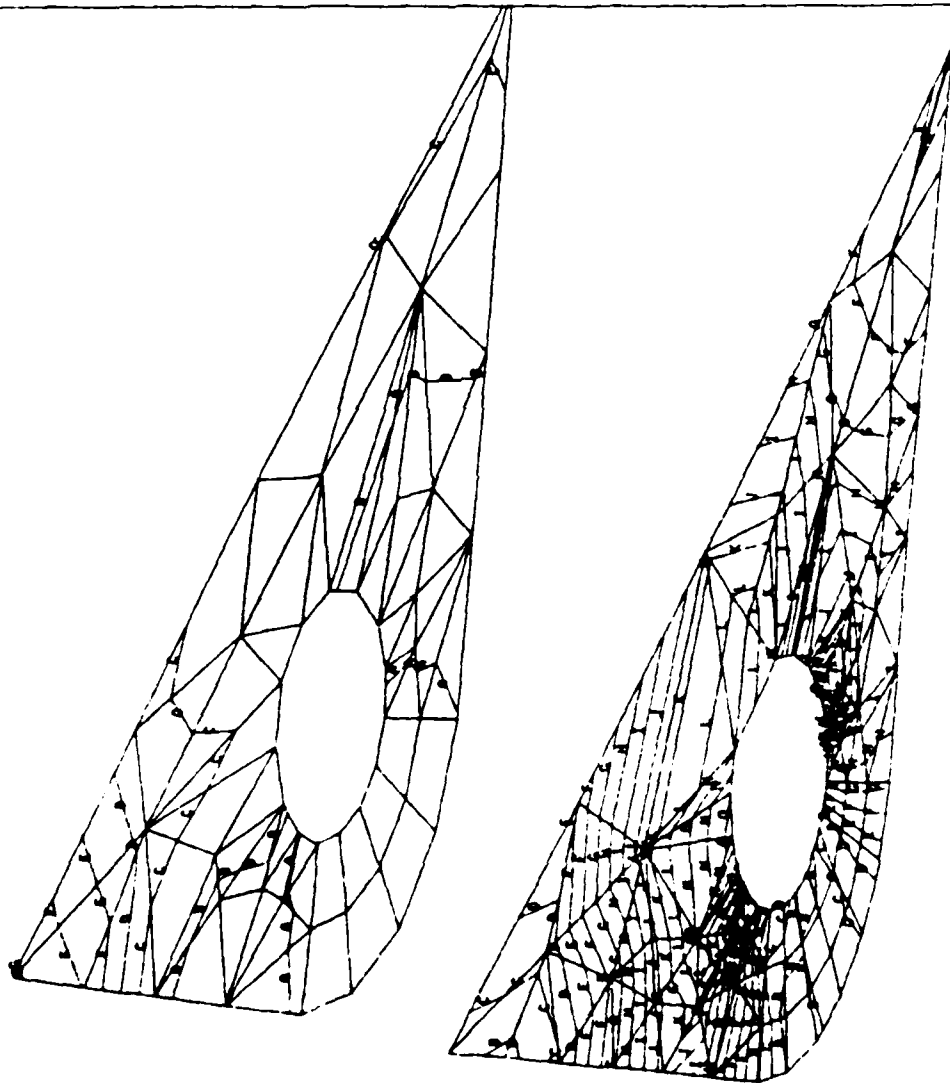


ANSYS 4.20
 JAN 19 1987
 15:58:40
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=1.004
 SLOC
 TOP

ZOOM
 RV=1
 VU=1
 ZU=1
 E 0187-7.56
 E 07-61.0
 E VF-13.2
 E 27-0.58
 KEYO=1.23
 MIDDLE
 RM=29163
 RM=121
 A=4960
 B=8801
 C=14648
 D=18483
 E=24324



50101 - IMP.



LYND-22-S.72 CASE, POOR LOADS

UPPER
TABS

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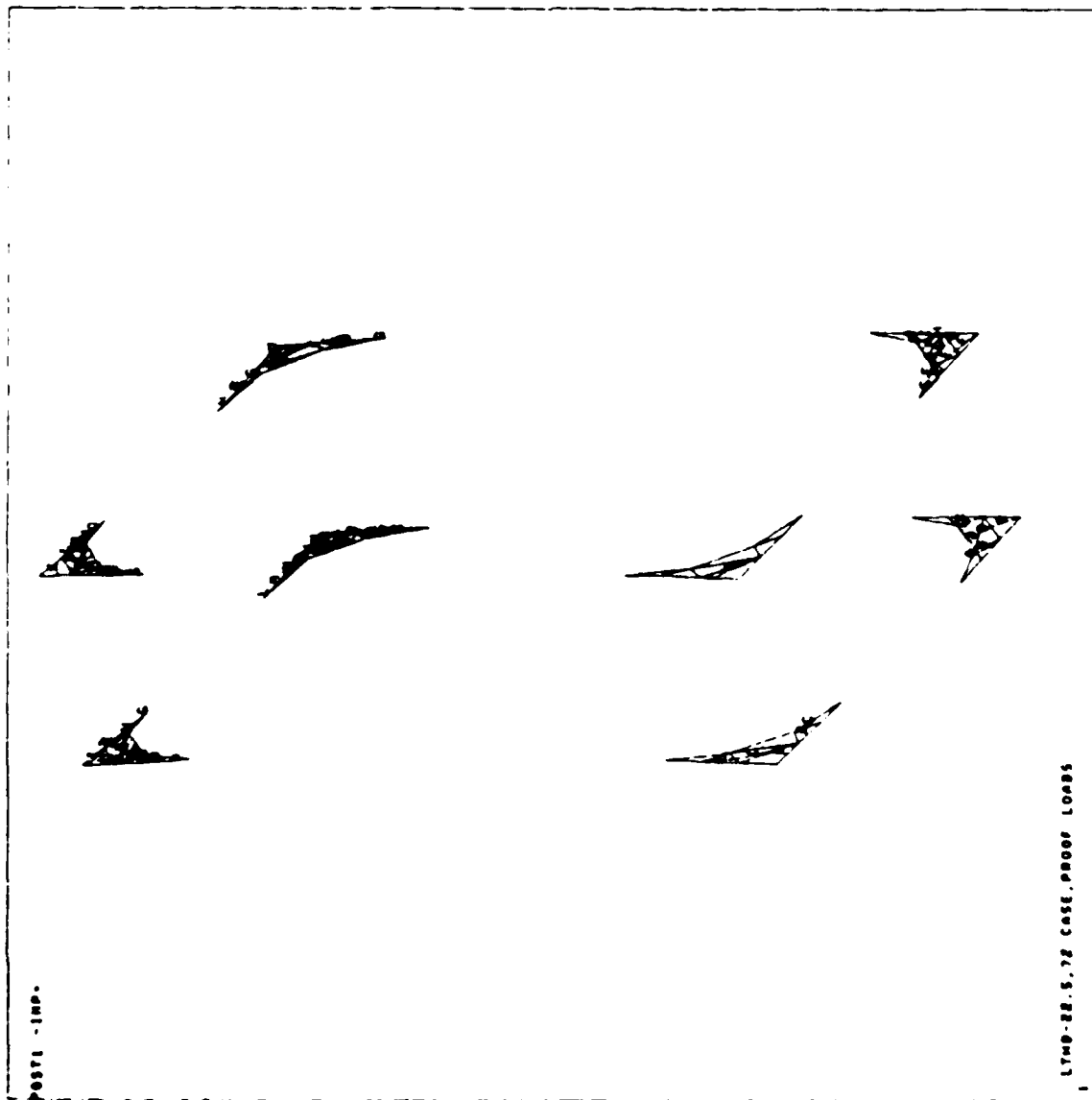
LYMB-22.5.72 CASE, POOR LOADS

535

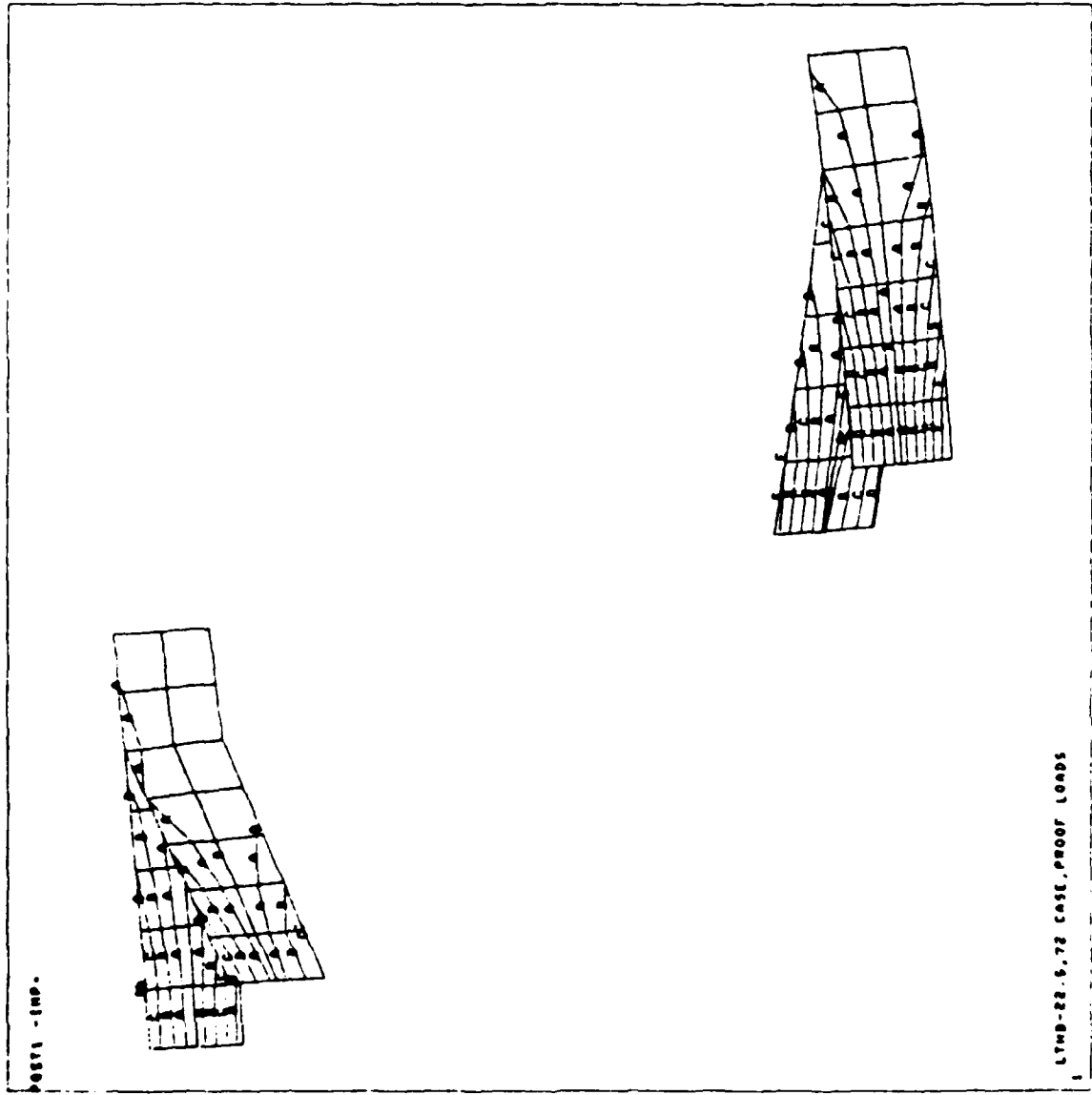
44

ANSYS 4.80
 JAN 19 1987
 16:00:27
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.304
 SICE
 TOP
 ZOOM
 RV=1
 VU=1
 20=1
 DIST=86.8
 RF=53
 VF=23.6
 ZF=-3.5
 WOTO=1.23
 WOTO=1.08
 MIDELM
 ME=98794
 ME=3461
 A=6200
 B=8124
 C=11858
 D=14792
 E=17626
 F=20450
 G=23294
 H=26128
 I=28962
 J=31786
 K=34630
 L=37464
 M=40288
 N=43132
 O=45966

GIMBAL REINFORCEMENTS

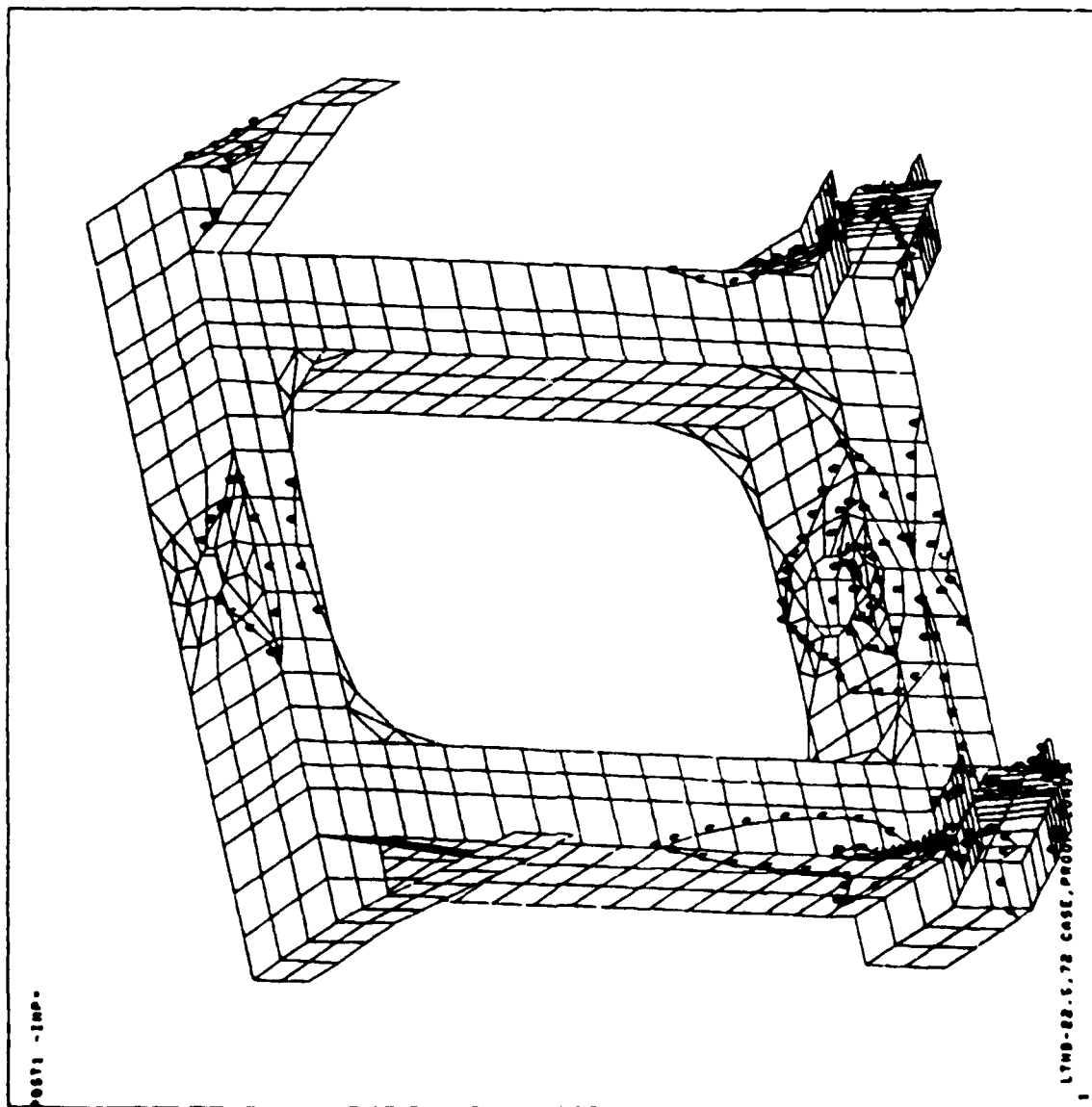


ANSYS 4.20
 JAN 19 1987
 16118116
 POST1 STRESS
 STEP=1
 ZEP=1
 TIME=.304
 SLOC
 TOP
 ZOOM
 KU=1
 VU=1
 ZU=1
 DISP=19.4
 XF=57.1
 YF=53.3
 ZF=-13.8
 XRTD=1.23
 YRTD=1.08
 MI00EN
 RM=251001
 RM=17469
 A=66556
 B=95603
 C=134720
 D=173817
 E=212004

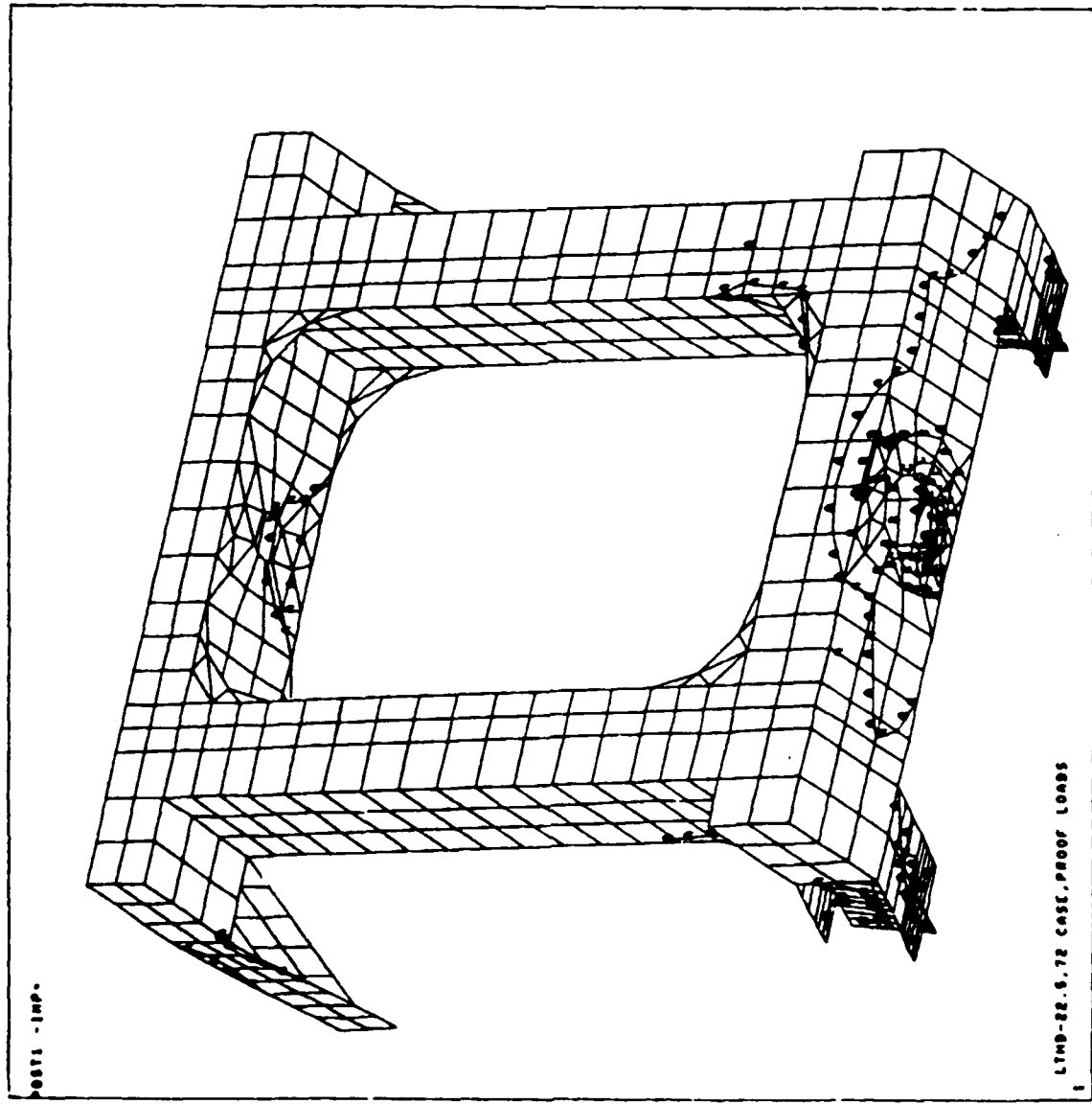


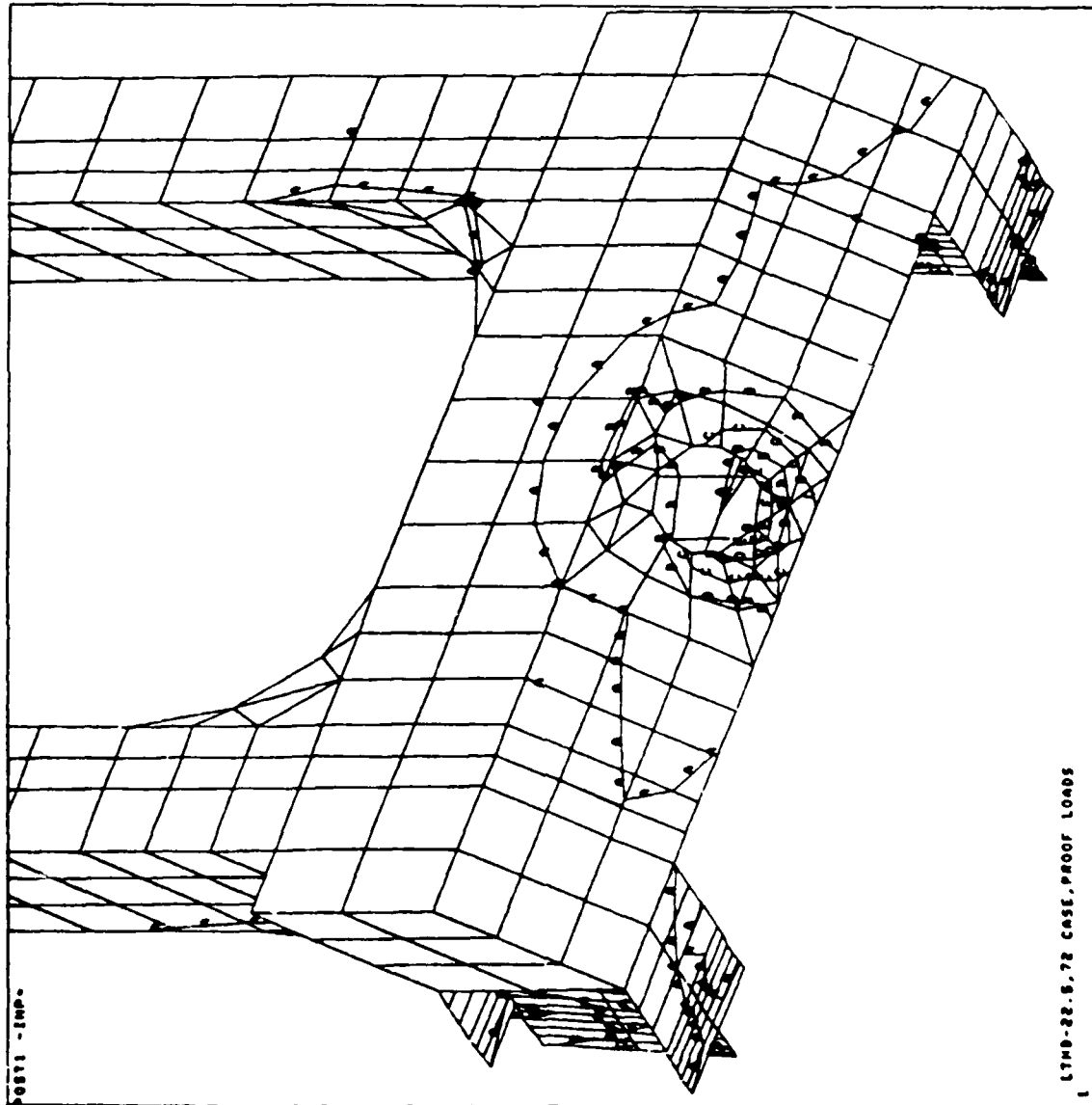
UPPER GIMBAL
MOUNTS

ANSYS 4.80
 JAN 20 1987
 17:09164
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.004
 SICE
 TOP
 NU=1
 VU=1
 ZU=-1
 DIST=88.7
 XF=53.8
 YF=35
 ZF=-7.83
 MIDDLE
 RX=165002
 MY=284
 A=27735
 B=55188
 C=82843
 D=110097
 E=137551



ANSYS 4.08
 JAN 20 1987
 16:25:07
 POST1 STRESS
 STEP=1
 ITEM=1
 TIME=.084
 SECT
 TOP
 KU=-1
 VU=-1
 ZU=1
 S1S7=25.7
 XF=63.9
 VF=25
 ZF=-7.93
 M100EM
 RM=165003
 RM=284
 A=27726
 B=56109
 C=22643
 D=110097
 E=137551

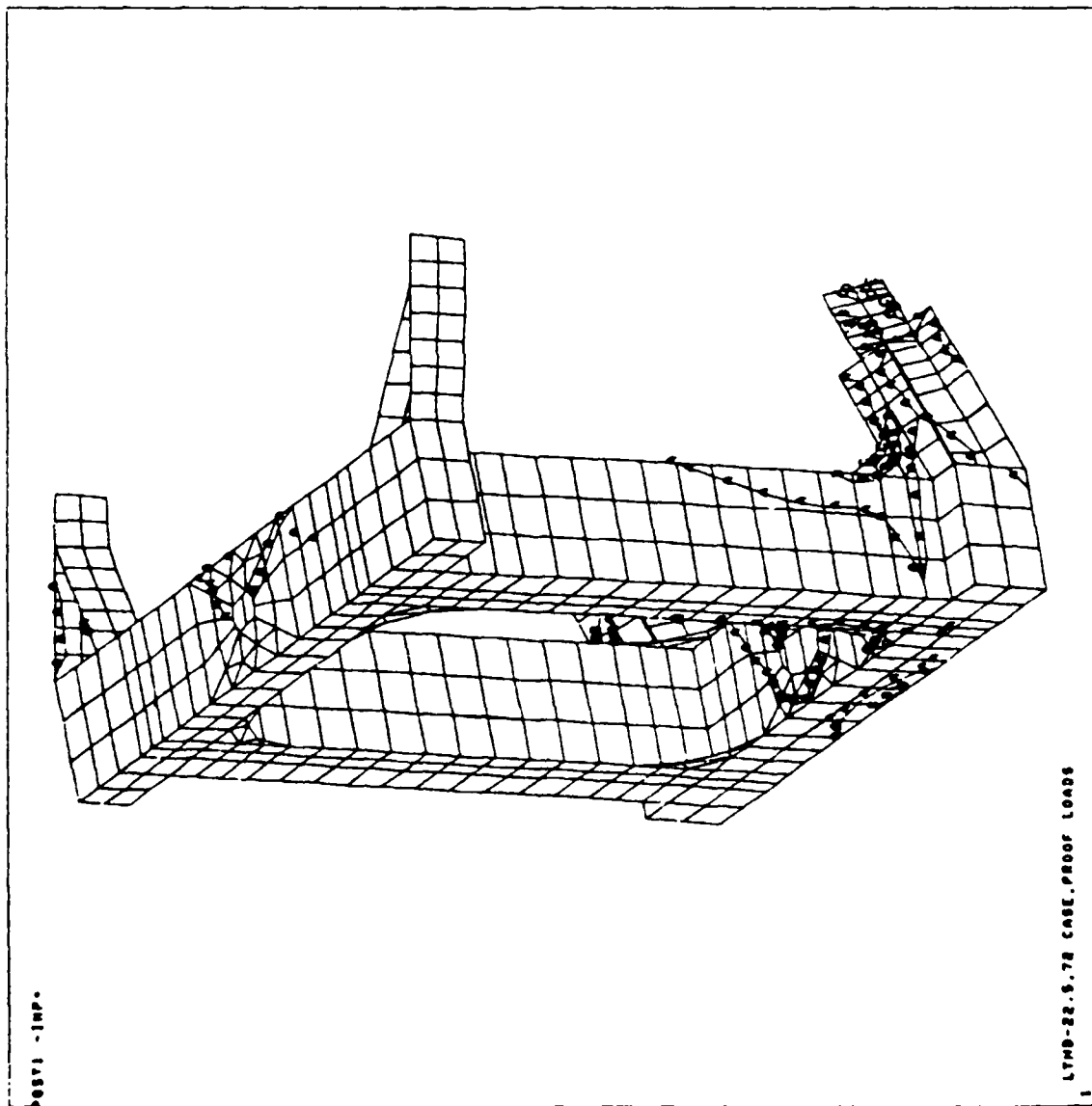




ANSYS 4.80
 JAN 20 1987
 10125107
 POST1 STRSS
 STEP=1
 ITER=1
 TIME=.004
 SIZ
 TOP
 ZOOM
 MU=-1
 VU=-1
 20=1
 1 0157-22.2
 2 07-60.3
 3 07-22.4
 4 27--14.2
 5 070-1.00
 6 10000
 7 10000
 8 10000
 9 10000
 10 10000
 11 10000
 12 10000
 13 10000
 14 10000
 15 10000
 16 10000
 17 10000
 18 10000
 19 10000
 20 10000
 21 10000
 22 10000
 23 10000
 24 10000
 25 10000
 26 10000
 27 10000
 28 10000
 29 10000
 30 10000
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 35 10000
 36 10000
 37 10000
 38 10000
 39 10000
 40 10000
 41 10000
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 83 10000
 84 10000
 85 10000
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 87 10000
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 89 10000
 90 10000
 91 10000
 92 10000
 93 10000
 94 10000
 95 10000
 96 10000
 97 10000
 98 10000
 99 10000
 100 10000

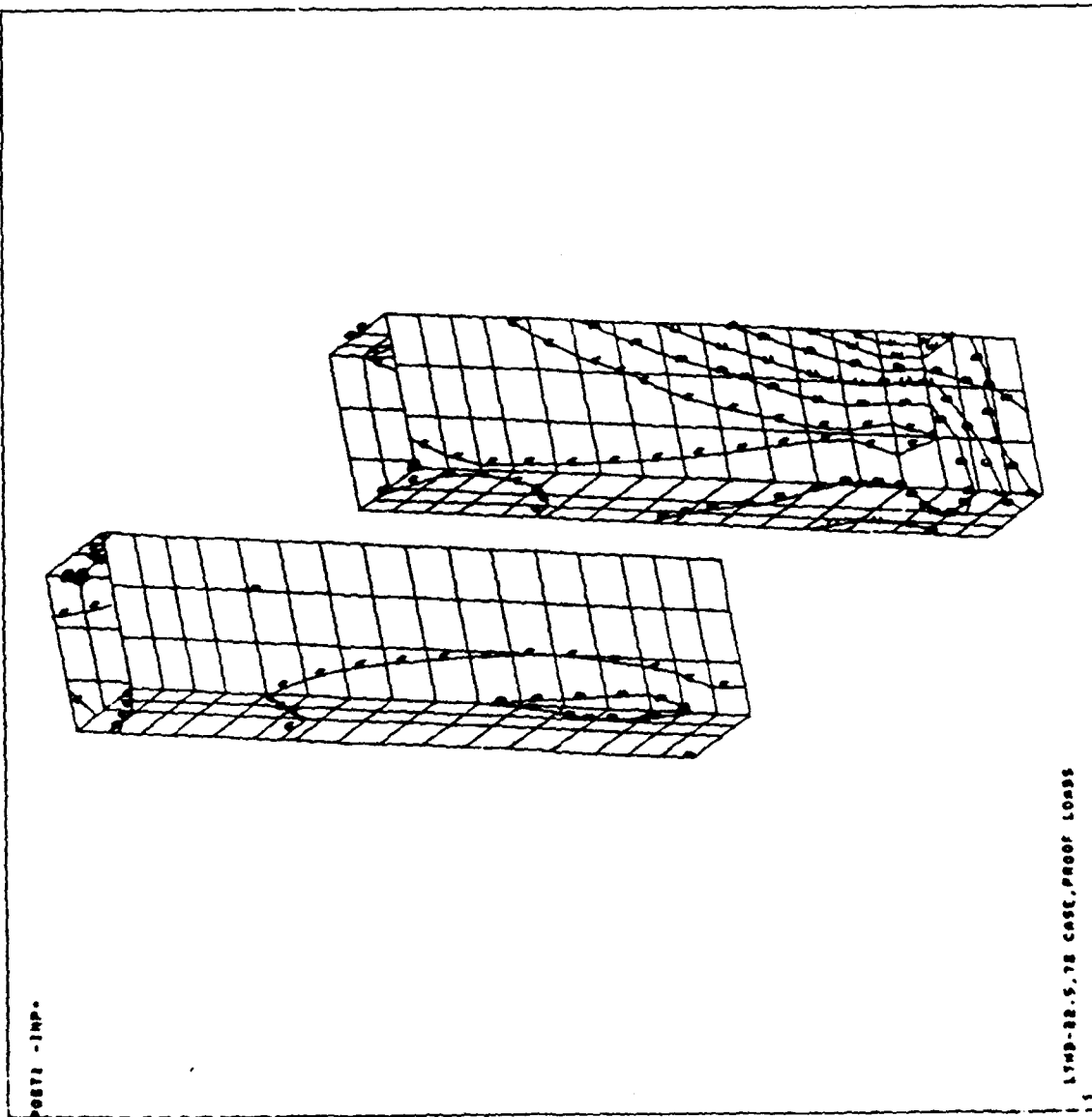
LTMD-22.5, 72 CASE, PROOF LOADS

ANSYS 4.20
 JAN 20 1987
 17105134
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.004
 SICE
 TOP
 KU=1
 VU=1
 ZU=1
 DIST=34.9
 KF=57.3
 VF=33.9
 ZF=-7.62
 MIDDEN
 MX=165003
 MY=284
 A=27735
 B=55189
 C=82643
 D=110097
 E=137551

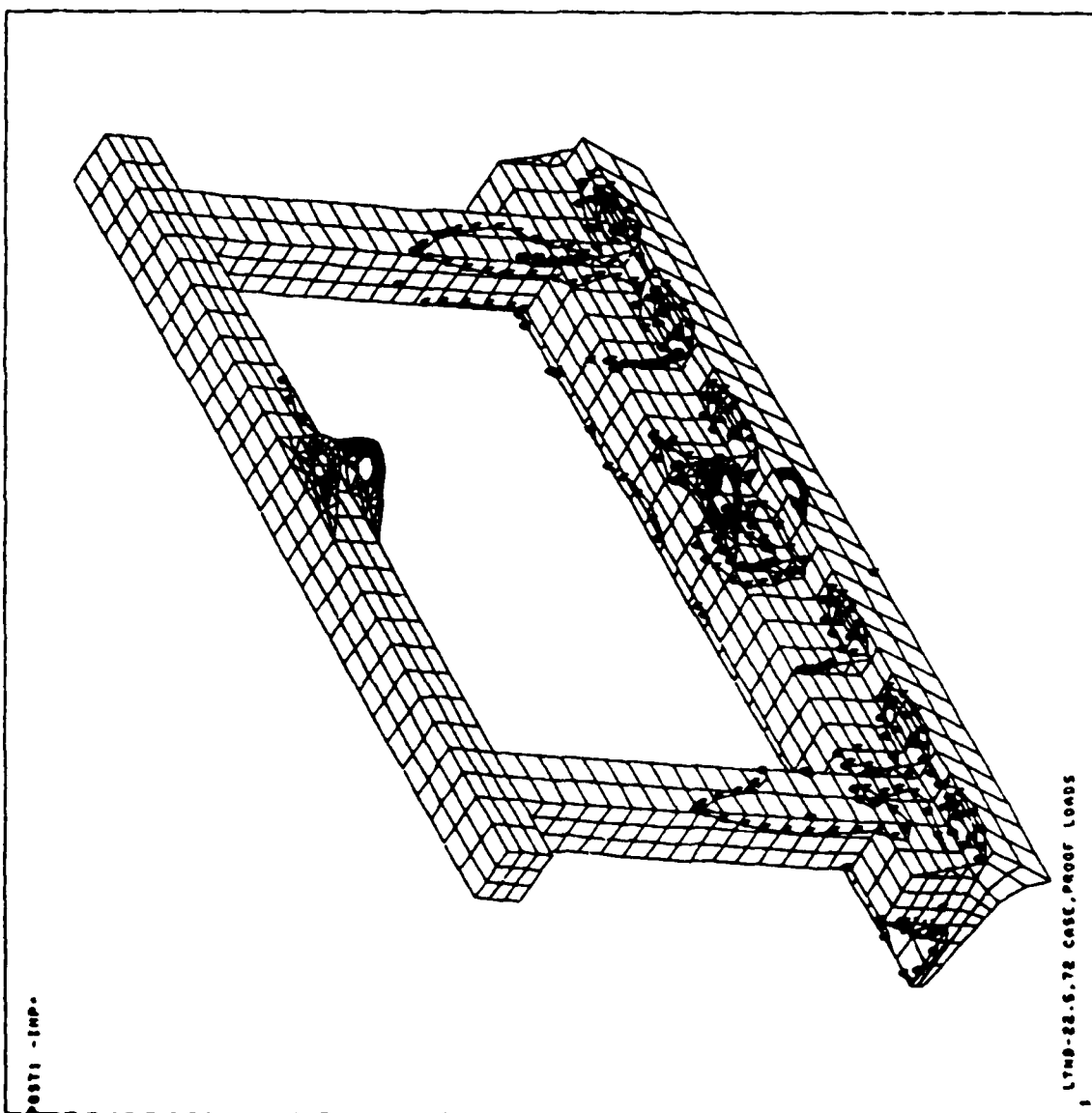


ANSYS 4.88
 JAN 29 1987
 17:30:12
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.004
 SLOC
 TOP
 ZOOM
 KU=1
 VU=1
 ZU=1
 D157=32.2
 KF=53
 VF=33.8
 ZF=-3.5
 KRT0=1.18
 VRT0=1.14
 M188CM
 MX=51658
 MY=2747
 P=9738
 Q=16719
 C=23786
 D=38693
 E=37680
 F=44687

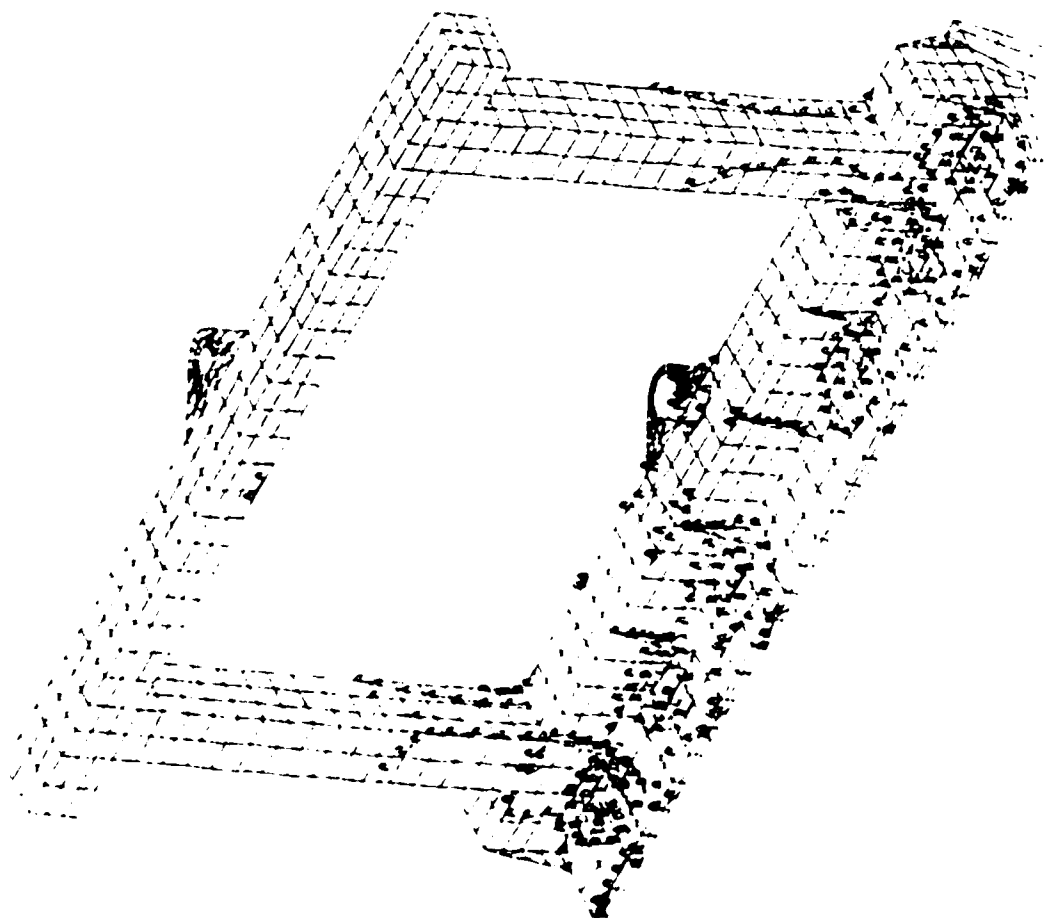
GIMBAL
 VERTICAL
 BOX BEAMS



ANSYS 4.28
 JAN 20 1987
 17112152
 POST1 STRESS
 STEP=1
 TIME=1
 TIME=0.004
 SLOC
 TOP
 KU=1
 VU=1
 ZU=-1
 9187-88.2
 XF-54.2
 YF-26.4
 ZF-4.85
 MIDDLE
 RX-149027
 RM=400
 A=25170
 B=48042
 C=74714
 D=99486
 E=124258



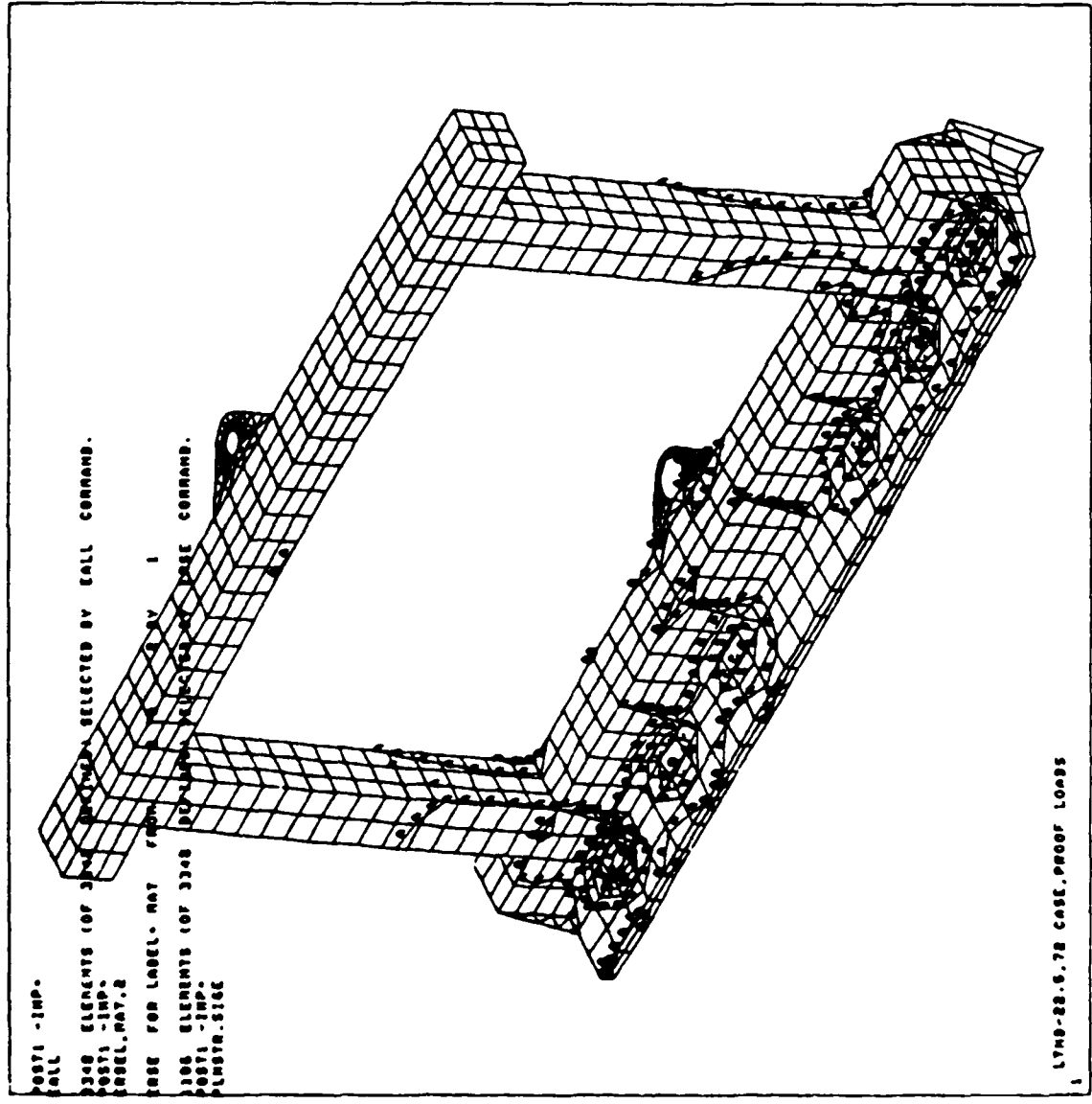
ARSYS 4.20
 JAN 20 1987
 16:12:27
 POST1 STRESS
 STEP=1
 TIME=1
 TIME= .004
 SIZE
 TOP
 RU=1
 VU=1
 ZU=1
 DIS=56.6
 SF=51.2
 VF=27.7
 ZI=4.17
 MIDDLE
 PK=149027
 RM=400
 A=25170
 B=40042
 C=74714
 D=99486
 E=124258



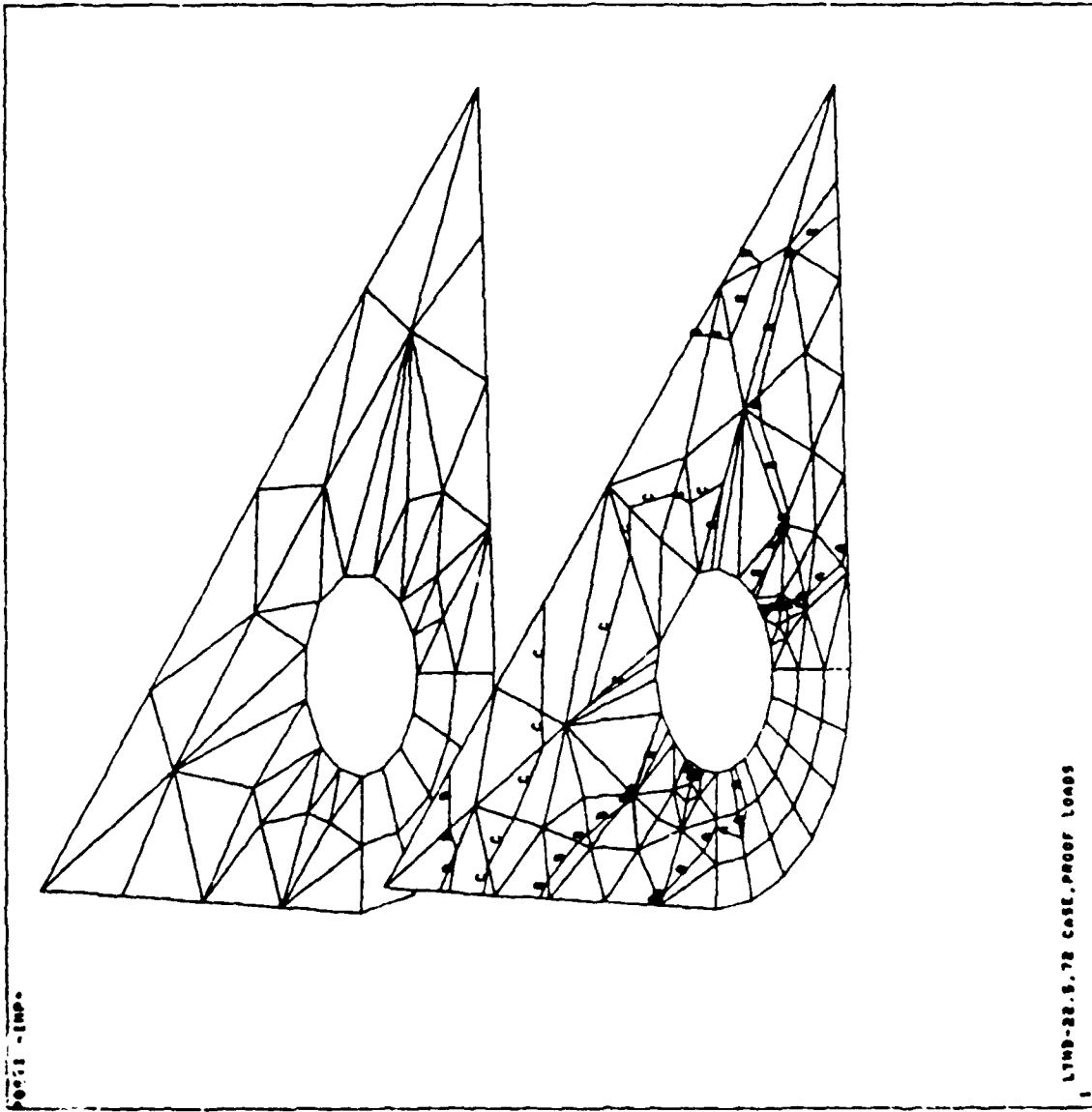
POST1 -IMP.

LTNG=22.5, 72 CASE, PRCL, LONES

ANSYS 4.80
 JAN 20 1987
 10:28:27
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.004
 SLOC
 TOP
 ZU=1
 VU=1
 ZU=1
 9167-60.6
 37-51.2
 37-27.7
 27-4.17
 HIDDEN
 ON=140087
 ON=400
 A=25170
 B=49042
 C=74714
 D=99406
 E=124250



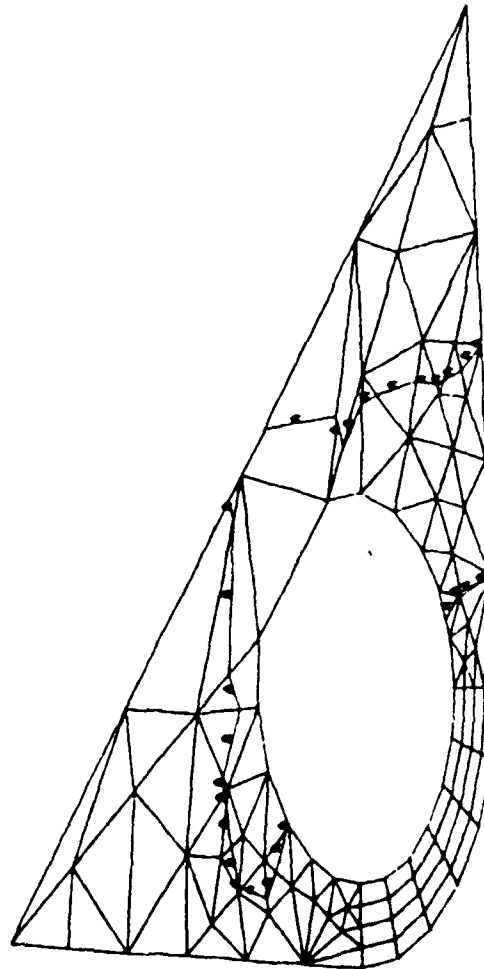
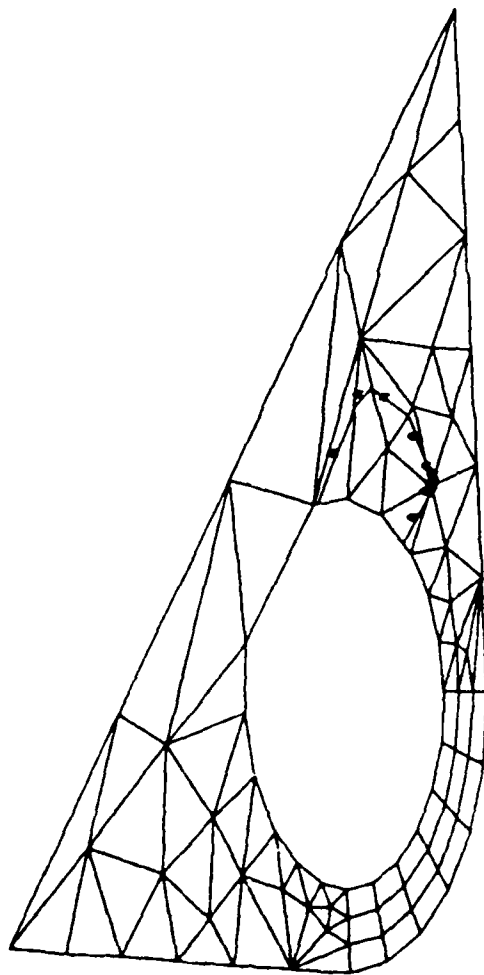
ANSYS 4.20
 JAN 20 1987
 17:24:23
 POST1 STRESS
 STEP=1
 IYEA=1
 TIME=.004
 SICE
 TOP
 ZOOM
 RV=-1
 VU=-1
 ZU=1
 DIST=8.74
 RF=3.4
 VF=48
 ZF=7.66
 XRT0=1.18
 YRT0=1.14
 HIDDEN
 MH=16884
 MH=619
 A=2816
 B=7316
 C=10716
 D=14116



TOP TABS

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CONFIDENTIAL

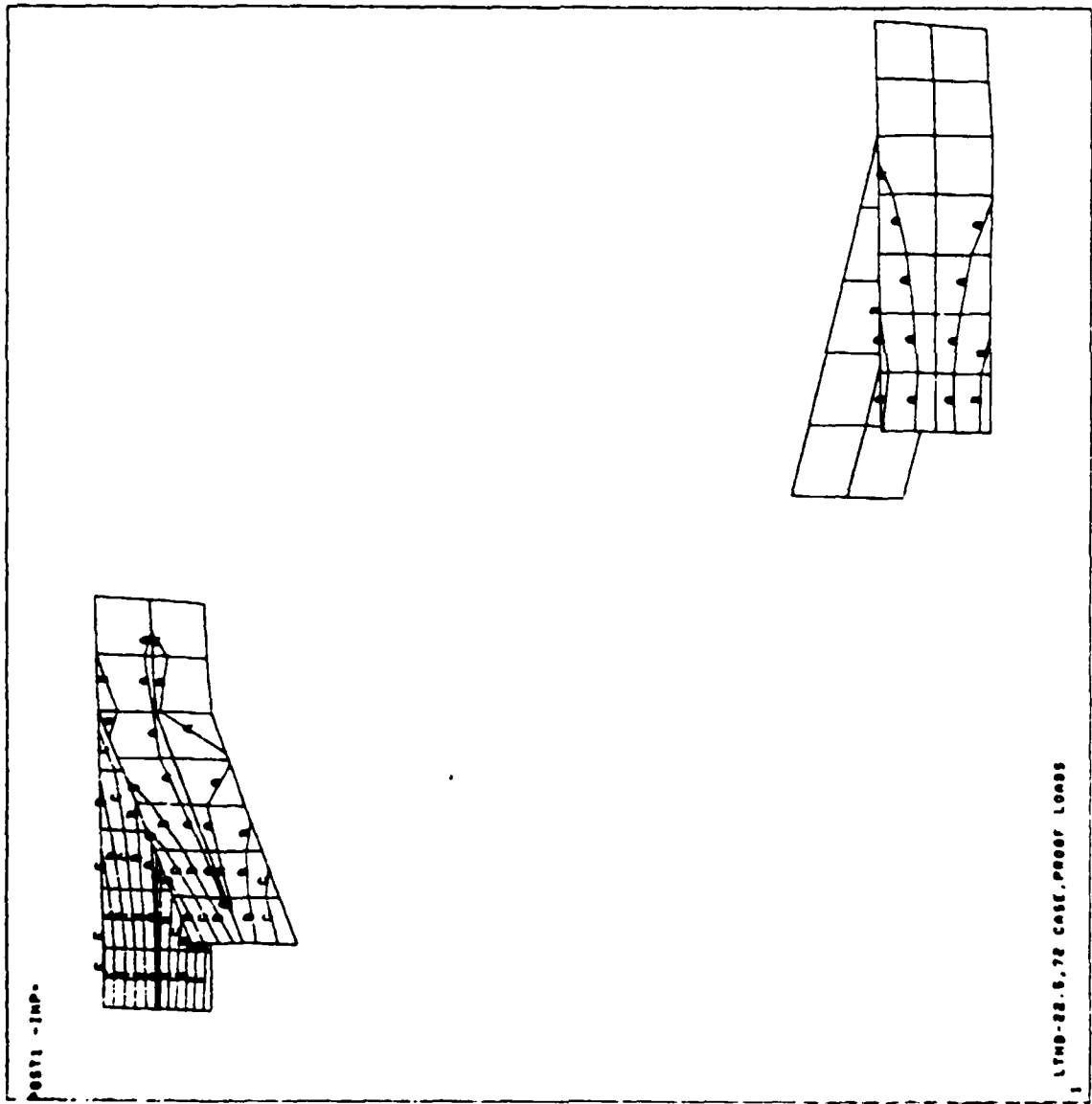


LYMB-22.5.72 CASE PROOF LOADS

550

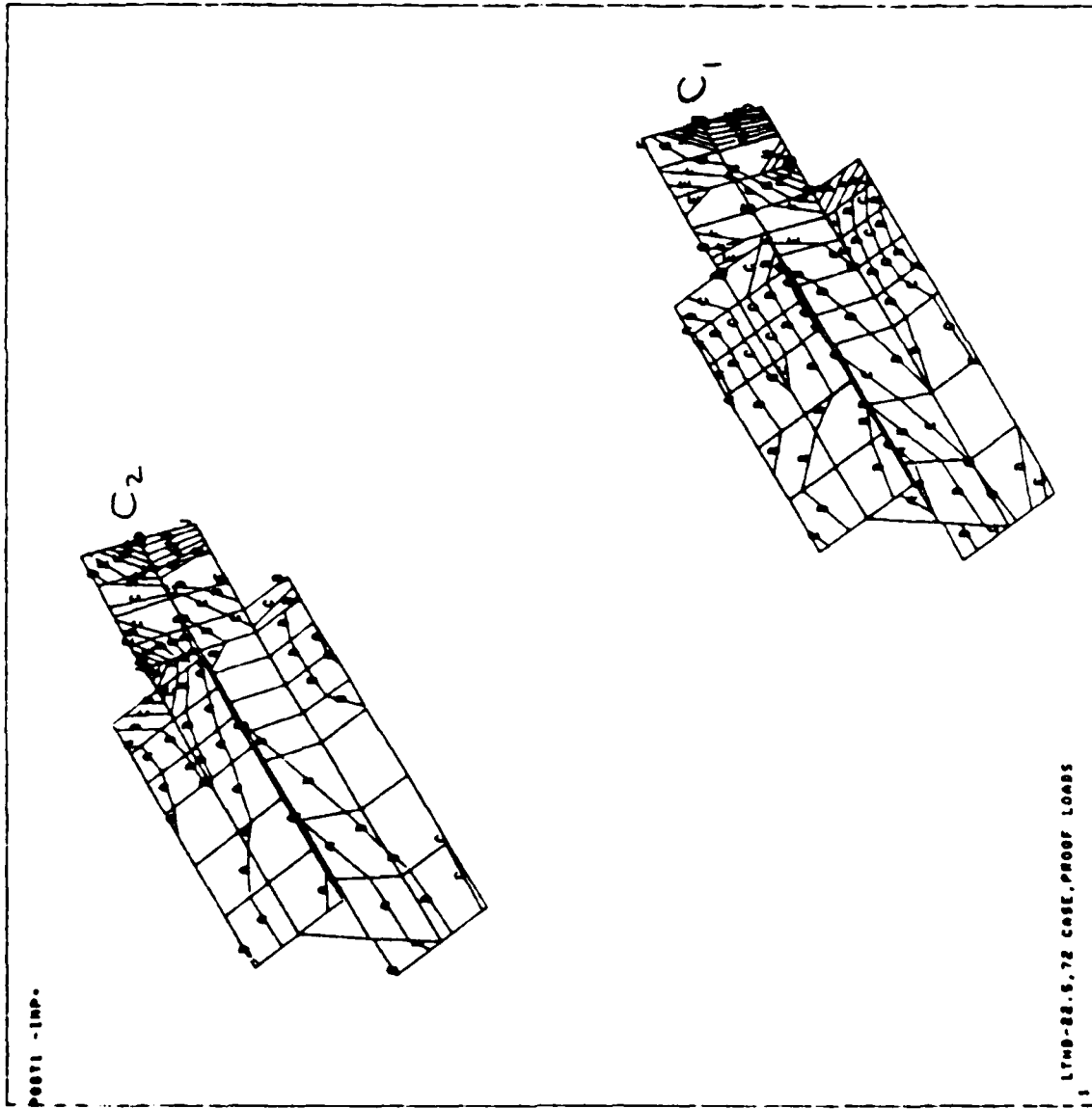
59

ANSYS 4.20
 JAN 20 1987
 17:27:50
 POST1, STRESS
 STEP=1
 ELEM=1
 TIME=0.004
 SLOC
 TOP
 ZOOM
 KU=1
 VU=1
 ZN=1
 DIST=18.7
 MF=67.1
 VF=63.3
 ZF=13.2
 M70=1.18
 M70=1.14
 M100EN
 M1=48146
 M1=284
 R=7120
 R=13958
 C=20796
 B=27634
 E=34472
 F=41310



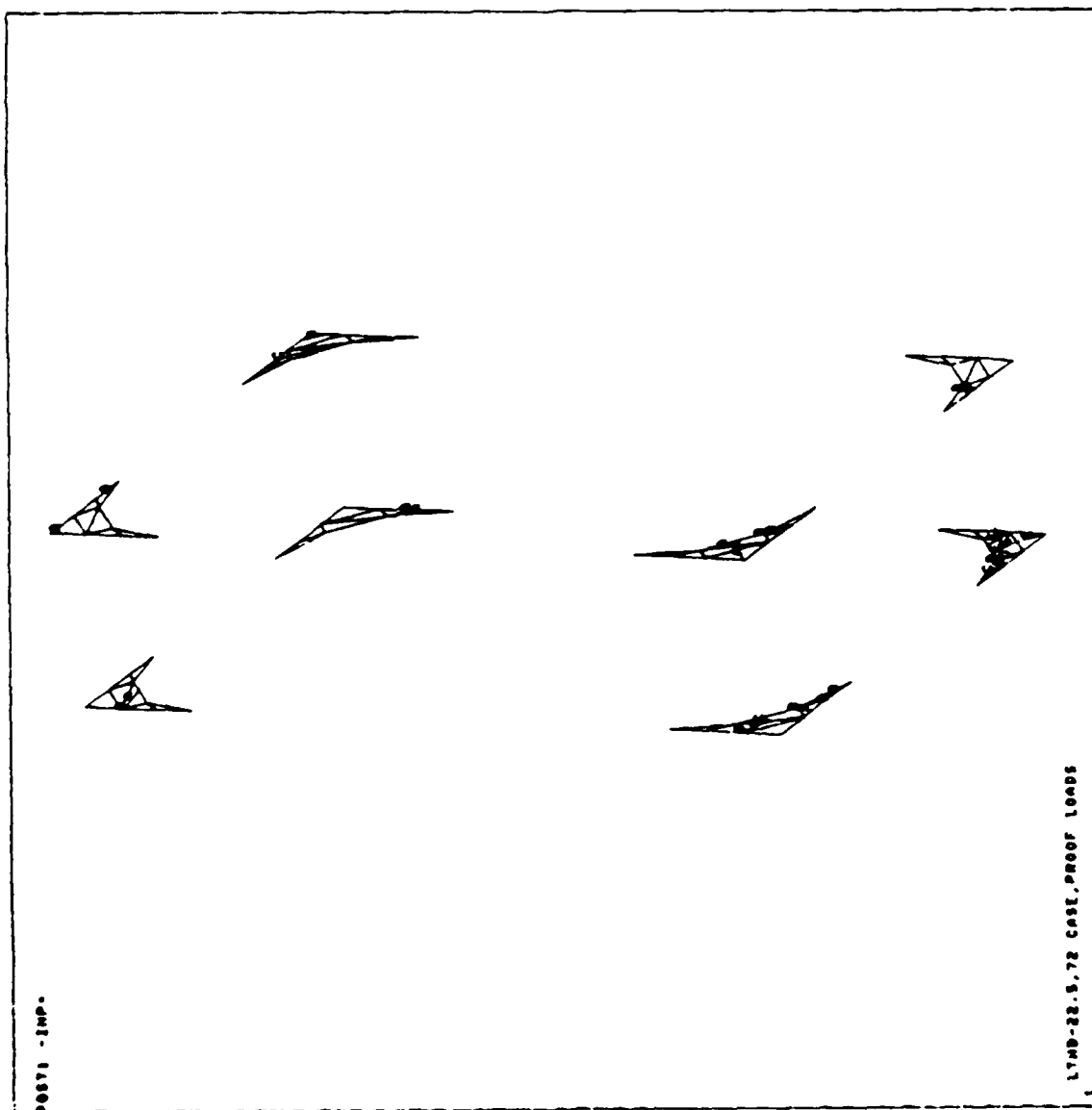
TOP ARMS,
GIMBAL

ANSYS 4.00
 JAN 20 1987
 17:26:00
 POST11 STRESS
 STEP=1
 ITER=1
 TIME=.004
 SLOC
 TOP
 ZOOM
 XU=1
 YU=1
 ZU=1
 B157.80
 XF=57.3
 YF=14.2
 ZF=-12.7
 XRT0=1.18
 YRT0=1.14
 MIDDEN
 MM=163566
 MM=6289
 A=15458
 B=24666
 C=33872
 D=43070
 E=52286
 F=61493
 G=70700
 H=79907
 I=89114
 J=98321
 K=107520
 L=116735
 M=125942
 N=135140
 O=144356



LOWER
 GIMBAL
 ARMS

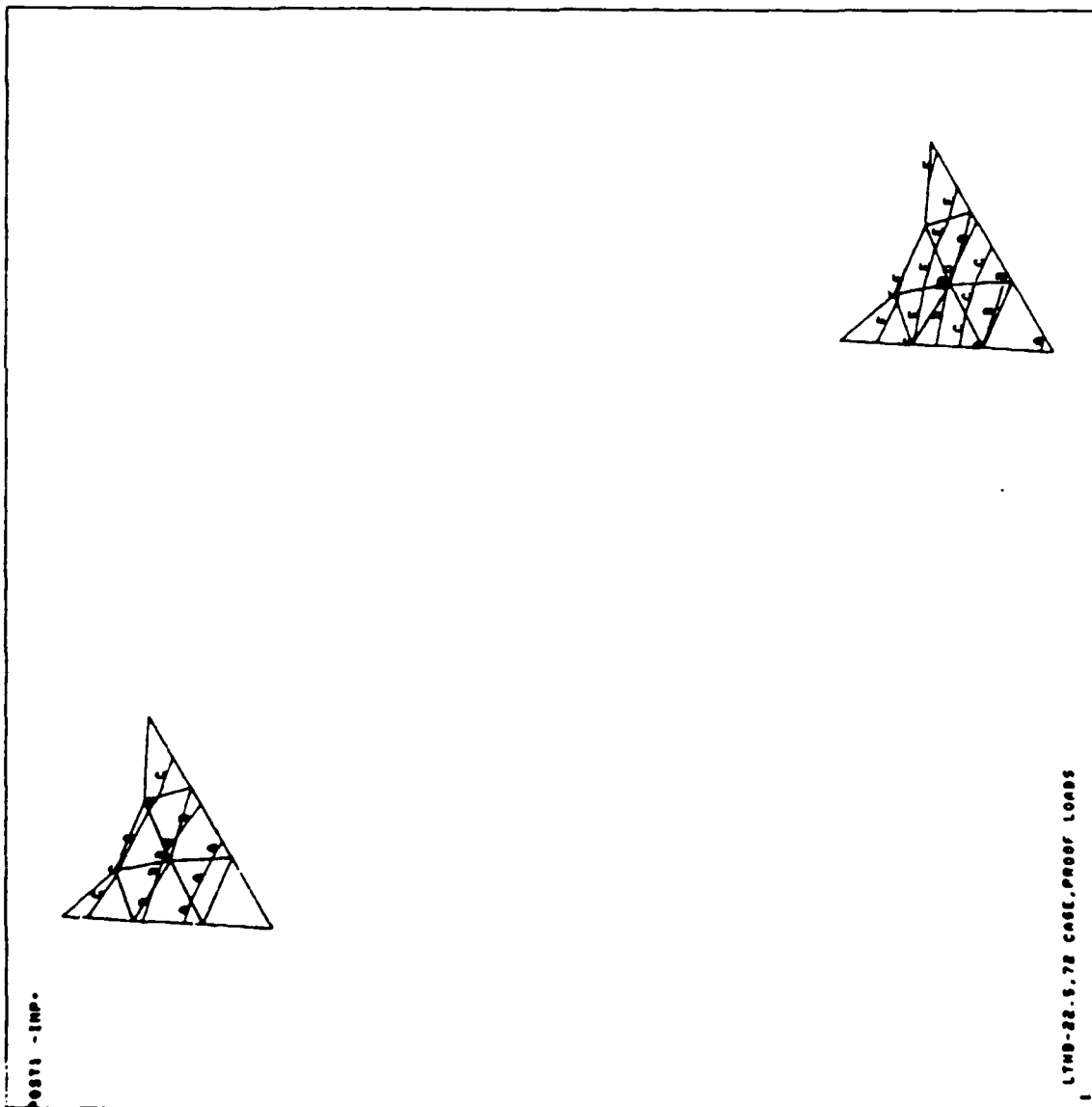
ANSYS 4.20
 JAN 20 1987
 17:28:36
 POST1, STRESS
 STEP=1
 ITER=1
 TIME=.004
 S10E
 TOP
 ZOOM
 RU=1
 VU=1
 VU=1
 0167-28.5
 VF=53
 VF=33.6
 ZF=-3.5
 MOTO=1.18
 VOTO=1.14
 MIDDEN
 M1=45780
 M2=3249
 A=8408
 B=15473
 C=21537
 D=27601
 E=33655
 F=38728



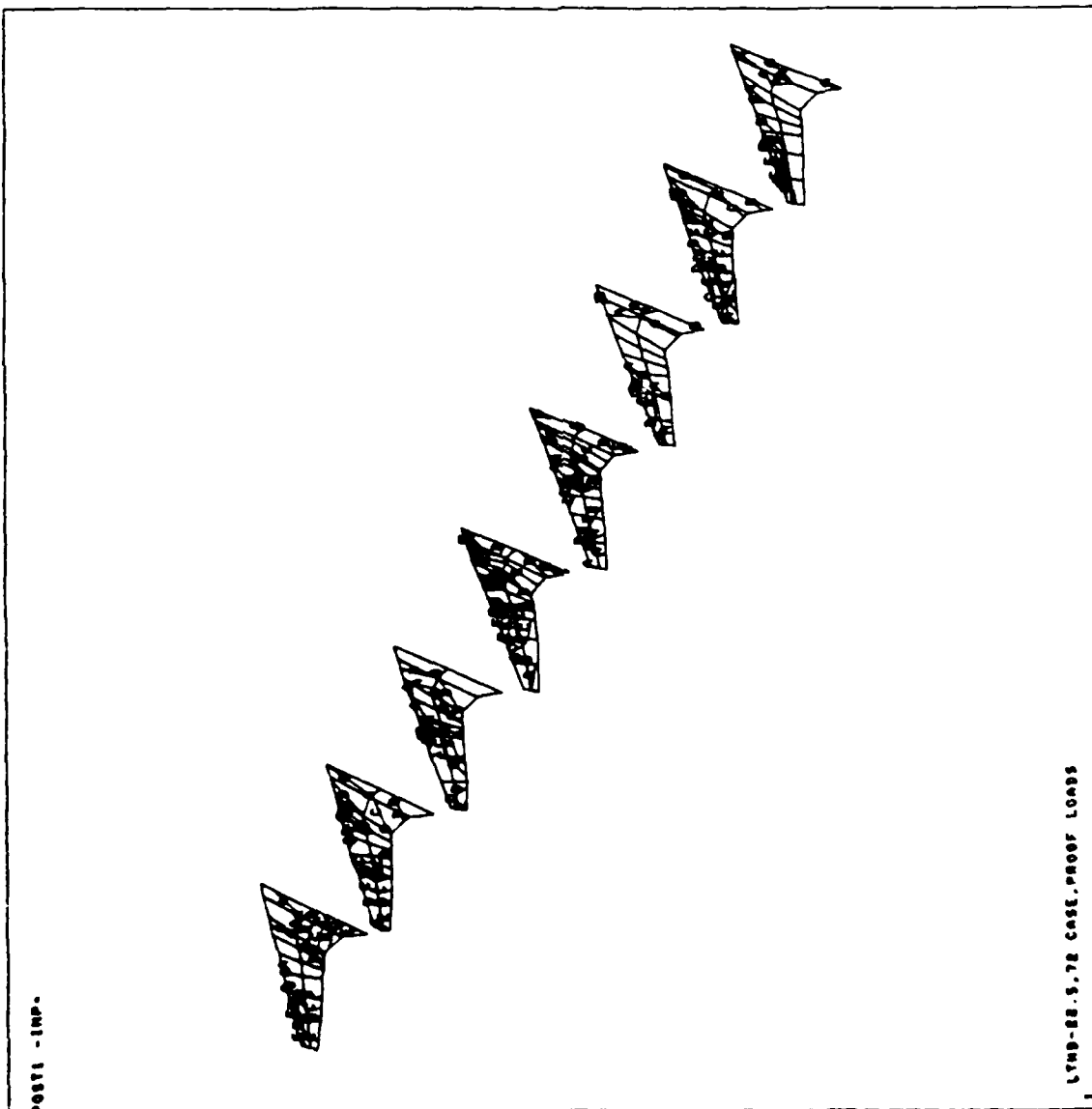
GIMBAL
 REINFORCING-
 TABS

ANSYS 4.20
 JAN 20 1987
 17132107
 POST1 STRESS
 STEP=1
 TIME=1
 TIME=.084
 SIZE
 TOP
 ZORR
 RU=1
 VU=1
 ZU=1
 D187-14.8
 XF-56.8
 YF-17.4
 ZF-8.97
 WRT0-1.18
 VRT0-1.14
 MIDDE
 MM-110534
 MM-18647
 A-31772
 B-44000
 C-58027
 D-71154
 E-84281
 F-97402

GIMBAL
 LOWER
 ARM
 REINFORCING
 TABS

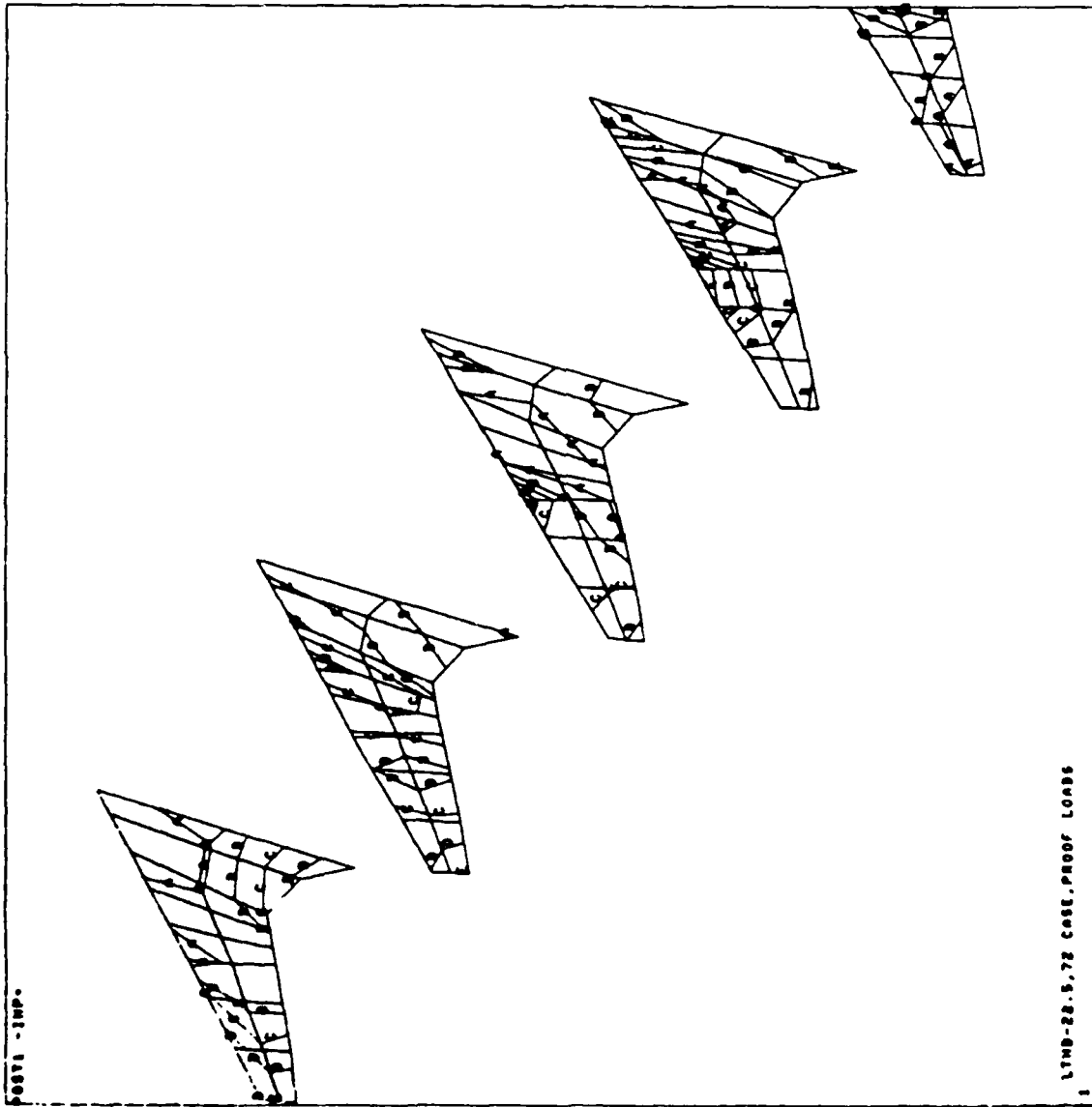


ANSYS 4.20
 JAN 20 1987
 17:38:22
 POST1, STESS
 STEP=1
 ITER=1
 TIME=.004
 SLOC
 TOP
 ZOOM
 XZ=1
 YZ=1
 ZY=1
 D167=57.8
 XZ=52.3
 YZ=1.87
 ZY=5.33
 KATO=1.18
 VATO=1.14
 MIBEN
 RM=44078
 RM=8488
 A=8487
 B=14369
 C=20311
 D=28253
 E=32195
 F=38137

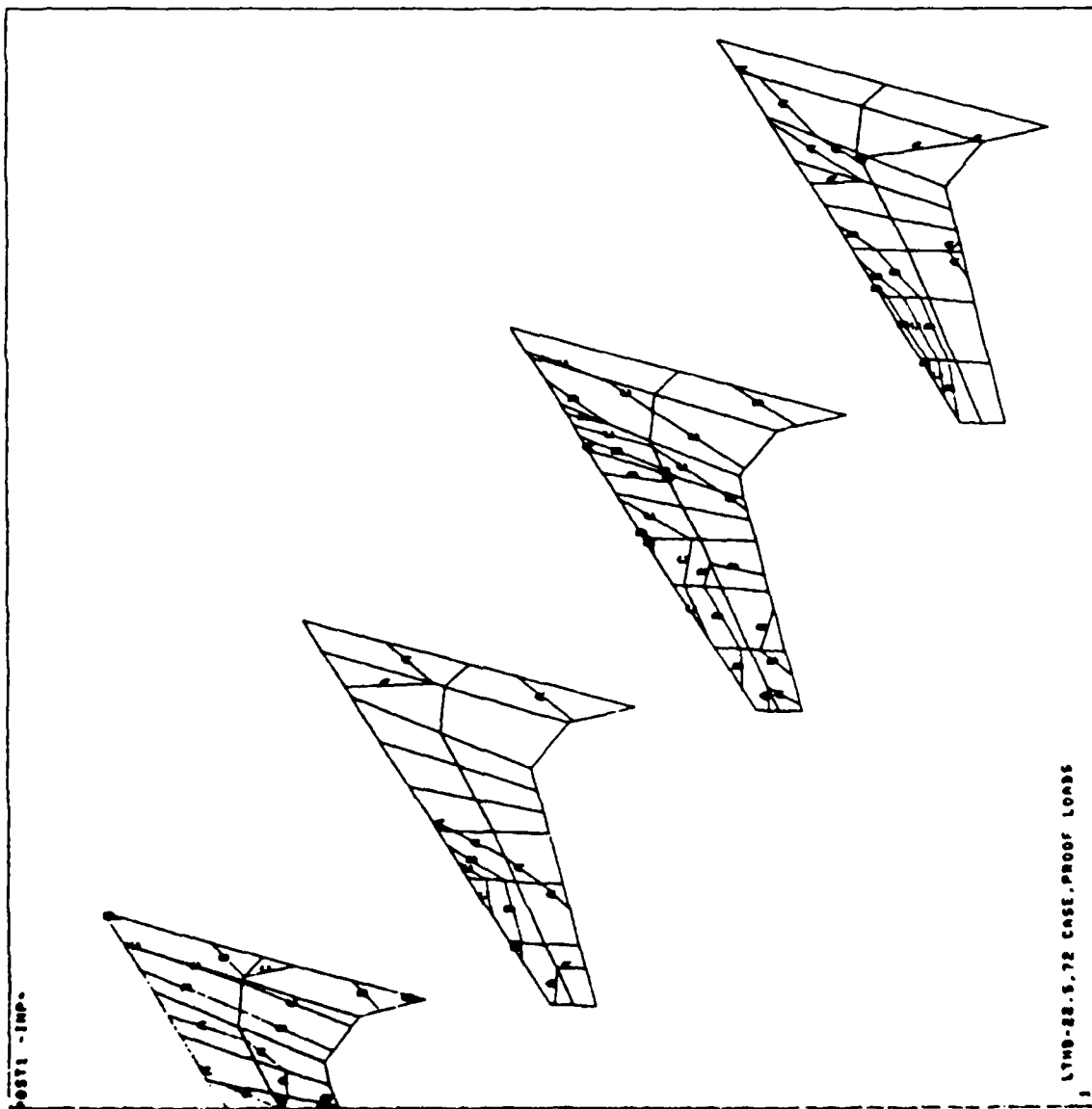


PLATFORM
 SPADE
 REINFORCING-
 VERTICAL
 ELEMENTS

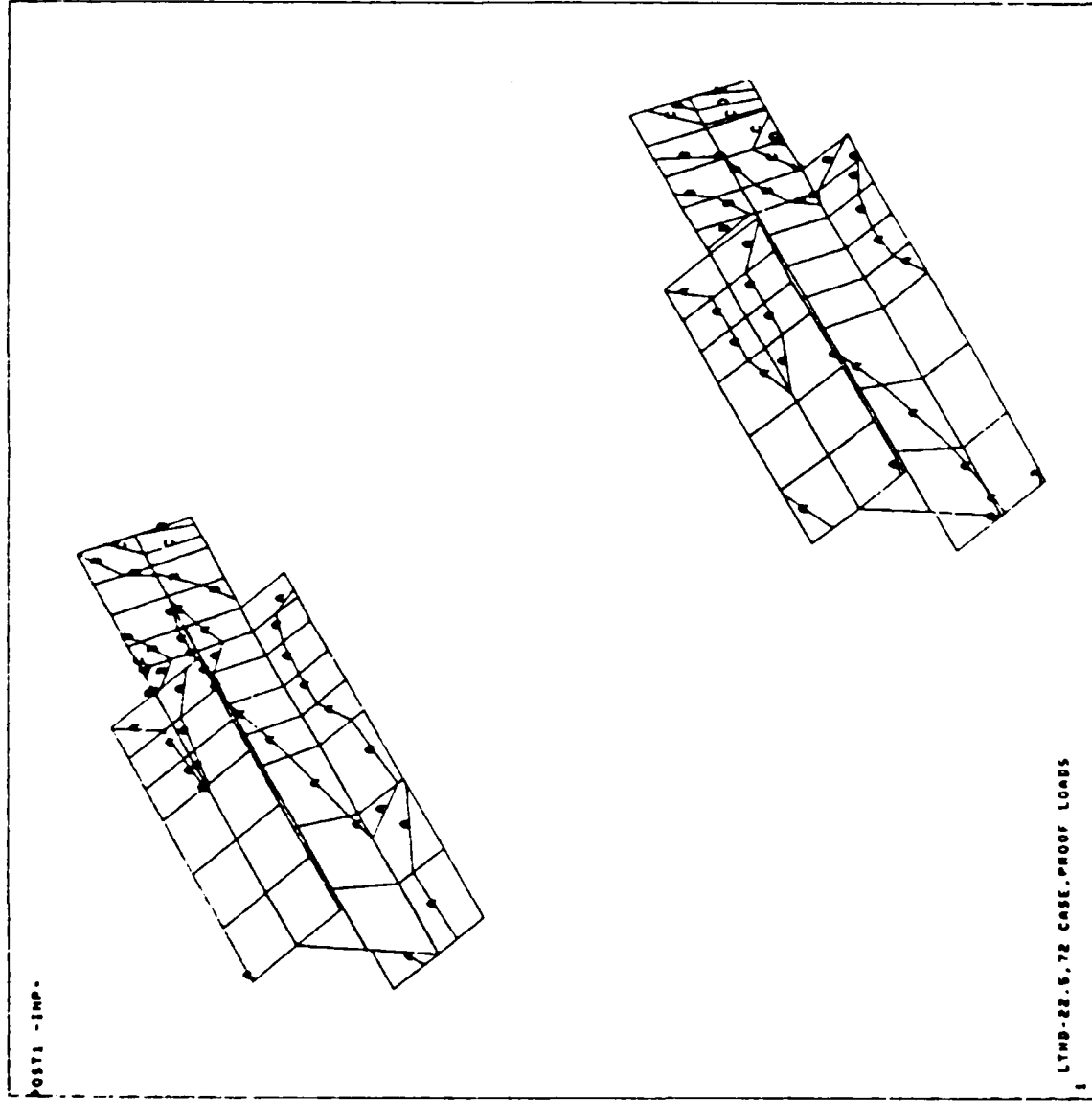
ANSYS 4.80
 JAN 20 1987
 17:38:33
 POST1 STRESS
 STEP=1
 ITEM=1
 TIME=.084
 SLOE
 TOP
 ZOOM
 KU=1
 VU=1
 ZU=1
 0 DIST=49.8
 8 XF=35.1
 8 VF=8.11
 8 ZF=14.8
 XRTQ=1.95
 VRTQ=2.4
 MIDSEN
 MK=44676
 MM=0
 A=9418
 B=16350
 C=22282
 D=30214
 E=37146



ANSYS 4.20
 JAN 20 1987
 17:37:21
 POST1 STRESS
 STEP=1
 ITEM=1
 TIME=.004
 SICE
 TOP
 ZOOM
 KU=1
 VU=1
 ZU=1
 0 DIS=39.4
 0 KF=75.4
 0 VF=-11.8
 0 ZF=-8.17
 XRTG=1.98
 YRTG=2.41
 MIBDEN
 MM=20179
 MM=2489
 A=9418
 B=18350
 C=22282
 D=30214
 E=31145

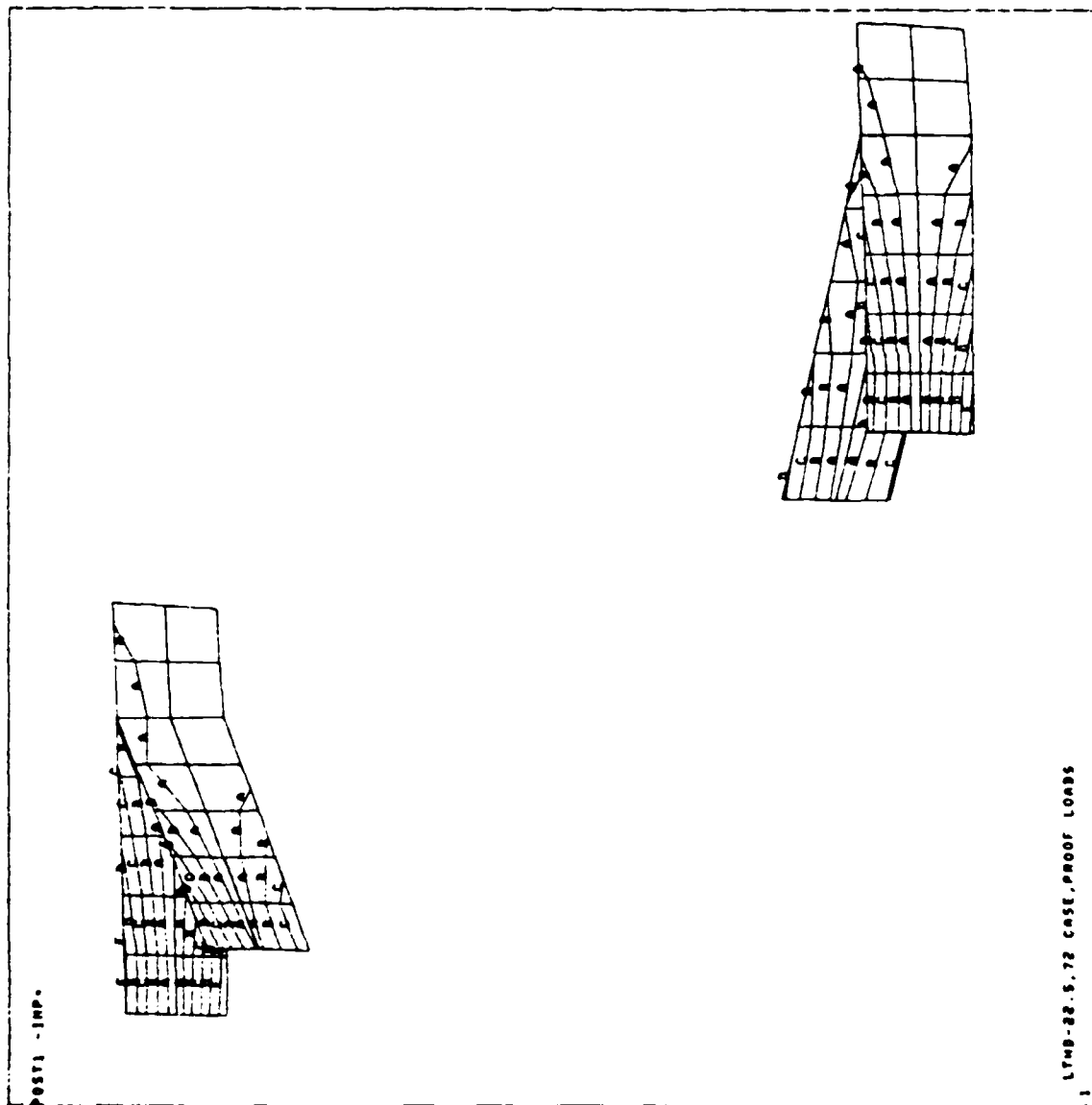


ANSYS 4.08
 JAN 21 1987
 1103110
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.237
 SDOF
 TOP
 ZOOM
 RV=1
 VU=1
 ZU=1
 DIST=21
 XF=57.3
 YF=14.2
 ZF=-12.7
 XROT=1.3
 YROT=1.2
 MIDDLE
 MX=76539
 MY=4261
 A=18306
 B=28353
 C=40400
 D=52447
 E=64494



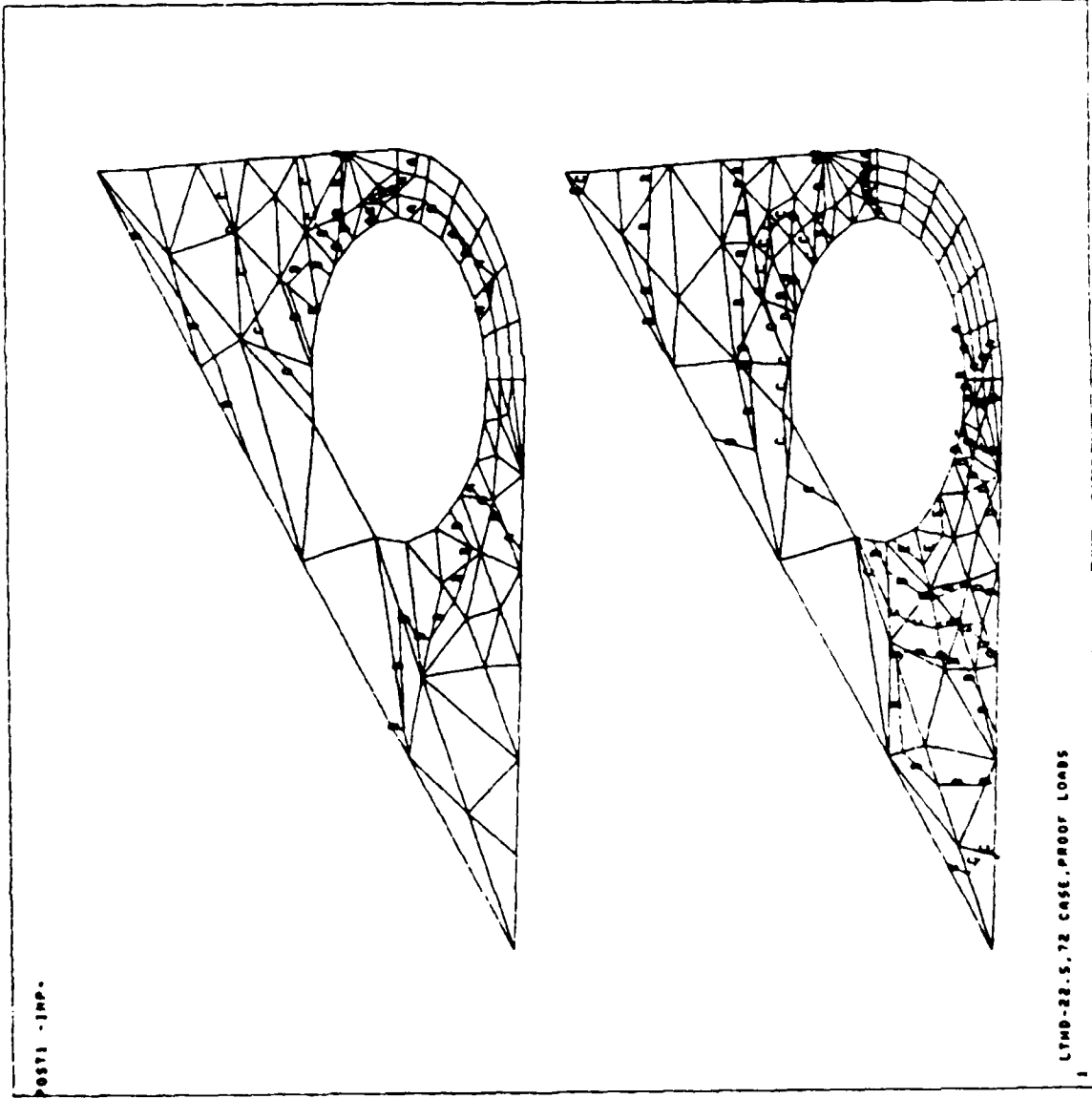
LOWER
 GIMBAL
 ARMS

ANSYS 4.20
 JAN 21 1987
 11:02:22
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.237
 SICE
 TOP
 2004
 XU=1
 YU=1
 ZU=1
 DIST=20.6
 XF=57.1
 YF=53.3
 ZF=13.2
 XRT0=1.3
 YRT0=1.2
 MIDEM
 PK=101923
 AN=8491
 A=2386
 B=39574
 C=55162
 D=70750
 E=86338



TOP
 GMBAL
 ARMS
 561

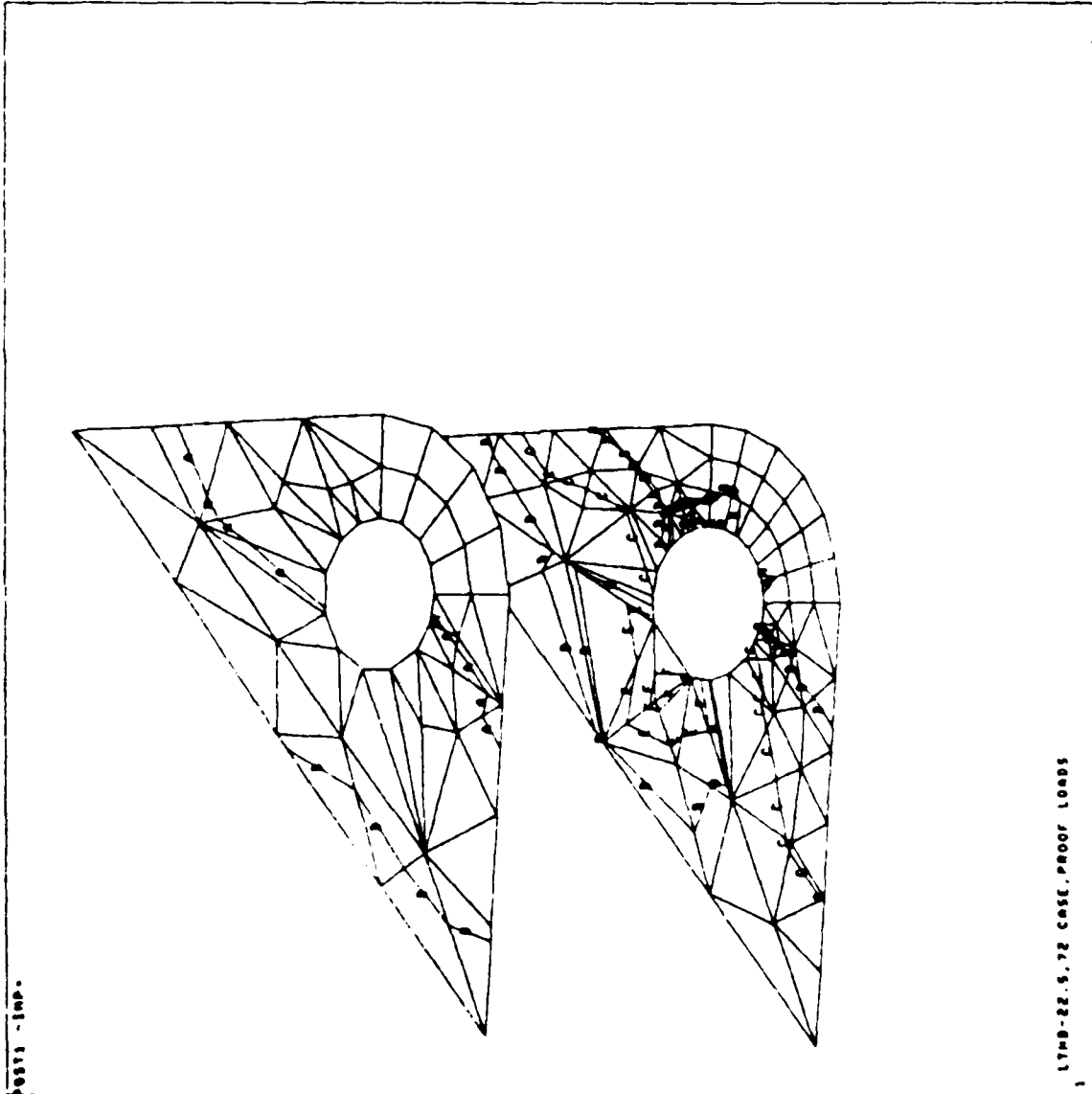
ANSYS 4.20
 JAN 21 1987
 10:59:40
 POST1 STRSS
 STEP=1
 ITER=1
 TIME=.237
 SLOC
 TOP
 ZOOM
 RU=1
 VU=1
 2U=-1
 F BIST-9.76
 F XF-62
 F VF-13.4
 F ZF-9.35
 XRTG-1.3
 VRTG-1.2
 MIDDEN
 RH-28018
 RH-1018
 A-4090
 B-9755
 C-14822
 D-19288
 E-24054



BOTTOM
 TABS

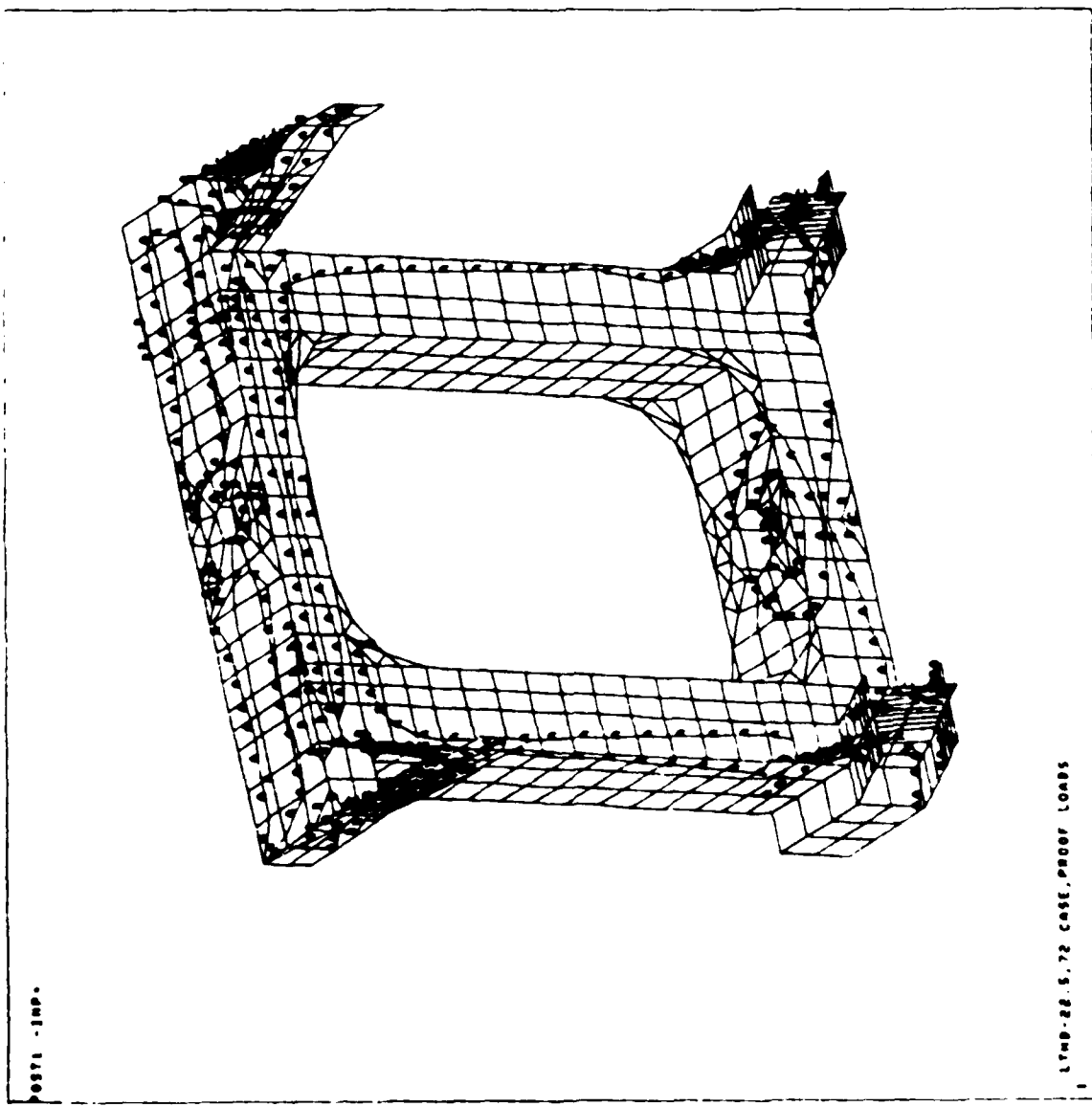
562

ANSYS 4.20
 JAN 21 1987
 10164104
 POST11 STRESS
 STEP=1
 ITER=1
 TIME=.237
 SICE
 TOP
 ZOOM
 RU=1
 VU=1
 ZU=-1
 0 0197-9.72
 0 01-41
 0 01-48.1
 0 27-4.46
 VR50=1.2
 MDOEN
 MH=11016
 MH=227
 A=2013
 B=3000
 C=5587
 D=7374
 E=9161
 F=10040

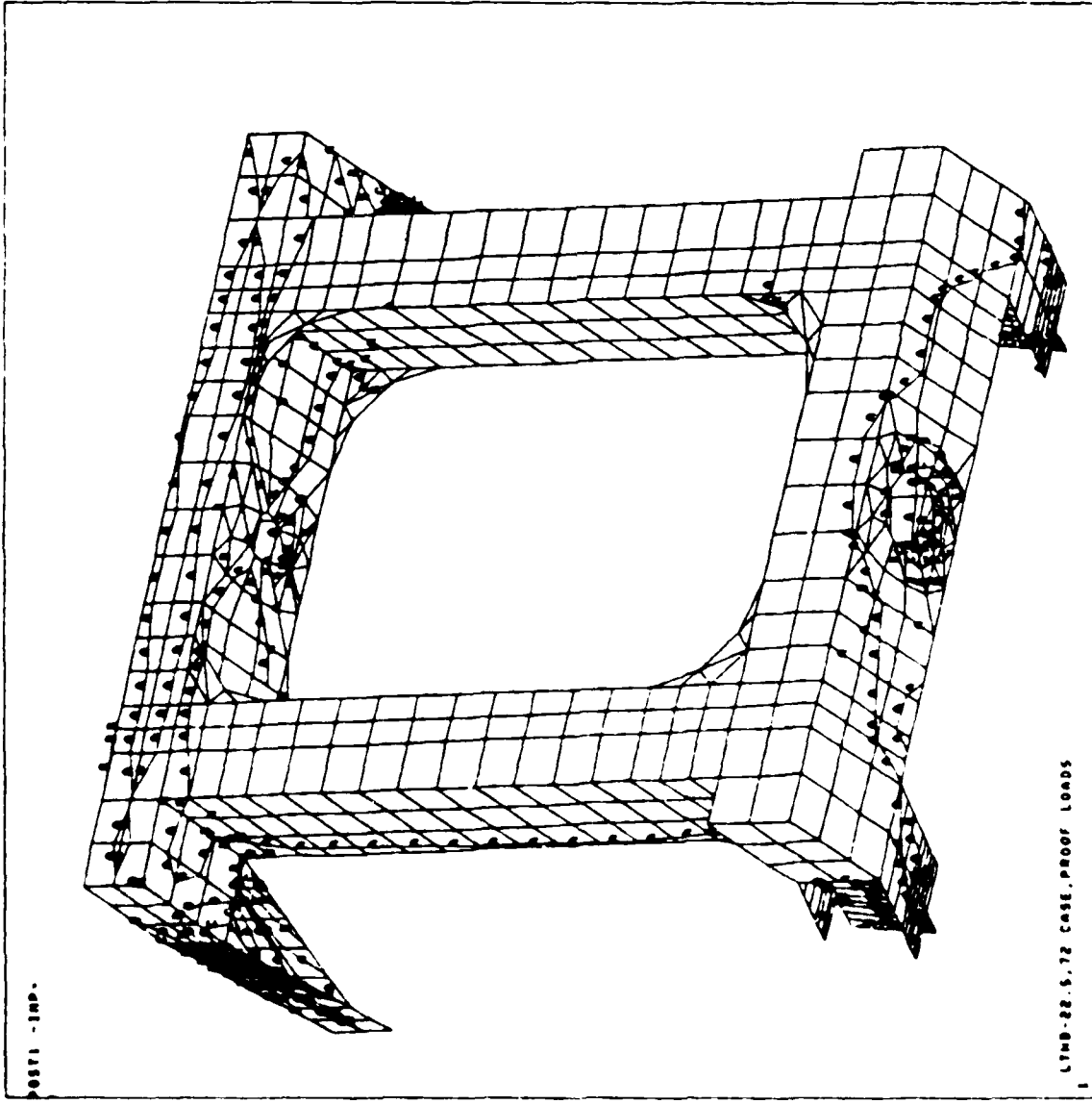


TOP
 TABS
 5/2

ANSYS 4.20
 JAN 21 1987
 10130115
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.237
 SICE
 TOP
 RV=1
 VU=1
 ZU=1
 E 8157-34.0
 C 46-67.3
 S 46-33.0
 E 26--7.62
 M100EN
 RM-95217
 RM-840
 A-16560
 B-22800
 C-48020
 D-03750
 E-79400



ANSYS 4.20
 JAN 21 1987
 0162154
 POST1 STRESS
 STEP=1
 LAYER=1
 TIME=.037
 SIZE
 TOP
 KU=-1
 VU=-1
 ZU=1
 0157-20.7
 XF=53.9
 VF=35
 ZF=-7.93
 MIDDEN
 RX=85217
 RM=840
 A=16568
 B=32200
 C=48020
 D=63758
 E=79400



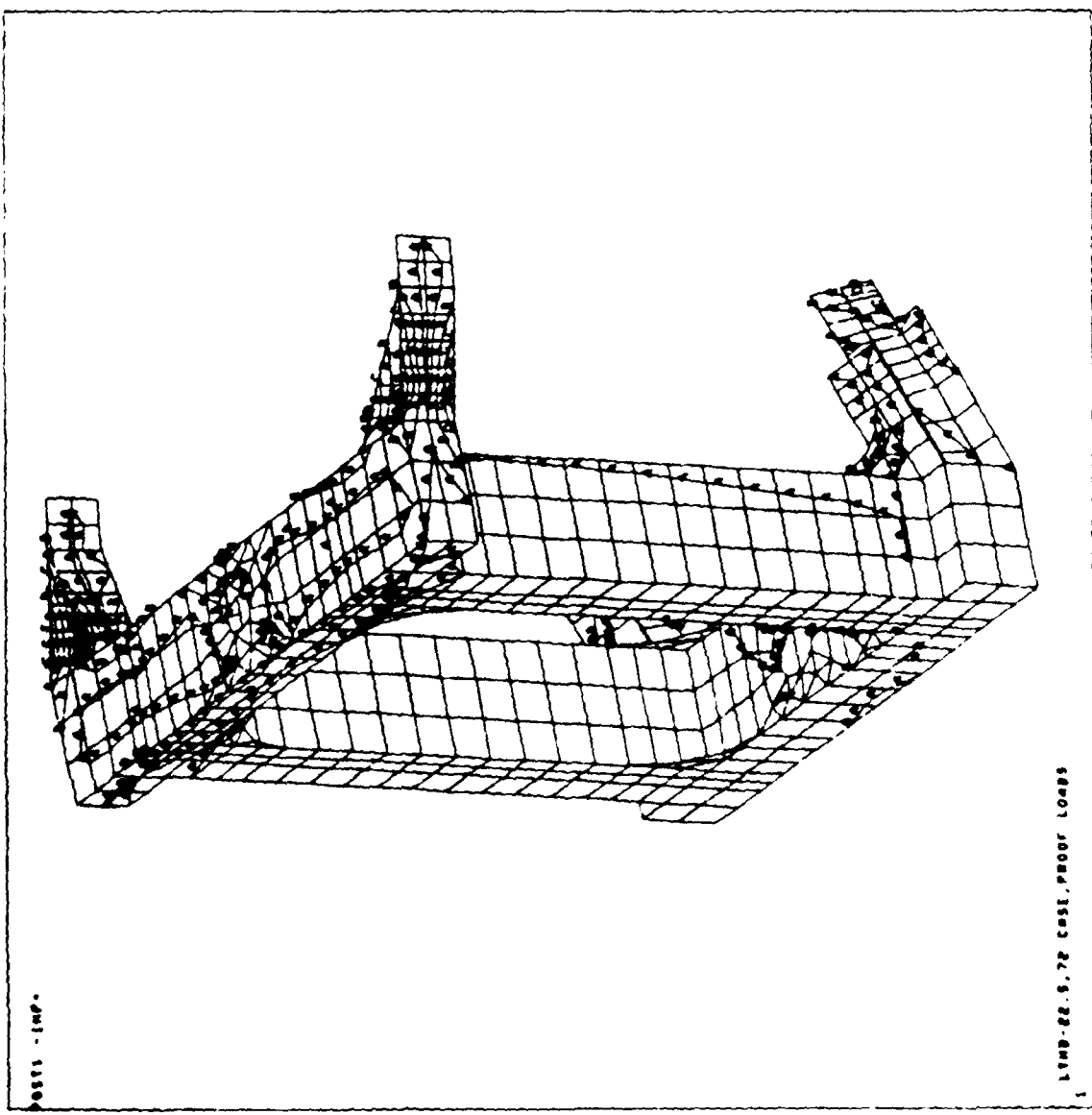
POST1 -IMP.

LTMB-22.5.72 CASE, PROOF LOADS

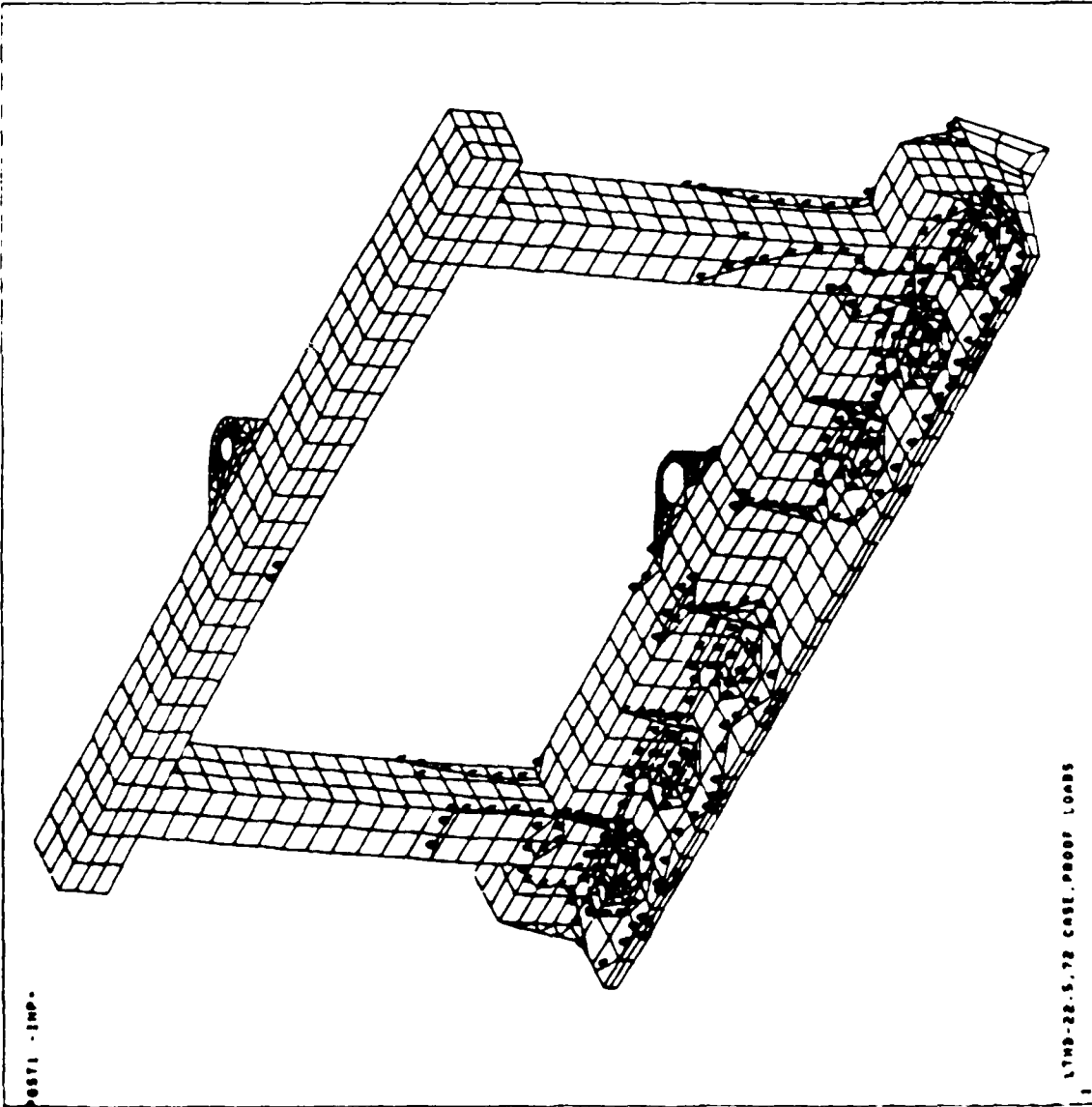
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ANALYSIS 4.20
JAN 21 1987
8160101
PROBLEM 1
STEP=1
ITER=1
TIME=.237
SLOC
TOP
AV=1
VV=1
ZU=1
SLOC=24.0
AV=57.3
VV=32.0
ZU=1.62
W180EN
GA=08217
AN=040
N=15550
B=22000
C=40000
D=63750
E=79400

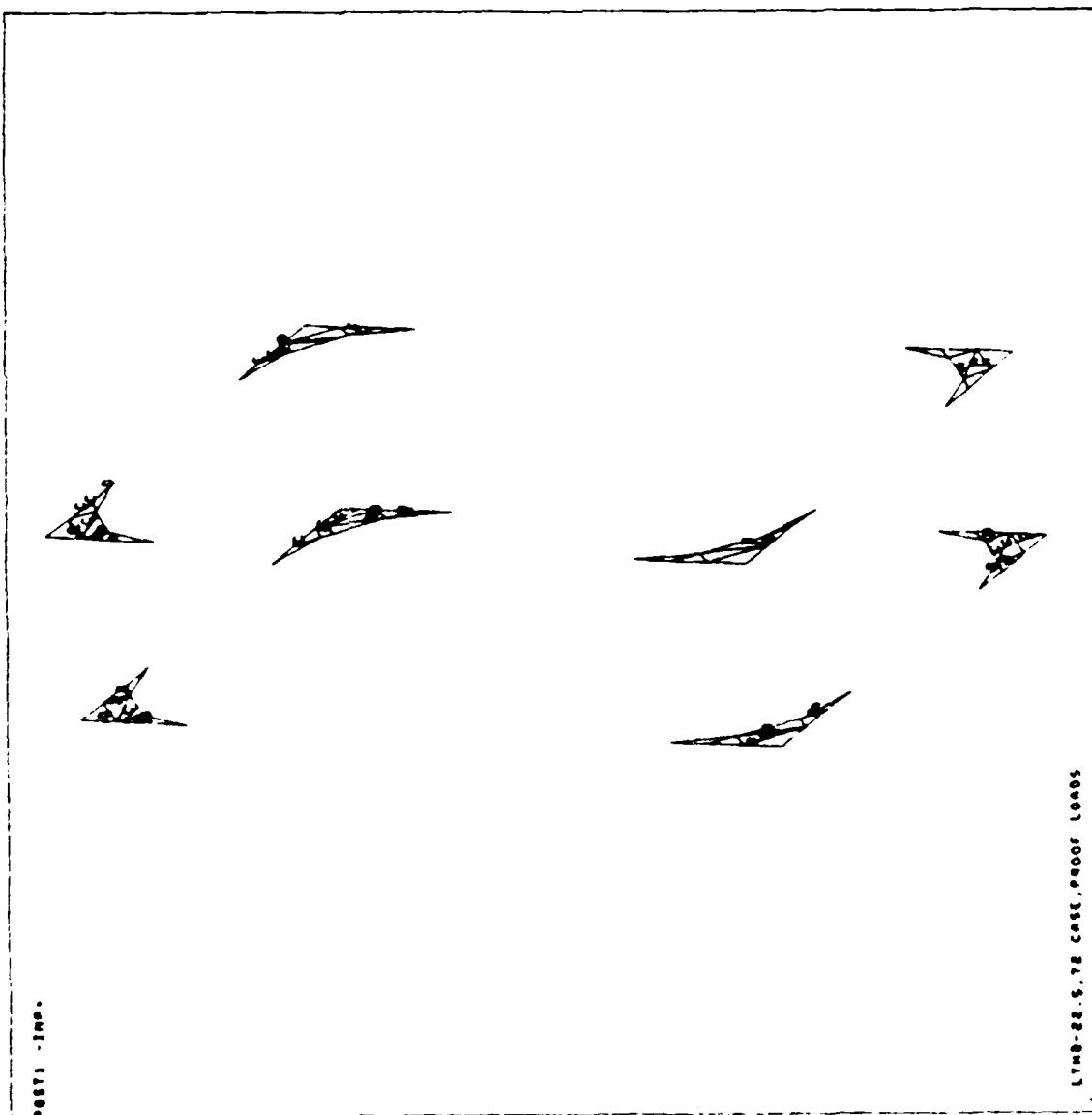
```



ANSYS 4.80
 JAN 21 1987
 10:10:02
 POST1 STRESS
 STEP=1
 TIME=1
 TIME=.237
 SLOC
 TOP
 XU=1
 YU=1
 ZU=1
 DIS=56.6
 RF=51.2
 VF=27.7
 ZF=4.17
 MIDEL
 ME=140063
 MM=227
 A=85531
 B=40030
 C=73145
 D=87452
 E=121759

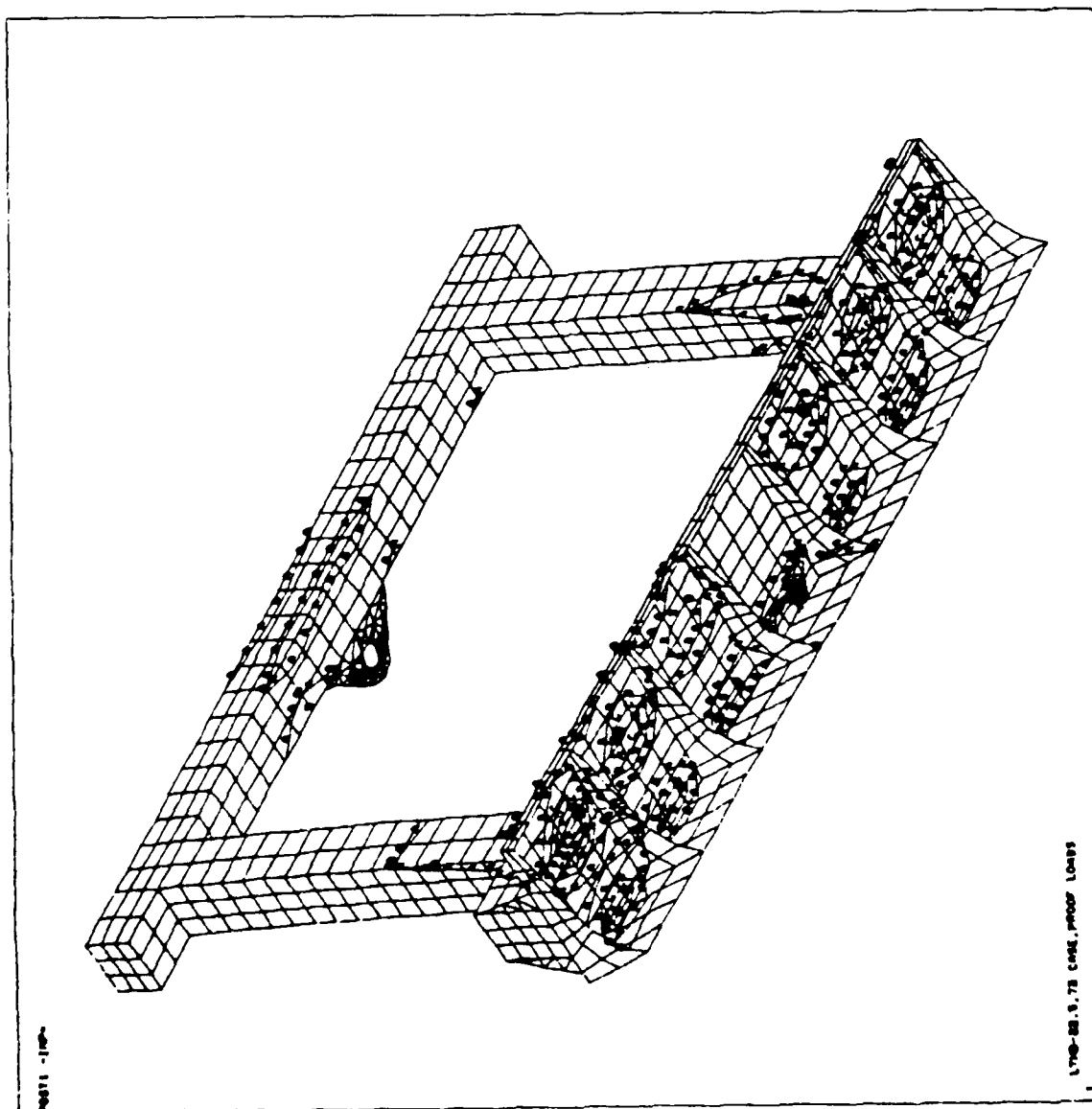


ANSYS 4.20
 JAN 21 1987
 11:04:11
 POST1, STRESS
 STEP=1
 ITER=1
 TIME=.837
 SICE
 TOP
 ZOOM
 X0=1
 Y0=1
 Z0=1
 D187=20
 XT=53
 YF=33.6
 ZF=3.5
 ROT0=1.3
 VRT0=1.2
 M18DEN
 M1=24078
 M1=2435
 A=6041
 B=9649
 C=13257
 D=16066
 E=20473



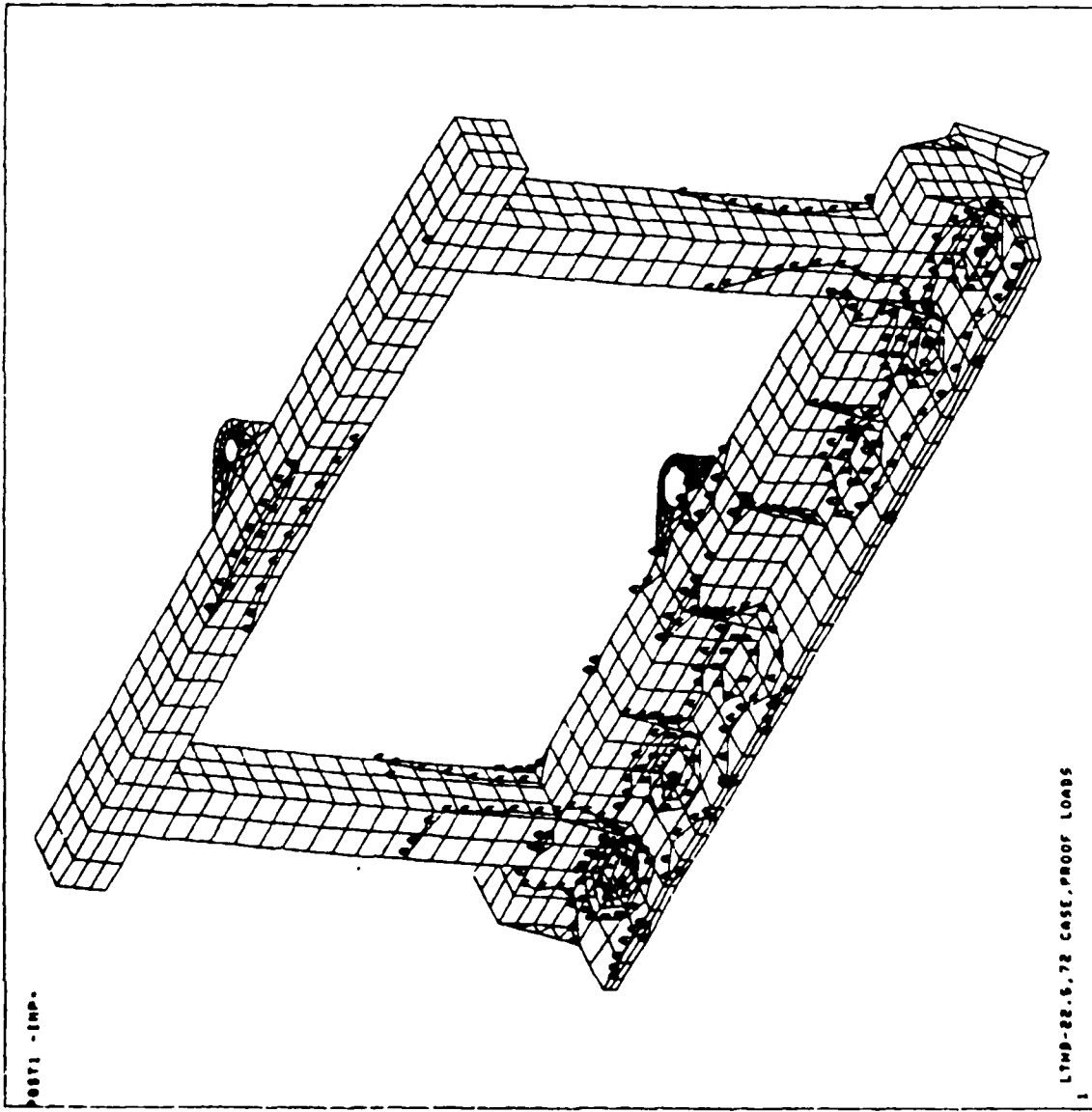
$t = 0.129 \text{ sec}$

44079 4.20
 JAN 81 1307
 13-07-100
 POST1, STRESS
 STEP=1
 ITER=1
 TIME=.129
 S16E
 TOP
 NU=1
 VU=1
 ZU=1
 BIST=50.0
 W=51.8
 W=56.4
 ZF=4.55
 H100EN
 R8=101607
 R8=200
 A=17154
 B=34904
 C=54004
 D=57504
 E=84704

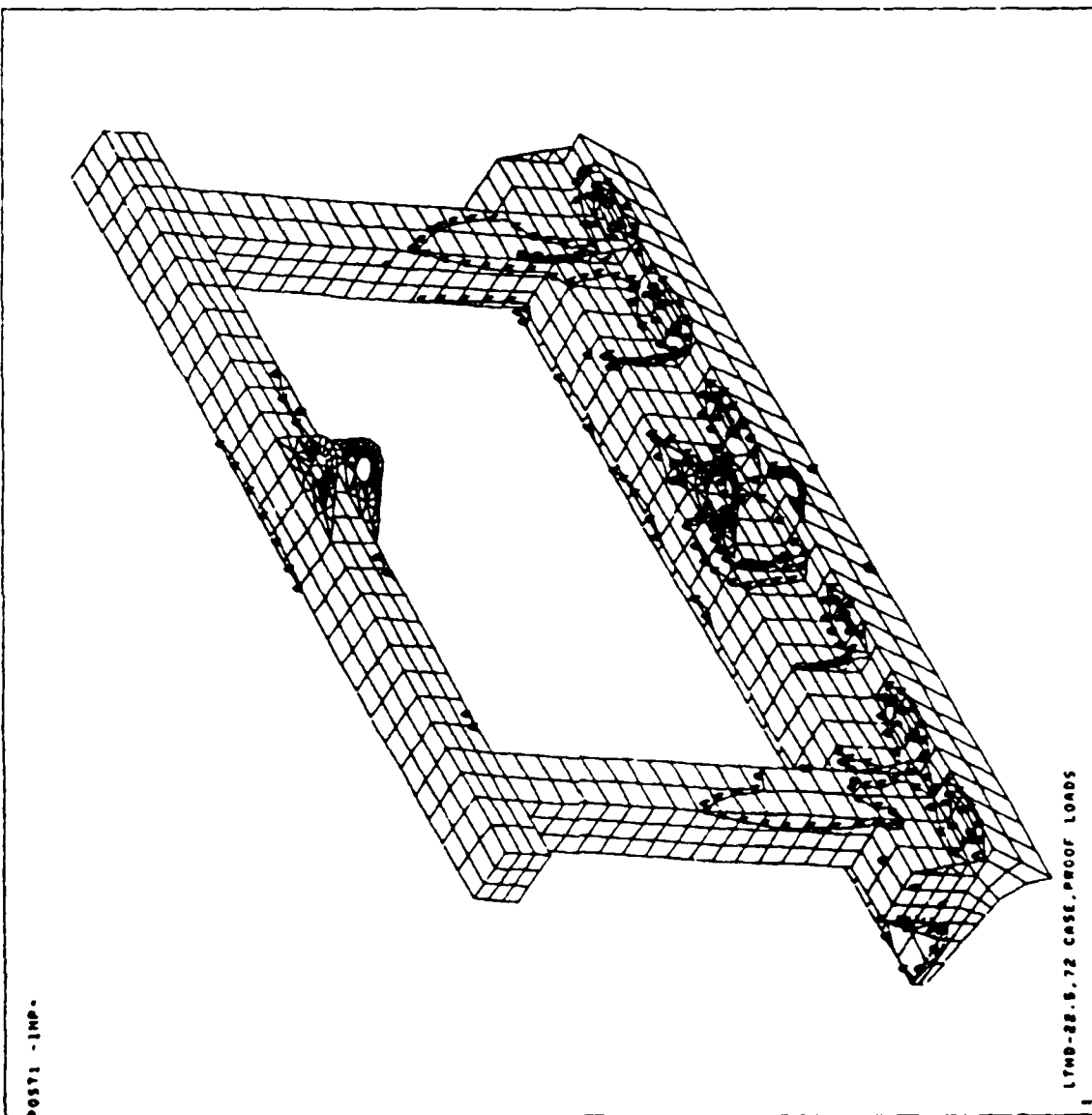


1700-00.5.75 CASE, PROOF LOADS

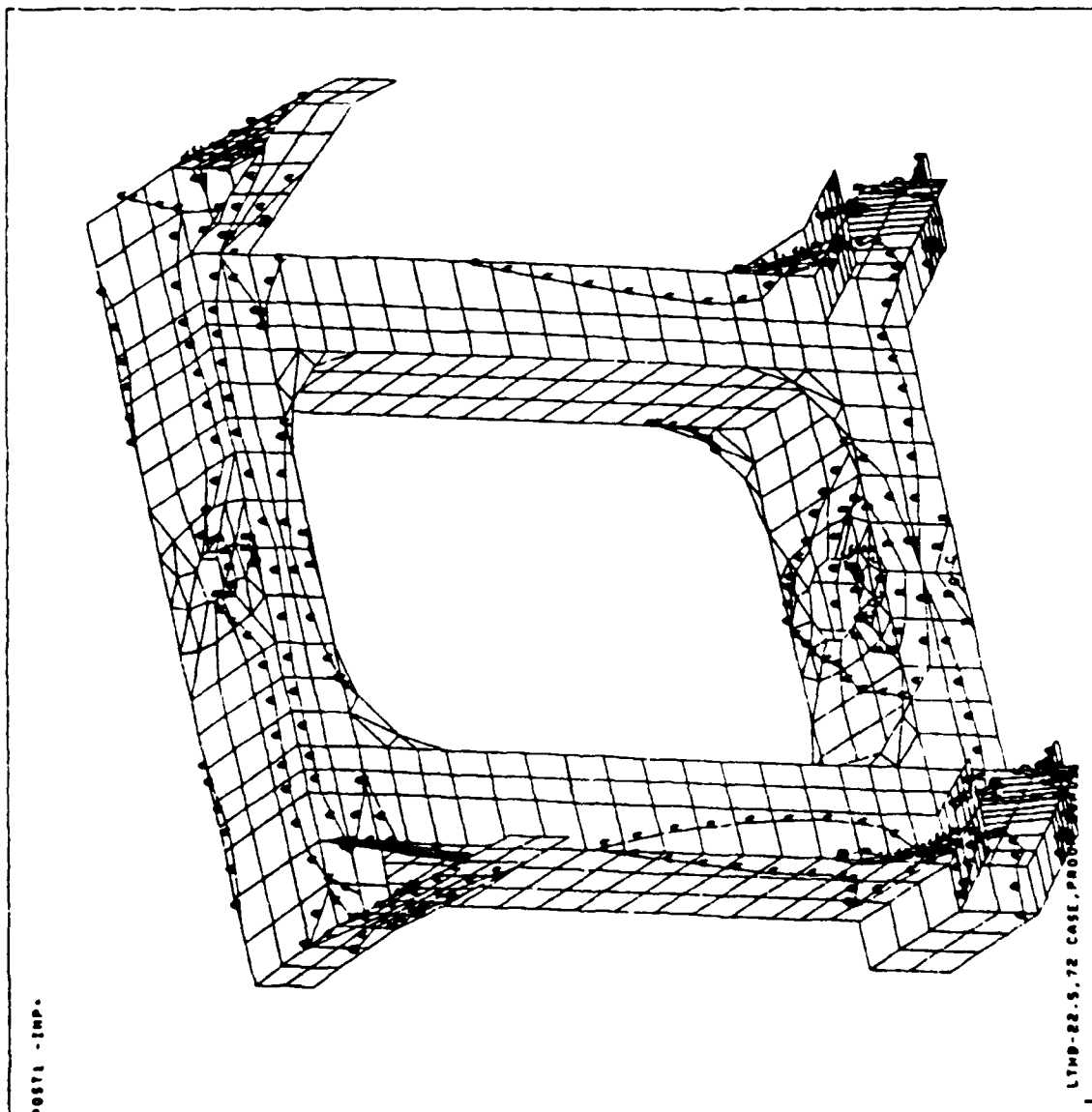
ANSYS 4.2D
 JAN 21 1987
 14-11-34
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.100
 SIZE
 TOP
 XU=1
 VU=1
 ZU=1
 BLST=66.6
 XF=51.2
 YF=27.7
 ZF=4.17
 HIDDEN
 MX=101803
 MY=206
 A=17184
 B=34084
 C=50884
 D=67884
 E=84784



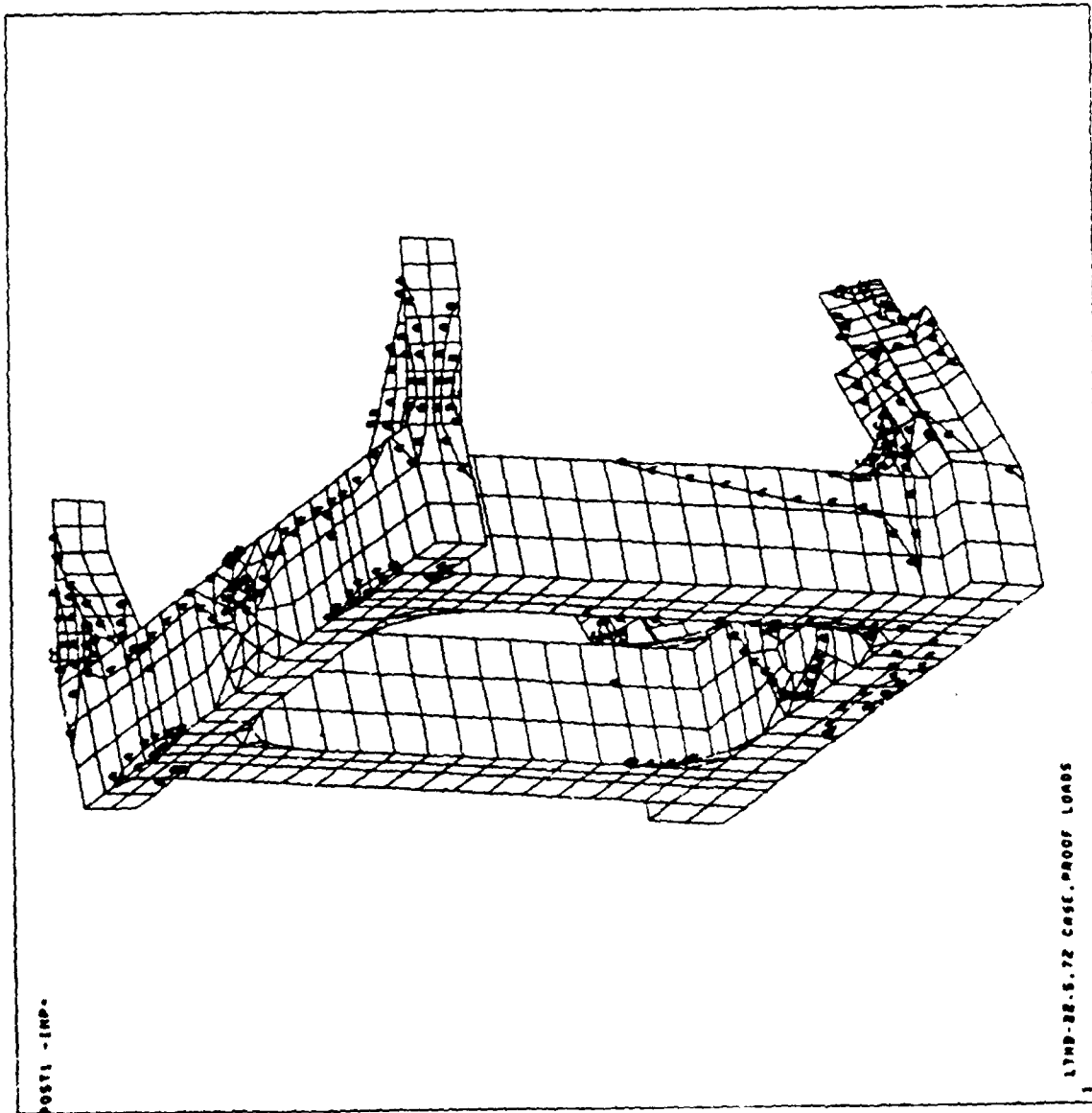
ANSYS 4.20
 JAN 21 1987
 14123103
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.120
 SIZE
 TOP
 KU=1
 VU=1
 ZU=-1
 DIST=58.2
 NF=54.2
 VF=26.4
 ZF=4.85
 MIBDEM
 MX=101603
 MY=206
 A=17184
 B=34004
 C=50004
 D=67004
 E=84704



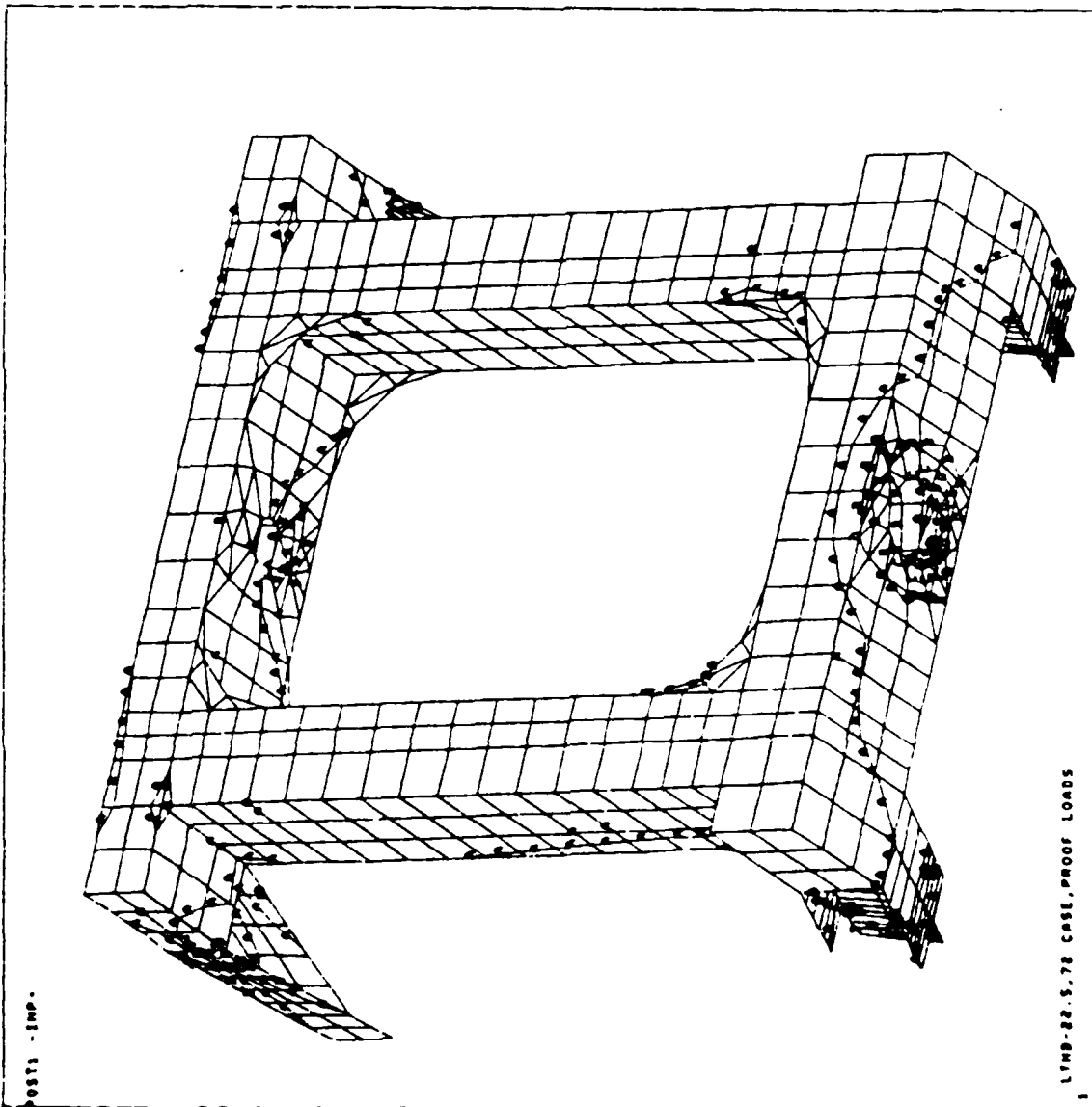
ANSYS 4.20
 JAN 21 1987
 14134107
 POST1, STRESS
 STEP=1
 ITER=1
 TIME=.129
 SIZE
 TOP
 KU=1
 VU=1
 20=1
 DIST=20.7
 XF=53.9
 VF=35
 ZF=-7.93
 MIDDLE
 MM=188221
 MM=1739
 A=82818
 B=43899
 C=64999
 D=86061
 E=107142

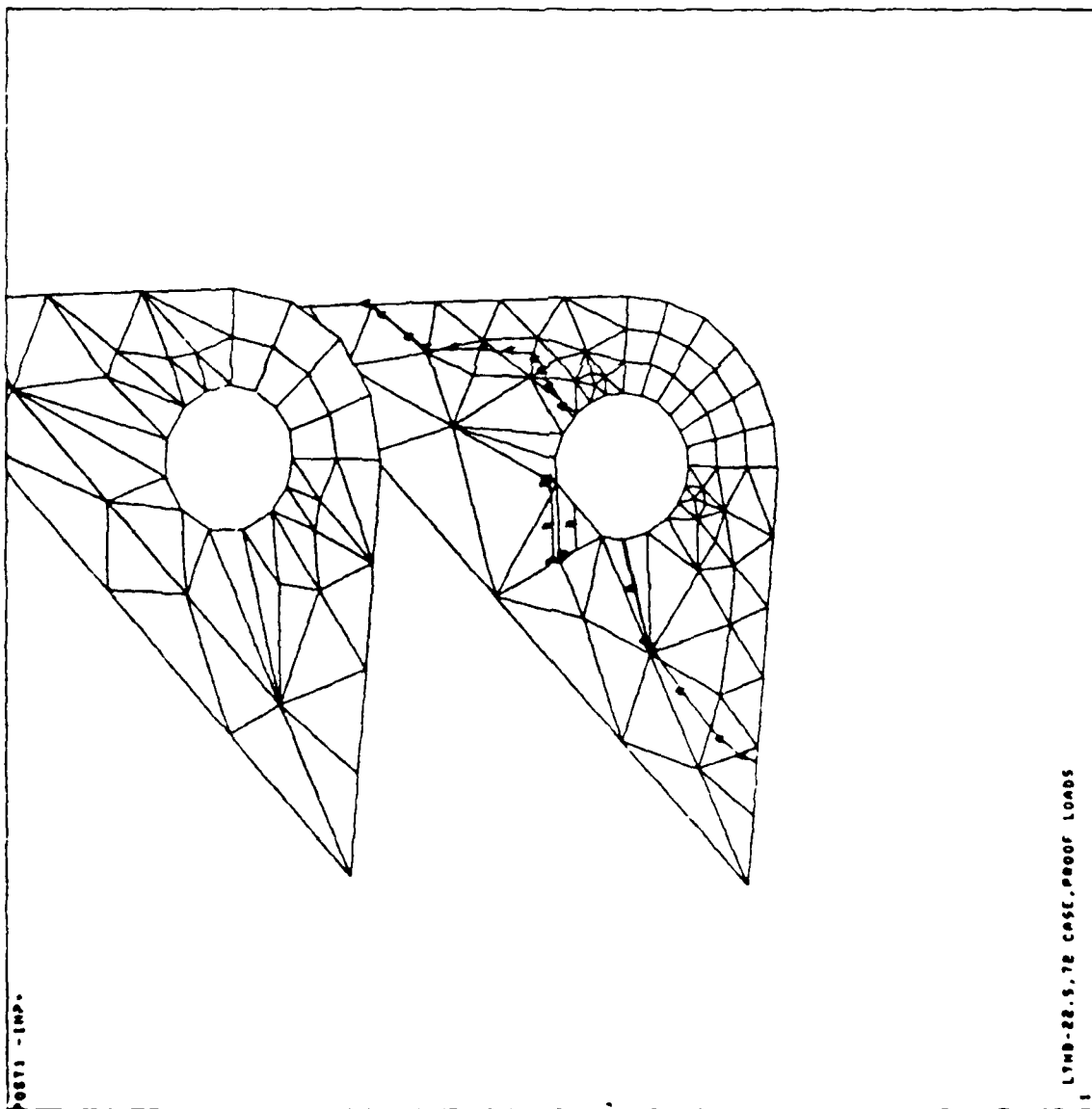


ABOVE 4.22
 JAN 21 1987
 14102133
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.100
 SICE
 TOP
 ZU=1
 YU=1
 ZU=1
 DIST=34.8
 XF=37.3
 YF=33.8
 ZF=-7.68
 MISOEM
 RM=128221
 CM=1136
 A=22818
 B=43880
 C=64880
 D=86861
 E=107142



ANSYS 4.20
 JAN 21 1987
 13158+27
 POST1, STRESS
 STEP=1
 TIME=1
 TIME=1.20
 SICE
 TOP
 RU=1
 VU=1
 ZU=1
 DIST=29.7
 XF=53.9
 YF=25
 ZF=7.93
 MIDDLE
 WK=128221
 WM=1730
 A=22810
 B=43800
 C=84980
 D=86061
 E=107142

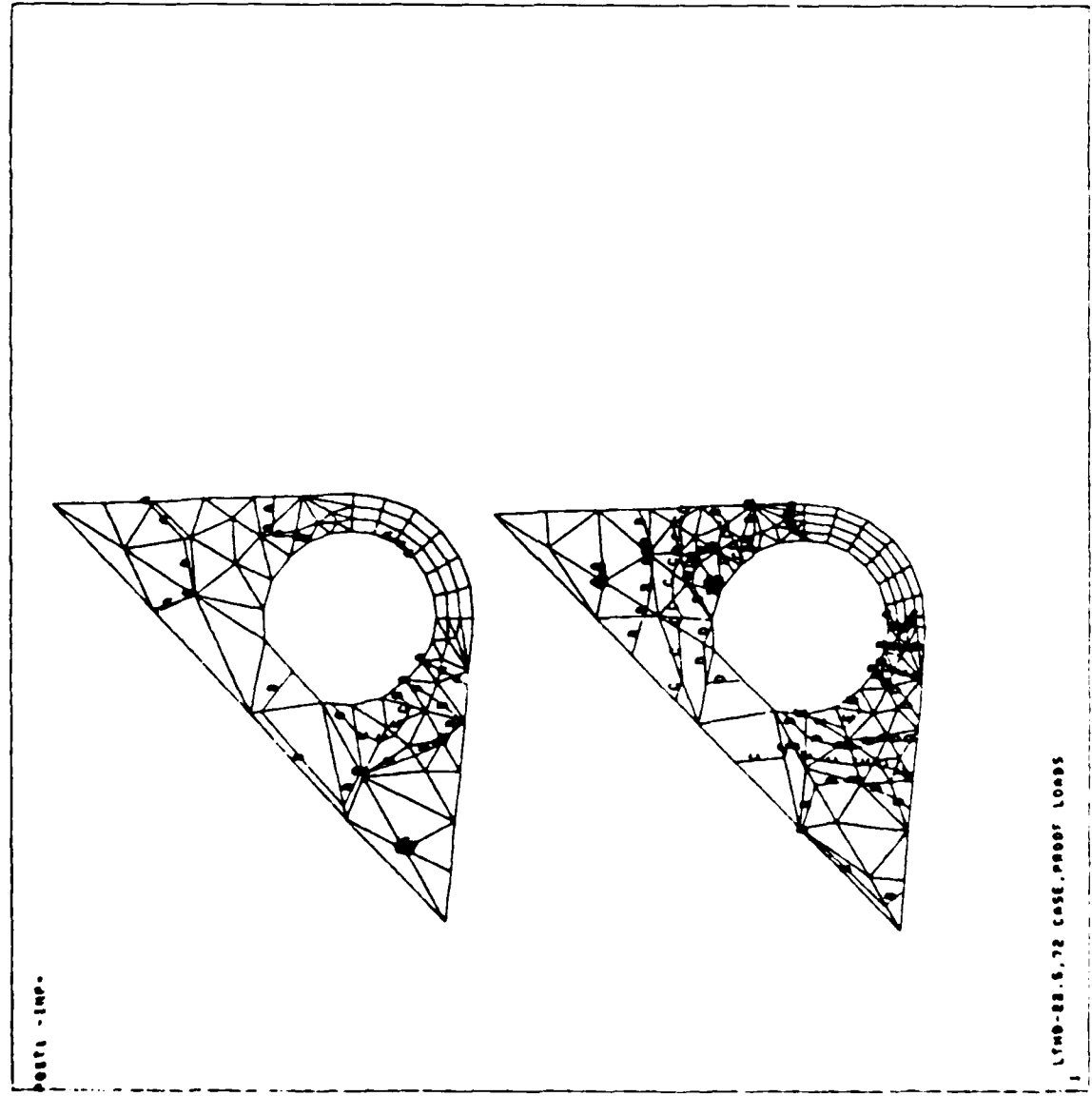




ANSYS 4.20
 JAN 21 1987
 14137146
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.180
 SLOC
 TOP
 ZOOM
 XU=1
 YU=1
 ZU=-1
 DIST=10.4
 XF=43.6
 YF=46.8
 ZF=5.75
 VRTO=1.54
 MIDDLE
 RM=14843
 MM=430
 Q=7380
 D=14181

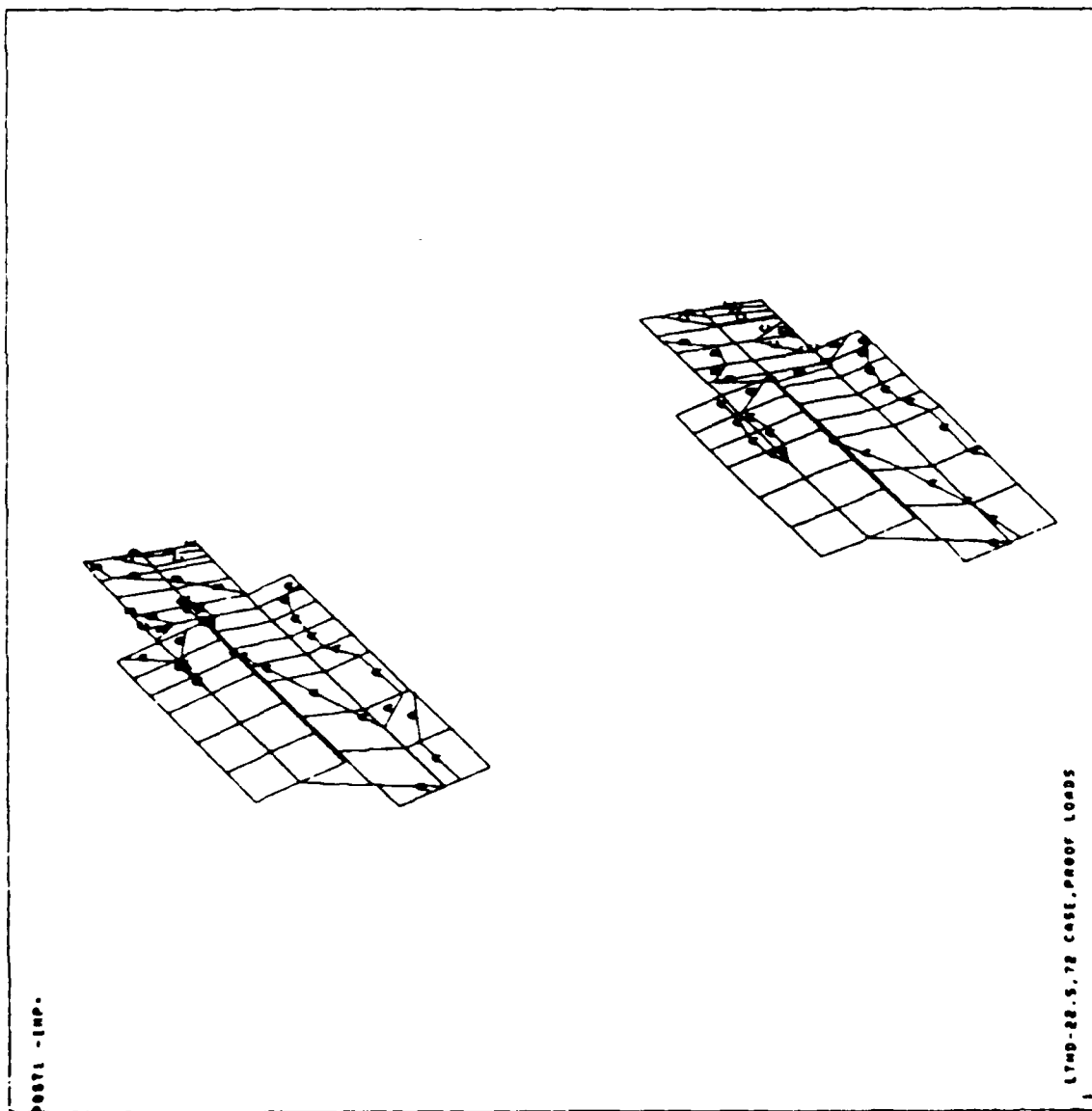
TOP TABS

ANSYS 4.02
 JAN 21 1987
 14146126
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.129
 SEQ
 TOP
 ZOOM
 RV=1
 VU=1
 ZU=-1
 DIST=14.1
 KF=89.4
 VF=12.4
 ZF=-12.9
 VRT0=1.67
 MIDDLE
 MX=41667
 MY=766
 A=7209
 B=14181
 C=21053
 D=27026
 E=34797



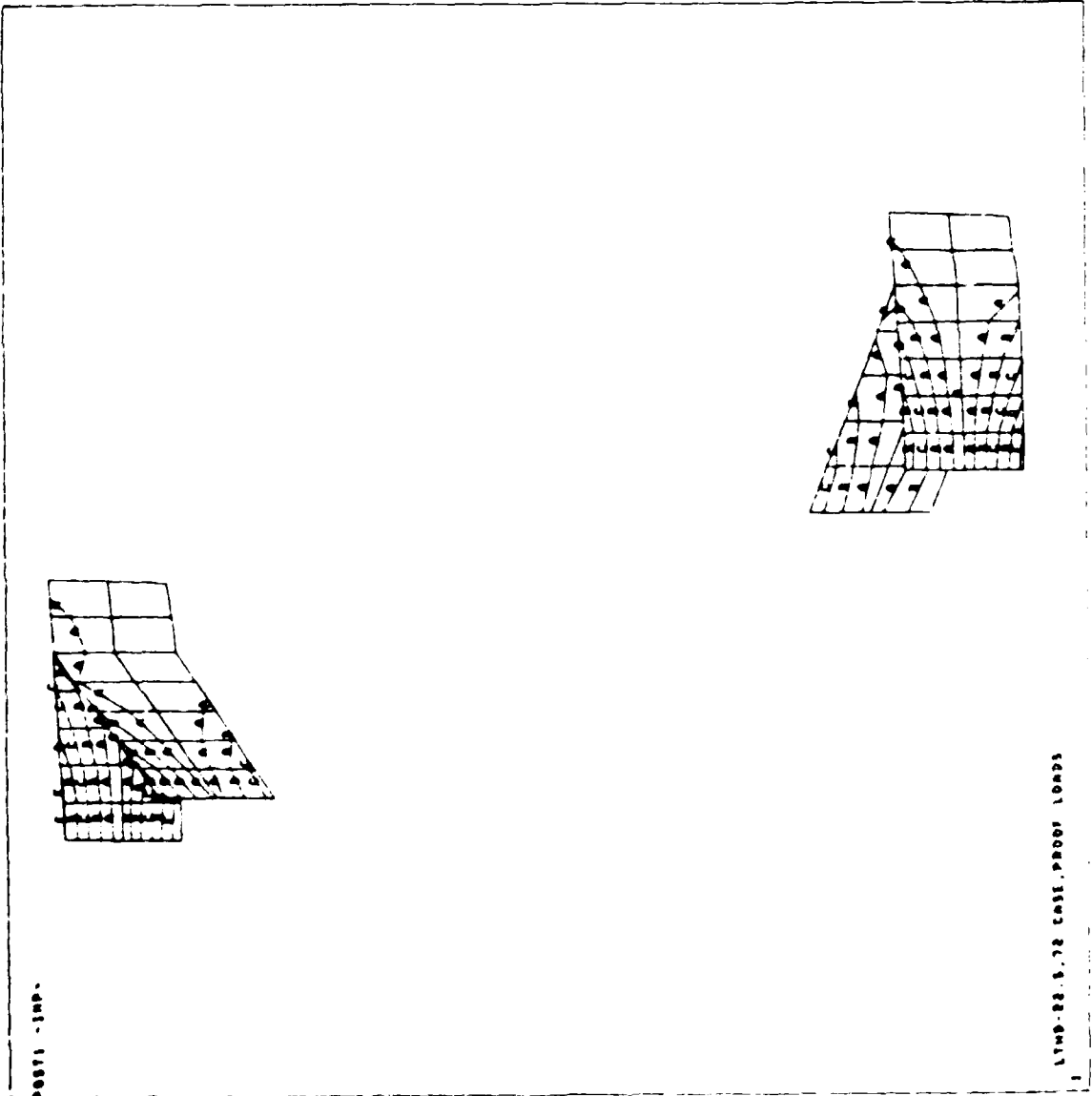
BOTTOM
 TABS

ANSYS 4.20
 JAN 21 1987
 14143142
 POST1 STRESS
 STEP=1
 ITER=1
 TIME=.120
 SIZE
 TOP
 ZOOM
 KU=1
 VU=1
 ZU=1
 0187.88.2
 KP=57.3
 VP=14.2
 ZP=-12.7
 V870=1.57
 MIDDLEM
 RA=124714
 RH=7700
 A=27205
 B=46771
 C=66257
 D=85743
 E=105229



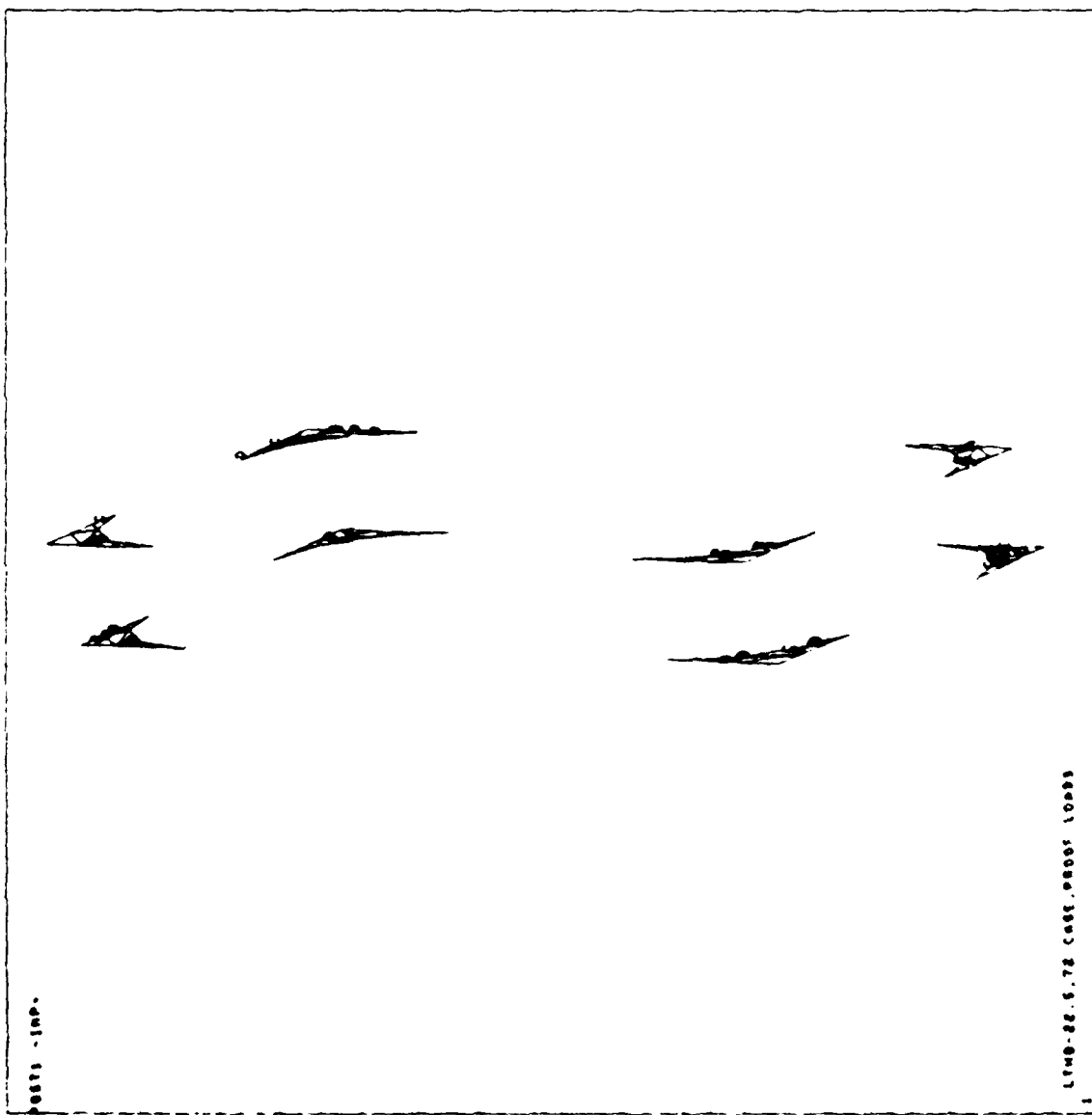
GIMBAL
 LOWER
 ARMS

ANSYS 4.20
 JAN 21 1987
 10145122
 POST1, STRESS
 STEP=1
 ITER=1
 TIME=1.00
 SICE
 TOP
 2004
 10-1
 20-1
 20-1
 0157.85
 27-57.1
 27-53.3
 27-12.2
 200-1.67
 11954
 22-77389
 22-8868
 2-18621
 2-30376
 2-42128
 2-52883
 2-65637



GIMBAL
 UPPER
 ARMS

ARSVE 1.20
 JAN 21 1987
 14145110
 POST1 STRESS
 STEP-1
 ITER-1
 TIME-1.00
 SICE
 TOP
 ZOOM
 VU-1
 VU-1
 VU-1
 DIST-41.7
 VF-53
 VF-23.6
 ZF-3.5
 V870-1.67
 M108CH
 AR-34610
 AR-3765
 A-8028
 D-14113
 C-19288
 D-24463
 E-28638



0.129 sec

PRINT ELEMENT STRESS ITEMS PER ELEMENT

***** POST1 ELEMENT STRESS LISTING *****

LOAD STEP	1	ITERATION	1	SECTION	1
TIME	0.12900		LOAD CASE	1	
ELEM	5010	502	50V1	5022	50V2
3069	42.720	-80.486	86.424	106.63	-109.26
3070	-18.506	-71.002	-416.54	-343.80	608.00
3071	46.720	-71.011	220.72	74.807	-330.34
3072	18302	-1671.31	-1671.31	1715.06	2573.0
3073	-8689.5	1087.8	3684.0	-1163.6	-4453.4
3074	1.4806	-306.62	-548.57	171.91	383.08
3075	-180.86	100.73	1927.0	186.88	-1121.0
3076	-16506	-821.72	-1624.0	1874.1	8816.4
3077	150.06	-31.197	-943.46	-81.733	809.00
3078	41.563	2544.0	1400.1	-8200.6	-1187.0
3079	117.72	-611.07	-686.46	401.02	606.62
3080	10.404	62.038	1436.7	-286.06	-1311.8
3081	5132.0	426.13	-506.73	537.30	713.87
3087	-228.55	9764.3	1866.8	-4686.5	-643.06

***** MORE IYES, NO OR CONTINUOUS *****

***** POST1 ELEMENT STRESS LISTING *****

LOAD STEP	1	ITERATION	1	SECTION	1
TIME	0.12900		LOAD CASE	1	
ELEM	5010	502	50V1	5022	50V2
3088	-9040.0	841.95	-326.39	-46.078	1070.8
3089	-1401.5	-302.03	383.21	433.96	-834.24
3090	401.31	676.89	-291.28	-657.82	800.26
3091	-71.536	-46.790	181.41	851.96	808.60
3092	-351.65	256.88	348.12	632.74	809.11
3093	3.5604	136.10	-176.00	-63.146	17.428
3094	212.13	-385.06	457.00	-377.00	647.58
3095	-39.378	-54.170	-84.127	187.07	204.28
3096	-204.05	496.17	689.55	510.28	580.16
3097	2.8844	48.166	-30.881	-33.789	18.765
3098	273.28	-282.56	312.25	-488.67	625.03
3099	-282.24	281.85	-264.01	386.99	-388.10
3100	168.05	-78.006	-144.10	-308.04	-644.10
3101	-380.53	360.20	-213.32	418.95	-288.88

***** MORE IYES, NO OR CONTINUOUS *****

ORDER POSTE ELEMENT STRESS LISTING 0000

LOAD STEP	1	ITERATION	1	SECTION	1
TIME	0.10000	LOAD CASE	1		
1100	1010	102	1011	1022	1012
1100	1010	104.03	1110.2	48.433	331.01
1100	1010	105.12	232.05	-172.76	130.05
1100	1010	106.05	256.02	-300.12	622.30
1100	1010	107.00	184.26	211.85	406.56
1100	1010	108.10	581.00	578.87	808.07
1100	1010	109.10	1110.24	-47.2274	28.828
1100	1010	110.00	585.77	-416.00	556.14
1100	1010	111.00	-48.000	-22.531	0.1051
1100	1010	112.00	741.21	800.21	300.23
1100	1010	113.00	231.13	-308.61	500.13
1100	1010	114.00	800.00	367.16	257.72
1100	1010	115.00	305.52	363.39	-200.00
1100	1010	116.00	-373.53	348.51	-374.72
1100	1010	117.00	-271.00	326.36	-300.05

... 11 00 00 CONTINUOUS ...

ORDER POSTE ELEMENT STRESS LISTING 0000

LOAD STEP	1	ITERATION	1	SECTION	1
TIME	0.10000	LOAD CASE	1		
1100	1010	102	1011	1022	1012
1100	1010	104.03	1110.2	48.433	331.01
1100	1010	105.12	232.05	-172.76	130.05
1100	1010	106.05	256.02	-300.12	622.30
1100	1010	107.00	184.26	211.85	406.56
1100	1010	108.10	581.00	578.87	808.07
1100	1010	109.10	1110.24	-47.2274	28.828
1100	1010	110.00	585.77	-416.00	556.14
1100	1010	111.00	-48.000	-22.531	0.1051
1100	1010	112.00	741.21	800.21	300.23
1100	1010	113.00	231.13	-308.61	500.13
1100	1010	114.00	800.00	367.16	257.72
1100	1010	115.00	305.52	363.39	-200.00
1100	1010	116.00	-373.53	348.51	-374.72
1100	1010	117.00	-271.00	326.36	-300.05

... 11 00 00 CONTINUOUS ...

AD-A183 994

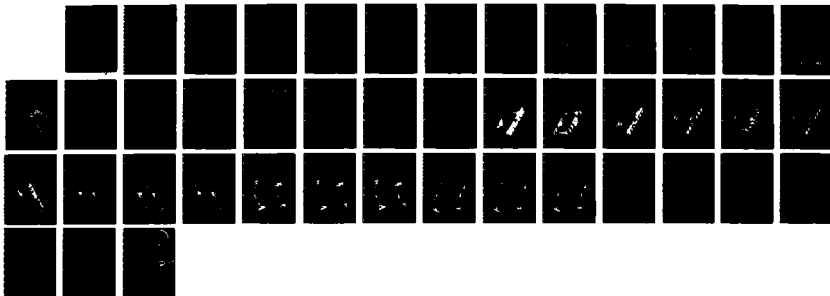
LIGHTWEIGHT TOWED HOWITZER DEMONSTRATOR PHASE 1 AND
PARTIAL PHASE 2 VOLUM (U) FMC CORP MINNEAPOLIS MINN
NORTHERN ORDNANCE DIV R RATHE ET AL APR 87
FMC-E-3041-VOL-D3-PT-2 DAAA21-86-C-0047

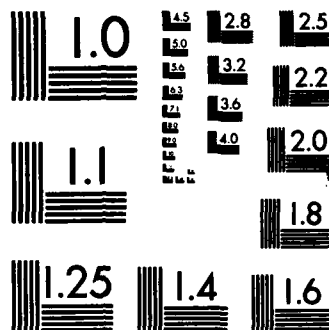
2/2

UNCLASSIFIED

F/G 19/6

NL





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

Y

***** POST1 ELEMENT STRESS LISTING *****

LOAD STEP	1	ITERATION	1	SECTION	1
TIME	0.18000		LOAD CASE	1	
ELEM	SDIR	S22	SDV1	S222	SDV2
3130	20.071	27.203	188.00	-42.532	-306.50
3131	109.92	47.308	428.20	-1.5135	48.400
3132	5.6342	-26.040	301.70	30.793	-103.20
3133	94.846	1.7101	81.621	-1.2023	-254.20
3134	109.40	57.367	-105.70	-28.157	408.00
3135	38.618	37.967	-843.10	-20.326	402.00
3136	-857.93	34.200	117.42	97.908	328.21
3137	7.7917	20.208	-150.42	-20.501	42.920
3140	305.00	-61.826	170.83	-26.144	364.66
3141	17.346	-25.385	-38.660	23.628	130.00
3142	-344.73	17.651	325.76	74.809	383.00
3143	8.8606	31.273	34.223	-24.065	-5.1631
3144	352.43	-69.642	300.48	-24.281	321.14
3145	-30.000	-37.275	85.710	32.240	-268.21

NOTE (YES, NO OR CONTINUOUS).

***** POST1 ELEMENT STRESS LISTING *****

LOAD STEP	1	ITERATION	1	SECTION	1
TIME	0.18000		LOAD CASE	1	
ELEM	SDIR	S22	SDV1	S222	SDV2
3146	-321.00	34.234	425.00	41.026	100.72
3147	5.4700	24.171	258.31	-20.318	-107.33
3148	200.45	-45.247	237.42	14.224	-152.42
3149	-57.060	-20.720	226.50	-1.5803	-400.00
3150	-98.910	22.615	171.32	-68.923	-327.06
3158	-322.37	974.46	-201.56	10145.	4360.2
3159	-849.00	25594.	9427.4	23688.	8900.7
3164	1182.7	35767.	10080.	-7552.6	-3510.1
3166	70.236	148.71	-269.87	0.79314E-03	-38.320
3167	-6.1000	625.19	347.65	-0.33644E-02	251.47
3168	15.017	48.204	319.36	0.17600E-02	90.516
3169	-340.10	-61.267	247.82	0.5510E-02	118.47
3170	-64.630	37.244	-64.762	0.65742E-02	-120.70
3180	-84.630	98.000	76.422	0.46407E-02	-48.956

NOTE (YES, NO OR CONTINUOUS).

0000 POST1 ELEMENT STRESS LISTING 0000

LOAD STEP 1 ITERATION 1 SECTION 1
TIME 0.10000 LOAD CASE 1

ELEM	SD18	SD2	SDV1	SD22	SDV2
3101	30.000	10.000	-376.10	0.87001E-03	-185.52
3102	-7.1100	41.017	-80.376	-0.30308E-04	-43.804
3103	-31.670	3600.9	64.90	0.89047E-01	835.66
3104	161.06	2313.1	-77.436	0.38101E-01	-107.70
3105	307.58	1699.5	-956.41	0.48000E-01	-482.73
3106	814.00	903.11	-3700.0	0.35303E-01	-1680.0
3107	500.00	1174.0	8000.1	0.42744E-01	1800.7
3108	346.03	1902.3	685.17	0.37800E-01	302.43
3109	60.000	2295.5	-152.03	0.30000E-01	-10000
3200	-01.600	2224.4	-944.02	0.10000E-01	-381.66
3201	-141.13	811.84	-893.08	0.10007E-01	-485.10
3202	-71.300	2040.0	-3466.8	0.12204E-01	-1827.2
3203	-200.14	-1.7349	8377.5	-0.10110E-01	905.00
3204	-011.00	-715.34	509.44	-0.46305E-02	317.75

NOTE (YES,NO OR CONTINUOUS).

0000 POST1 ELEMENT STRESS LISTING 0000

LOAD STEP 1 ITERATION 1 SECTION 1
TIME 0.10000 LOAD CASE 1

ELEM	SD18	SD2	SDV1	SD22	SDV2
3205	-100.00	-1153.0	-92.014	-0.05478E-03	26.051
3206	-12.107	-2326.7	-556.61	-0.20308E-02	-814.45
3207	-101.01	148.00	-85.671	0.80300E-01	-144.77
3208	408.01	-2466.7	6331.0	1000.6	-2023.6
3209	1694.6	4042.1	-7092.0	-0.42414E-01	-4908.1
3210	975.56	-1093.4	-1302.0	-0.50001E-03	-876.41
3211	813.20	-1236.2	1093.5	-0.59100E-04	357.97
3212	-100.52	-493.64	1002.0	0.20000E-04	308.24
3213	-226.05	-1070.4	-364.16	-0.00172E-03	-101.13
3214	378.01	-2207.5	-548.14	-0.03003E-03	-243.10
3215	400.74	6495.2	-4341.3	-0.001.0	2309.9
3216	1758.6	215.14	-3271.9	-0.72005E-02	-1522.5
3217	649.09	-857.70	7020.0	-0.20701E-01	3046.6
3218	702.70	-3166.4	-50.336	-0.55110E-02	116.56

NOTE (YES,NO OR CONTINUOUS).

2000 POST1 ELEMENT STRESS LISTING 20000									
LOAD STEP		1 ITERATION.		1 SECTION.		1			
TIME.		0.10000		LOAD CASE.		1			
ELEM	1000	102	1001	1001	1001	1001	1001	1001	1001
2000	-7.1000	-1000.02	-707.47	-707.47	-707.47	-707.47	-707.47	-707.47	-707.47
2000	-2000.0	3212.3	1202.7	1202.7	1202.7	1202.7	1202.7	1202.7	1202.7
2001	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2001	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2002	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2002	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2003	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2004	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2005	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2006	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2007	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2008	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2009	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2010	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2011	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2012	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2013	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2014	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2015	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2016	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2017	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2018	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2019	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2020	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2021	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2022	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2023	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2024	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2025	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2026	-2000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
2027	-2000.0	1000.0	1000.0	1000.0	1000.0				

2020 POST1 ELEMENT STRESS LISTING 2020									
LOAD STEP 1		ITERATION-		SECTION- 1					
TIME. 0.12000		LOAD CASE. 1							
ELEM	SDIR	SDZ	SDY1	SDZ2	SDY2	SDY3	SDY4	SDY5	SDY6
2023	1011.5	00.006	22.951	0.23041E-03	19.207				
2024	1207.0	-6.7007	55.924	0.16065E-04	17.004				
2025	1200.0	101.34	-10.731	0.28724E-03	-8.0024				
2026	1200.0	00.450	37.776	0.20613E-03	-7.0327				
2028	4705.7	37.776	31.494	0.20937E-03	14.155				
2029	4841.3	-30.510	-101.09	0.15828E-03	34.948				
2040	2142.6	-0.47581	-178.62	-0.41067E-04	68.360				
2041	2020.6	96.640	11.450	0.24277E-03	7.0406				
2042	362.60	25.017	-12.43	0.15545E-05	-0.7375				
2043	3715.1	00.020	18.126	0.24920E-03	10.705				
2044	4005.7	22.097	59.322	0.24130E-03	28.352				
2045	1400.0	-100.00	23.332	-0.30251E-03	8.7001				
2046	-2312.3	-140.83	0.3158	-0.04037E-03	6.7545				
2047	-4009.6	-127.36	19.029	-0.00916E-03	11.158				

8888 POST1 ELEMENT STRESS LISTING 8888

LOAD STEP 1 ITERATION-1 SECTION-1
TIME-0.10000 LOAD CASE-1

ELEM	8010	802	80V1	80V2	80V3
3000	-1037.0	-88.070	-80.03	-0.30330E-03	-0.4.500
3001	-4000.0	-104.00	-110.41	-0.14130E-03	-30.100
3002	-3314.0	-101.13	-10.911	-0.03300E-03	-0.0000
3003	-8046.0	104.00	36.300	-0.23010E-03	11.007
3004	10.100	-0.0001	-86.104	-0.00300E-04	3.0760
3005	-11000.0	10.000	-5.0750	-0.70047E-05	-0.00370
3006	3040.0	-97.201	-0.01700	-0.00304E-04	0.15100
3007	-907.43	-0.4004	130.50	-0.00140E-04	43.100
3008	-1010.0	-10.720	143.00	-0.10410E-02	50.010
3009	712.61	-84.014	-26.040	-0.00000E-03	-15.010
3010	-0114.1	-70.305	64.500	-0.77000E-03	25.000
3011	-3314.0	-101.13	-10.911	-0.03300E-03	-0.0000
3012	-3046.0	-97.201	-0.01700	-0.00304E-04	0.15100
3013	-1079.1	-70.305	1.1007	-0.00000E-03	0.0000

NOTE (VES.NO ON CONTINUOUS).

8888 POST1 ELEMENT STRESS LISTING 8888

LOAD STEP 1 ITERATION-1 SECTION-1
TIME-0.10000 LOAD CASE-1

ELEM	8010	802	80V1	80V2	80V3
3000	0014.1	-11.100	10.007	1.0001	-0.1300
3001	-1.400	977.00	0.00000E+00	1340.0	0.00000E+00
3002	-1.000	-030.00	0.00000E+00	-3147.0	0.00000E+00
3003	0.07000E-01	700.03	0.00000E+00	-070.30	0.00000E+00
3004	-0.00037E-01	004.00	0.00000E+00	-700.03	0.00000E+00
3005	-0.00070	-120.10	0.00000E+00	100.00	0.00000E+00
3006	-1.0730	003.33	0.00000E+00	-1200.0	0.00000E+00
3007	0.10003	-130.00	0.00000E+00	120.70	0.00000E+00
3008	0.00035	1000.0	0.00000E+00	-0100.7	0.00000E+00
3009	0.00037	-0217.1	0.00000E+00	1000.3	0.00000E+00
3010	0.00003	-1027.1	0.00000E+00	1046.2	0.00000E+00
3011	-0.23030E-01	-113.57	0.00000E+00	100.00	0.00000E+00
3012	0.02000	1010.0	0.00000E+00	-774.00	0.00000E+00
3013	-0.02000	-774.00	0.00000E+00	-12700.	0.00000E+00

NOTE (VES.NO ON CONTINUOUS).

0000 POST1 ELEMENT STRESS LISTING 0000

LOAD STEP 1 ITERATION 1 SECTION 1
TIME 0.10000 LOAD CASE 1

ELEM	SE18	SE2	SEV1	SE28	SEV2
3300	0.16307E-10	-0.53500E-06	0.00000E+00	0.18000E-00	0.00000E+00
3301	-1.00000	-77.000	0.00000E+00	00.200	0.00000E+00
3302	0.24123	-140.14	0.00000E+00	130.93	0.00000E+00
3303	-0.00001	907.84	0.00000E+00	-8030.0	0.00000E+00
3304	-0.57620	-1375.8	0.00000E+00	800.18	0.00000E+00
3305	75.330	-108.23	104.91	400.42	-833.14
3306	68.014	507.36	-850.10	-152.30	40.630
3307	90.000	709.10	-300.13	-80.303	10.401
3308	-65.426	500.74	-187.46	-500.66	820.66
3309	-80.043	87.000	-4.0306	-300.46	181.00
3310	-201.73	147.00	101.40	-300.93	-800.07
3311	-200.04	-800.36	-606.04	800.06	-1703.0
3312	0.3007	103.04	80.430	-80.718	-80.374

NOTE 1YES,NO OR CONTINUOUS).

0000 POST1 ELEMENT STRESS LISTING 0000

LOAD STEP 1 ITERATION 1 SECTION 1
TIME 0.10000 LOAD CASE 1

ELEM	SE18	SE2	SEV1	SE28	SEV2
3313	170.16	-95.774	-104.60	200.86	207.64
3314	200.87	-205.47	-823.17	186.56	100.00
3315	400.36	-100.10	-804.78	340.86	420.70
3316	011.06	-407.30	-374.30	132.00	102.30
3317	705.10	-104.64	-104.10	100.30	355.46
3318	000.04	-349.07	-317.23	60.000	800.16
3319	000.73	-51.816	-72.704	95.057	10.100
3320	-214.10	821.12	1024.0	-1377.0	-800.03
3321	-802.44	47.050	546.00	6.3012	255.00
3322	-224.44	109.13	-56.710	-257.00	304.00
3323	-800.33	320.24	-316.01	-172.77	170.00
3324	-100.64	807.41	-100.27	-400.00	370.00
3325	-87.041	407.40	-300.00	-800.67	800.30
3326	-110.76	62.502	-32.023	-90.070	70.007

NOTE 1YES,NO OR CONTINUOUS).

0000 POST1 ELEMENT STRESS LISTING 0000

LOAD STEP 1 ITERATION 1 SECTION 1
TIME 0.10000 LOAD CASE 1

ELEM	S01R	S02	S0V1	S02R	S0V2
3267	8.0043	111.86	-40.824	-07.300	-26.160
3268	83.084	-800.00	86.301	170.03	-82.70
3269	-046.37	534.10	-284.08	146.38	-11.307
3270	-050.31	401.24	-186.00	-260.80	150.50
3271	-1350.0	607.46	-375.05	-160.00	170.11
3272	-1350.0	607.46	-375.05	-160.00	170.11
3273	-0100.0	407.37	-800.00	-21.00	201.00
3274	-2111.1	77.000	13.010	-300.00	813.30
3275	-0003.7	510.01	-80.306	-400.30	1740.0
3276	-050.06	1000.0	61.008	-800.00	7.0438
3277	-077.10	914.04	86.549	-801.00	-14.807
3278	-000.05	501.05	-54.034	-600.00	61.177
3279	-1407.1	1274.0	-22.103	-132.10	83.700
3281	-1400.1	717.05	47.001	-810.74	-10.818

NOE 1765 MO ON CONTINUOUS 1.

0000 POST1 ELEMENT STRESS LISTING 0000

LOAD STEP 1 ITERATION 1 SECTION 1
TIME 0.10000 LOAD CASE 1

ELEM	S01R	S02	S0V1	S02R	S0V2
3270	-1051.4	1035.6	-50.000	-305.70	100.00
3271	-1051.4	1035.6	-50.000	-305.70	100.00
3272	-0003.3	630.00	-180.00	-440.00	800.00
3273	-0003.3	630.00	-180.00	-440.00	800.00
3274	-0003.3	630.00	-180.00	-440.00	800.00
3275	-0003.3	630.00	-180.00	-440.00	800.00
3276	-0003.3	630.00	-180.00	-440.00	800.00
3277	-0003.3	630.00	-180.00	-440.00	800.00
3278	-0003.3	630.00	-180.00	-440.00	800.00
3279	-0003.3	630.00	-180.00	-440.00	800.00
3280	-0003.3	630.00	-180.00	-440.00	800.00
3281	-0003.3	630.00	-180.00	-440.00	800.00
3282	-0003.3	630.00	-180.00	-440.00	800.00
3283	-0003.3	630.00	-180.00	-440.00	800.00
3284	-0003.3	630.00	-180.00	-440.00	800.00
3285	-0003.3	630.00	-180.00	-440.00	800.00

NOE 1765 MO ON CONTINUOUS 1.

0000 0001 ELEMENT STRESS LISTING 0000			
LOAD STEP	ITERATION-	SECTION-	
TIME, 0.10000	LOAD CASE, 1		
ELEM	SIZE	SIZE	SIZE
2300	872.00	100.5	200.05
2301	867.97	107.14	-24.208
2302	861.64	171.5	-0.800
2303	100.21	-00.26	-284.48
2304	100.21	-00.26	1145.7
2305	1.1774	-105.24	132.77
2306	-0.0015	-00.23	70.100
2307	84.448	-245.27	86.114
2308	1.0001	-072.06	87.977
2309	30.448	-000.20	197.76
2310	76.000	-114.00	130.84
2311	00.274	-030.28	244.11
2312	304.50	-035.04	-043.93
2313	3776.7	0.07775*00	0.28700*00
2314	3776.7	0.08200*07	0.18170*00
2315	3776.7	0.08200*07	0.18170*00
2316	3776.7	0.08200*07	0.18170*00
2317	3776.7	0.08200*07	0.18170*00
2318	3776.7	0.08200*07	0.18170*00
2319	3776.7	0.08200*07	0.18170*00
2320	3776.7	0.08200*07	0.18170*00
2321	3776.7	0.08200*07	0.18170*00
2322	3776.7	0.08200*07	0.18170*00
2323	3776.7	0.08200*07	0.18170*00
2324	3776.7	0.08200*07	0.18170*00
2325	3776.7	0.08200*07	0.18170*00
2326	3776.7	0.08200*07	0.18170*00
2327	3776.7	0.08200*07	0.18170*00
2328	3776.7	0.08200*07	0.18170*00
2329	3776.7	0.08200*07	0.18170*00
2330	3776.7	0.08200*07	0.18170*00
2331	3776.7	0.08200*07	0.18170*00
2332	3776.7	0.08200*07	0.18170*00
2333	3776.7	0.08200*07	0.18170*00
2334	3776.7	0.08200*07	0.18170*00
2335	3776.7	0.08200*07	0.18170*00
2336	3776.7	0.08200*07	0.18170*00
2337	3776.7	0.08200*07	0.18170*00
2338	3776.7	0.08200*07	0.18170*00
2339	3776.7	0.08200*07	0.18170*00
2340	3776.7	0.08200*07	0.18170*00
2341	3776.7	0.08200*07	0.18170*00
2342	3776.7	0.08200*07	0.18170*00
2343	3776.7	0.08200*07	0.18170*00
2344	3776.7	0.08200*07	0.18170*00
2345	3776.7	0.08200*07	0.18170*00
2346	3776.7	0.08200*07	0.18170*00
2347	3776.7	0.08200*07	0.18170*00
2348	3776.7	0.08200*07	0.18170*00
2349	3776.7	0.08200*07	0.18170*00
2350	3776.7	0.08200*07	0.18170*00
2351	3776.7	0.08200*07	0.18170*00
2352	3776.7	0.08200*07	0.18170*00
2353	3776.7	0.08200*07	0.18170*00
2354	3776.7	0.08200*07	0.18170*00
2355	3776.7	0.08200*07	0.18170*00
2356	3776.7	0.08200*07	0.18170*00
2357	3776.7	0.08200*07	0.18170*00
2358	3776.7	0.08200*07	0.18170*00
2359	3776.7	0.08200*07	0.18170*00
2360	3776.7	0.08200*07	0.18170*00
2361	3776.7	0.08200*07	0.18170*00
2362	3776.7	0.08200*07	0.18170*00
2363	3776.7	0.08200*07	0.18170*00
2364	3776.7	0.08200*07	0.18170*00
2365	3776.7	0.08200*07	0.18170*00
2366			

BESS POST ELEMENT STRESS LISTING 2020									
LOAD STEP	ITERATION-	SECTION- 1	LOAD CASE- 1	922	921	922	922	922	922
TIME- 0.12000									
1120									
1304	3750.7	0.05500E+00	-0.2728E+00	0.00000E+00	-0.2350E+07	-0.2350E+07	-0.2350E+07	-0.2350E+07	0.3550E+07
1400	3750.3	0.00400E+07	-0.2350E+07	-0.2323E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
1500	3771.4	-0.2213E+00	0.07420E+00	-0.2304E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
1607	3774.0	0.00000E+00	0.16020E+00	-0.1751E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
1700	4073.0	0.0181E+00	0.2324E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
1800	4074.4	0.00000E+07	-0.4007E+00	-0.2324E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
1910	4075.0	-0.2334E+00	0.1214E+00	-0.2300E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
2011	4077.2	-0.2800E+00	0.1700E+00	-0.2647E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
2113	4079.5	-0.2647E+00	0.1755E+00	-0.1400E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
2216	4079.0	0.1400E+00	-0.7400E+07	0.1501E+00	-0.2000E+00	-0.2000E+00	-0.2000E+00	-0.2000E+00	-0.2000E+00

0167-28.0
K1-4E
V1-88.0
24-92.0
X10-12.0
V14-01A

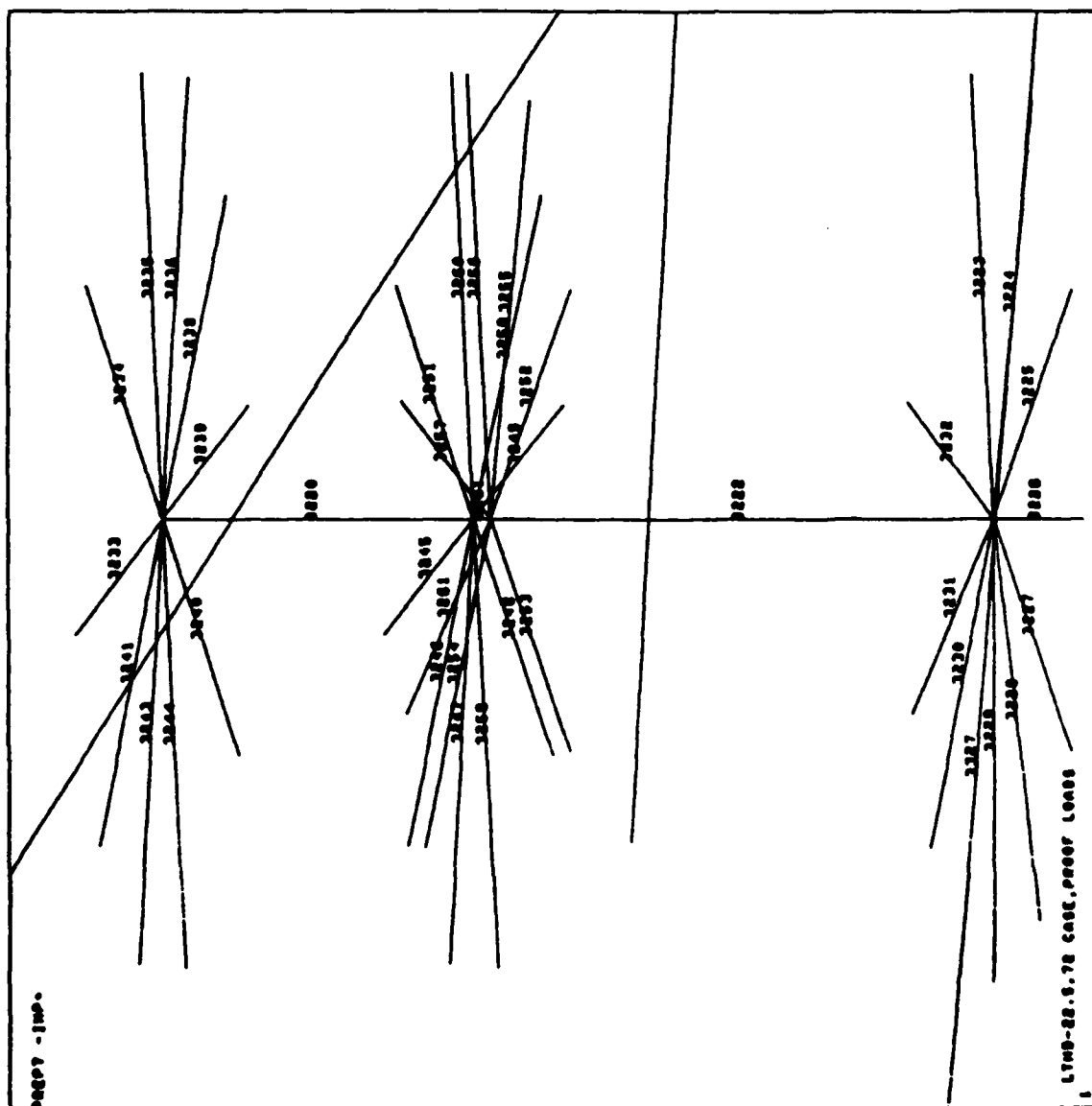
15:



ARS 4.00
 JAN 83 1987
 11:13:41
 PROXY ELEMENTS
 EMBL-1

200H
 200-1
 200-1
 20-1
 0107-20.6
 27-148
 27-06.0
 27-02.6
 270-10.8
 270-4.18

LOWER
 SHAFT
 ELEMENTS

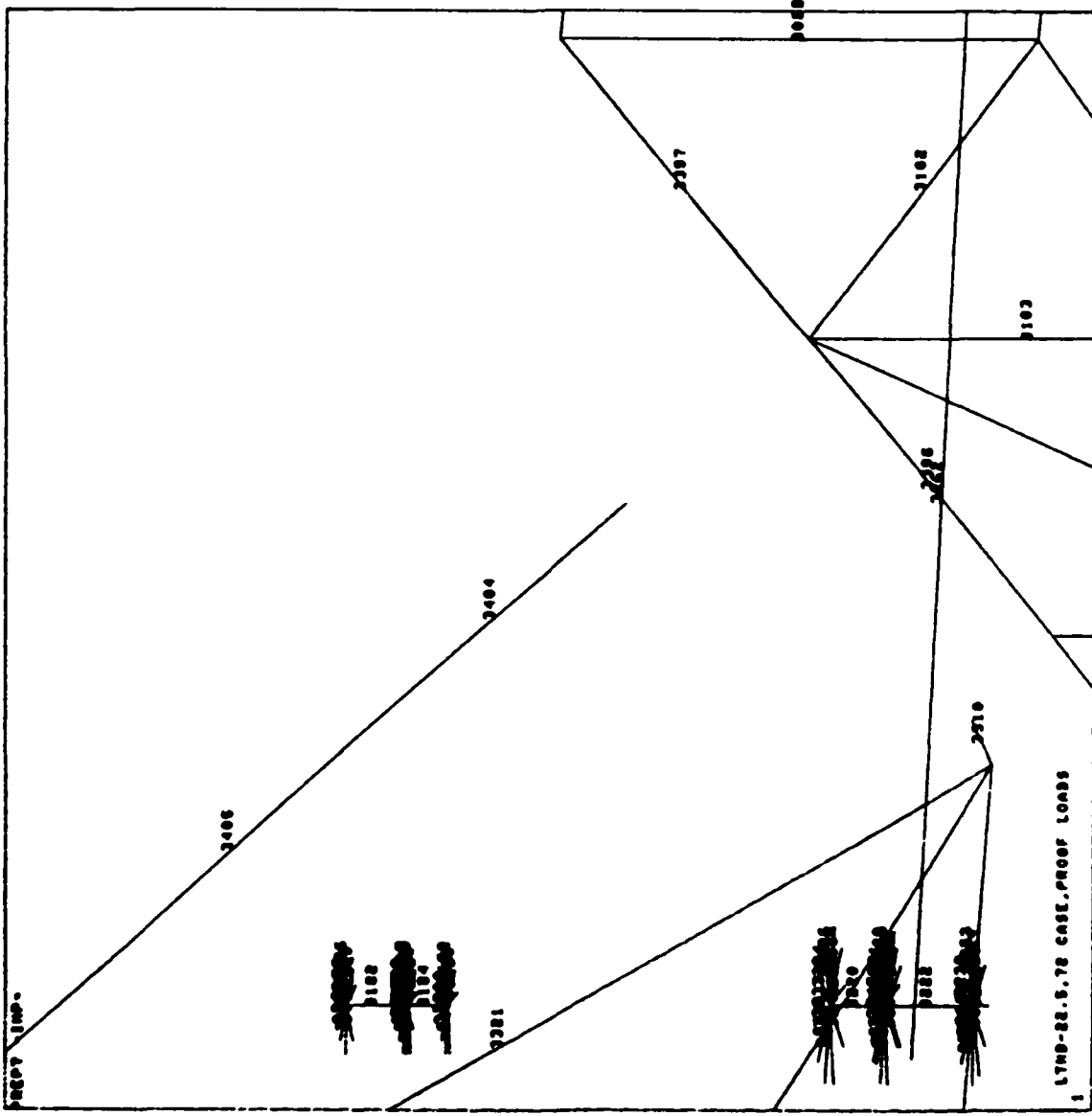


0107-27
X7-125
V4-130
ZF--75.0
X070-10.4
V070-6.51

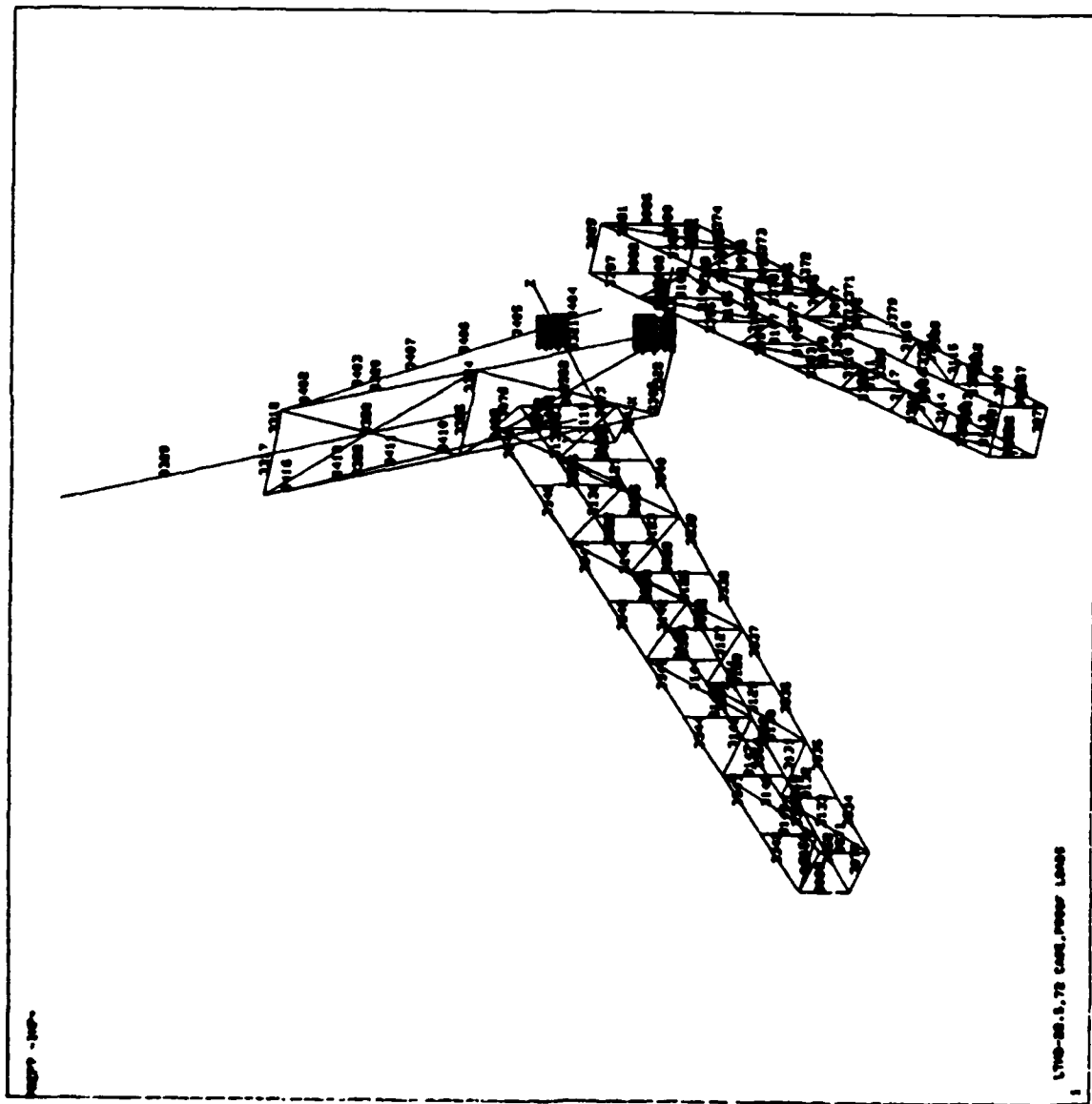
LYND-28.6.72 CASE PROOF LOADS

ANSYS 4.20
 JAN 23 1987
 11:15:44
 PREPT ELEMENTS
 ENUR-1

200N
 XV--1
 YU--1
 ZV--1
 0 D157-805
 0 XF-130
 0 YF-120
 0 ZF--72
 XRTD-19.4
 YRTD-6.51



00000 4.00
 JUN 03 1987
 11:10:04
 PREP7 ELEMENTS
 ENUS-1
 100--1
 100--1
 20--1
 0107-011
 07-00-0
 10-130
 07--100
 0070-4.00
 0070-3.00



D3/210

CEL MEMO: JANUARY 28, 1987

FMC Central Engineering Laboratories
Santa Clara

Interoffice

To J. Ries Date Jan. 28, 1987

From C. R. Ortloff cc E. Thuse
A. Amberg
R. Kazares
R. Rathe
L. Libhardt

Subject REMAINING TASKS ON THE LVHD PROJECT -
COST/BENEFIT ANALYSIS FOR SEVERAL PROPOSED TASK

To date (1-28-87) about \$43K out of \$65K authorized has been expended for labor charges on system analysis tasks. Allowing for a \$15K expenditure by M. Rodemaker for cradle analysis tasks, and a \$5K recapture from C. C. Chen, about \$12K remains, without consideration of computer charges. Retroactive computer charges starting 1-1-87 essentially reduce the remaining budget to a few thousand dollars. Further clarification is needed on the start date for these charges as well as an understanding on final management agreements on general issues regarding computer charges past 1-28-87. *

From a phone conversation with you (JR) on 1-28-87, the request was made to list important analysis tasks remaining together with cost estimates (including labor and computer charges). Accordingly, these tasks are listed below:

1. Computation of the effects of a "soft soil" model on system stress levels and stability. Linear springs will be input between the spade and ground to model the rearward motion of the system as the spade compresses the "soft soil" under recoil and torque impulse loads applied to the cradle.

*No
geom.
change.*

The 00-00 or 22.50-00 FE system model will be used as the rearward reaction force is largest for these cases. Stress contour plots for the gimbal and platform for "soft soil" and "hard soil" spade emplacement will be made and compared to assess the magnitude of the difference. Results will determine the amplification/diminition effect of the soil model on stress levels. Results will be reported in memo form.

*one - lum
- no
instab. h*

Labor: (Set-up, Report & Postprocessing time) 25 hours = \$2,000
Computer charges: 30 CPU hours @ \$125/hour = \$3,750
Total: \$5,750

2. Modification of the existing FE system model to incorporate the next design iteration in the platform and gimbal structure (plus a possible update in trail and cradle "beam" representations). The magnitude of this task depends on the degree of FEM modification required. An estimate of (say) one week (labor) for model modification and a computer run for the "worst load case" is as follows:

*need
form
out*

J. Ries
Remaining Tasks on the LVHD Project
Cost/Benefit Analysis for Several Proposed Tasks

January 28, 1987
Page 2

*probably
more
than
this*

Labor: (Model Modification, Report & Postprocessing) 54 hrs. = \$4,400
Computer Charges: 50 CPU hours @ \$125/hour = \$6,250
(includes average of interactive and batch run times)
Total: \$10,400

Totals for this task may be higher if FE model changes are very significant (or lower if only local thickness changes are prescribed).

3. Computation of static 16g LAPES loads on a system model with folded trails.

*old
geometry*

Labor: (Model Modification, Postprocessing & Final Memo) 30 hours = \$2,400
Computer Charges: 30 CPU hours @ \$125/hour = \$3,750
Total: \$6,150

No

For the above, a run involving towing loads could be substituted for the same cost.

4. Final report (includes results of Item 2)

Labor: 60 hours = \$5,000

yes

*old
geom*

5. Detailed multilamina FE model of the new cradle design and layer-by-layer analysis of stresses, deflections and failure mechanisms. This run will be a static analysis and dynamic amplification factors from M. Rodamaker's analysis will be used to obtain maximums of the dynamic stress state. Use of part of these computed results can be used in an inertial buckling analysis to determine cradle stability under impulse loads. The new FE model will be for the current cradle geometry, cable mount and (new) materials selection.

No.

Labor: (includes new FE model, postprocessing time and final memo) 60 hours = \$ 4,800
Computer charges: 60 CPU/hours @ \$125/hour = \$ 7,500
Total \$12,300

The above tasks 1, 2 and 4 total \$11,400 for labor and \$12,000 for computer charges and probably constitute the minimum work package necessary to assure that the new system design is adequate. Tasks 3 and 5 are optional but serve to increase confidence in the adequacy of the design.

C. R. Ortloff

C. R. Ortloff

CRO/870128/01

*Recommend do 2, and 4.
My estimate*

~ \$17K

Have 2 due about mid-Feb

As an addendum to this memo, a few additional comments on Task 5 are in order. As in prior cradle analyses, a detailed FE model needs to be made for this task. Use of ANSYS STIF 53 (3 node composite elements) and STIF 45 (8 node isoparametric elements) will reduce run times considerably. Use of the maximum stress failure criteria will be continued for the following reason: Since the Tsai-Hill failure ellipse lies within the maximum stress rectangle in stress space, if the structure fails based on maximum stress criteria, it will certainly fail based on Tsai-Hill criteria. Thus, use of the maximum stress criteria gives some margin of conservatism in the design. *

One remaining concern with the cradle is local buckling (or crippling) of the edge-loaded Gr/Ep sandwich structure near the forward manifold under proof impulse loads. Some computer work can be done in this task to investigate this effect. Certainly, this problem should be addressed in the test phase of future work on the cradle. Since ARDEC appears interested in tests/calculations that are at hot, wet composite conditions, the analysis tasks can easily include these effects and uncover problems that may be missed in the test phase unless environmental test chambers are used. Additional tasks related to thermal expansion deformation and stress levels can also be addressed by this task item.

Carroll

D3/220

CEL MEMO: FEBRUARY 12, 1987

Interoffice

To: L. Libhardt*

From: C. R. Ortloff

Subject: **STATIC PROOF LOAD ANALYSIS OF
THE LWED SYSTEM MODEL (22.5°-72° LOAD CASE)**

Date: Feb. 12, 1987

cc: E. Thuse
A. Amberg
R. Kazares
D. Langerud
R. Rathe
J. Ries

Attached are Figures 602-623* detailing the equivalent stress and UX, UY, UZ deflections of the gimbal and platform under statically applied proof loads. The recoil and firing torque maximum load vectors are decomposed into X, Y and Z components and applied at their normal point(s) of application on the beam representation of the cradle. In this manner, the subsidiary effects of the cradle "moment arm" are preserved with respect to load transmittal into the gimbal/platform and spade. For the present run, the trail ends are subject to a UY=0 constraint based on the fact that these ends rest on the ground while static loads are applied. An additional 1g gravity load is applied uniformly to all discrete and continuously distributed masses in the model. The spade is fixed at its lower edge while the horizontal spade plate is subject to UY=0 constraints.

Stresses shown are for the TOP (outside), MIDDLE and BOTTOM (inside) surface planes of STIF 43 plate elements used to compose the FE model. The SIGE stress follows the definition

$$\sigma_E = \frac{1}{\sqrt{2}} \sqrt{(\sigma_{xx} - \sigma_{zz})^2 + (\sigma_{xx} - \sigma_{yy})^2 + (\sigma_{yy} - \sigma_{zz})^2 + 6(\tau_{xy}^2 + \tau_{xz}^2 + \tau_{yz}^2)}$$

where σ_{ij} denotes normal stress and τ_{ij} denotes shear stress. The plasticity criteria used is $\sigma_E = 120$ ksi; this value denotes the boundary between elastic and plastic regions (for which $\sigma_E > 120$ ksi). The coordinate system used is shown in Figure 609. All displacements are with respect to global axes. Note that some of the UZ displacements toward the top part of the platform are large. As the lower arms of the gimbal absorb all of the proof recoil loads, a large moment (M_x) is input into the platform causing bending in the -Z direction. Note also that deflections are calculated on the basis of linear elasticity even though stresses may exceed the yield stress value. Actual deflections, calculated on the basis of plasticity theory, would be larger than elastic values shown (although stresses would be less than those shown).

Comparison of results from this load case to that run on the basis of dynamic input load (Memo: CRO to R. Rathe, 1-26-87) gives some indication as to dynamic amplification factor for stress and deflection levels.

L. Libhardt
Static Proof Load Analysis of
the LVHD System Model (22.5₀-72⁰ Load Case)

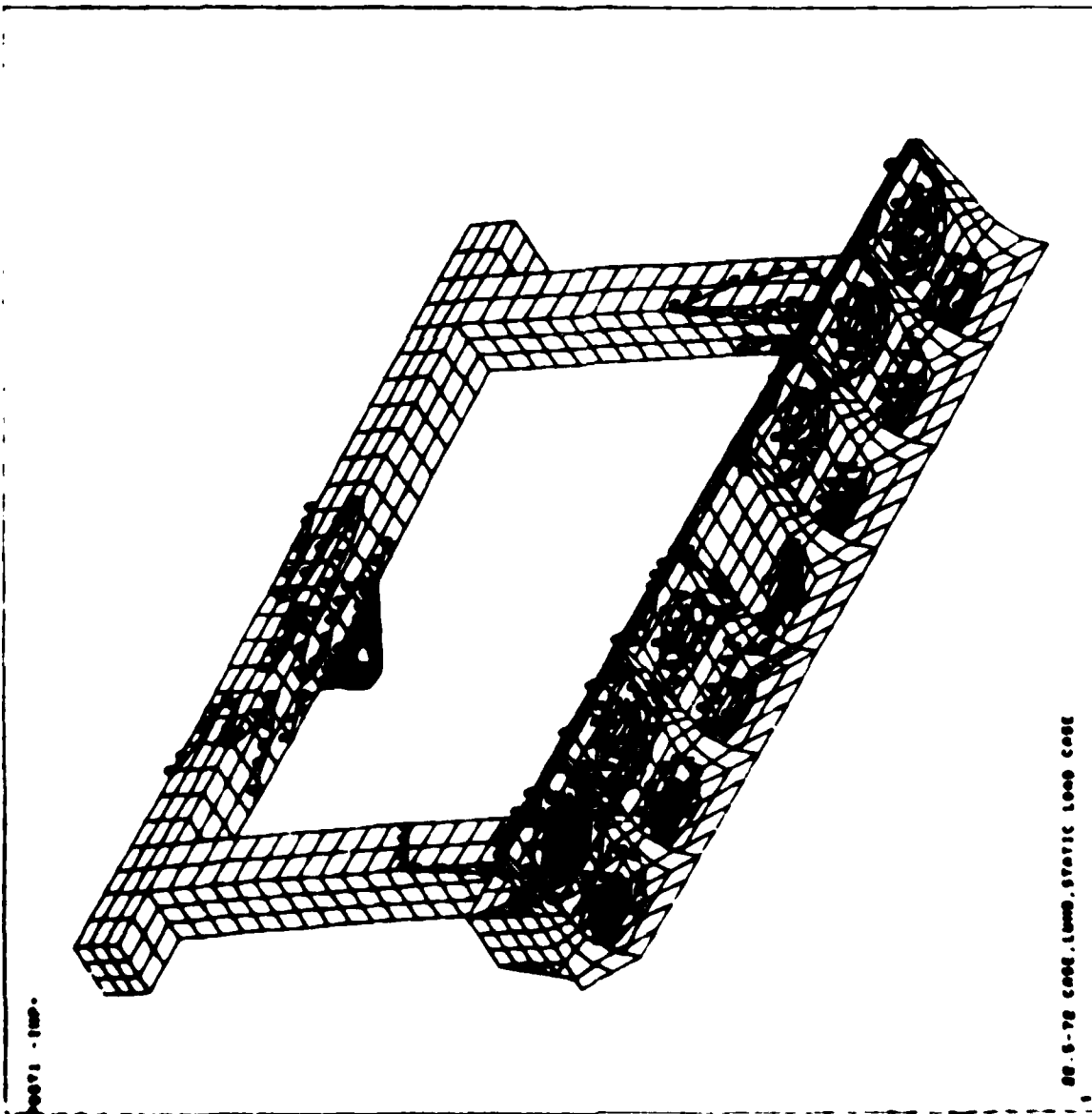
Feb. 12, 1987
Page 2

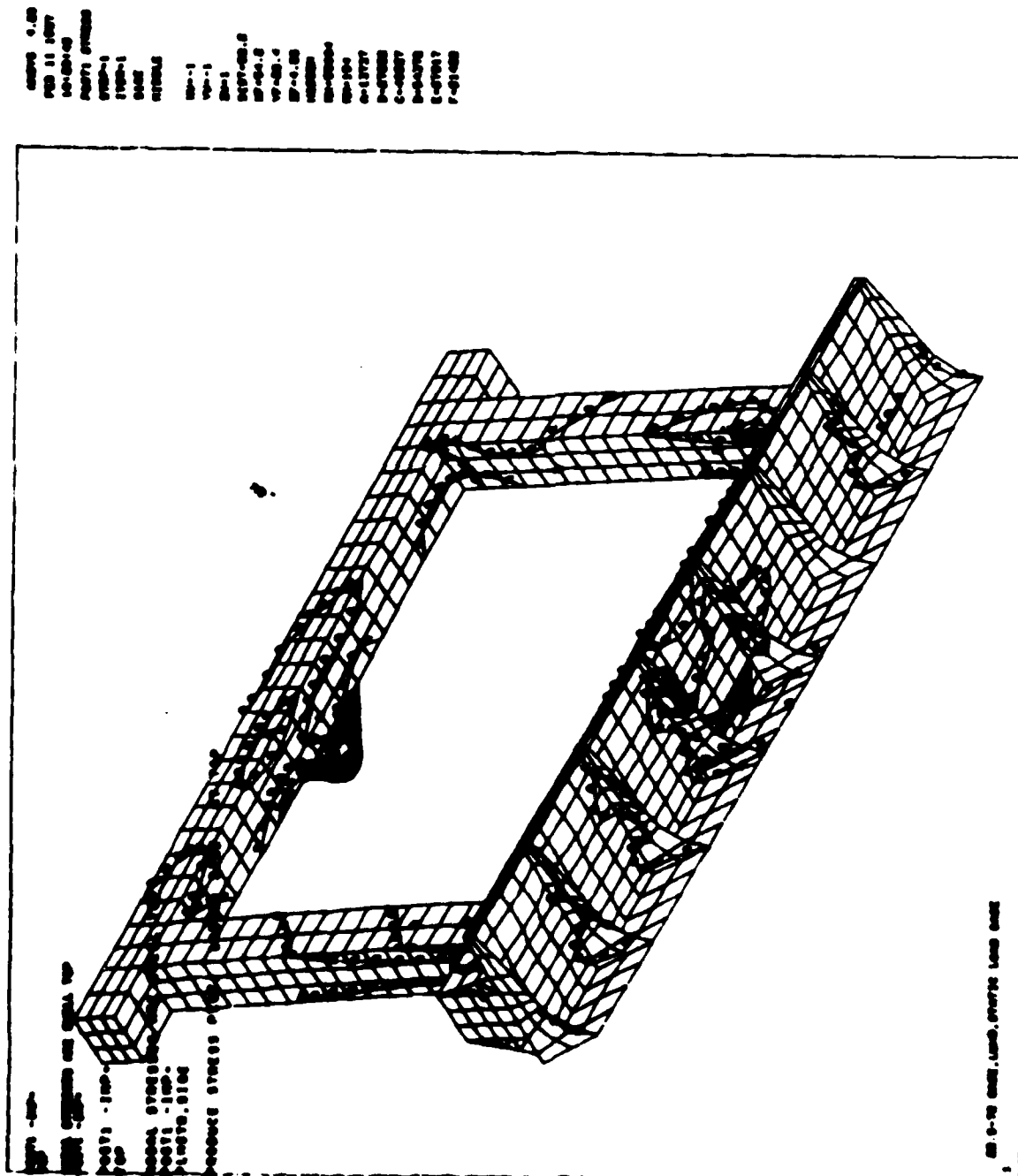
As a further note on thermal expansion effects in the Gr/Ep cradle, it should be noted that the "through the thickness" thermal expansion coefficient can be 20-50X the in-plane thermal expansion coefficient. It may be important to include these effects and examine stress and deflection results in the vicinity of ply-drop-off regions and fillet regions where the Gr/Ep layers transition into "thick walls."



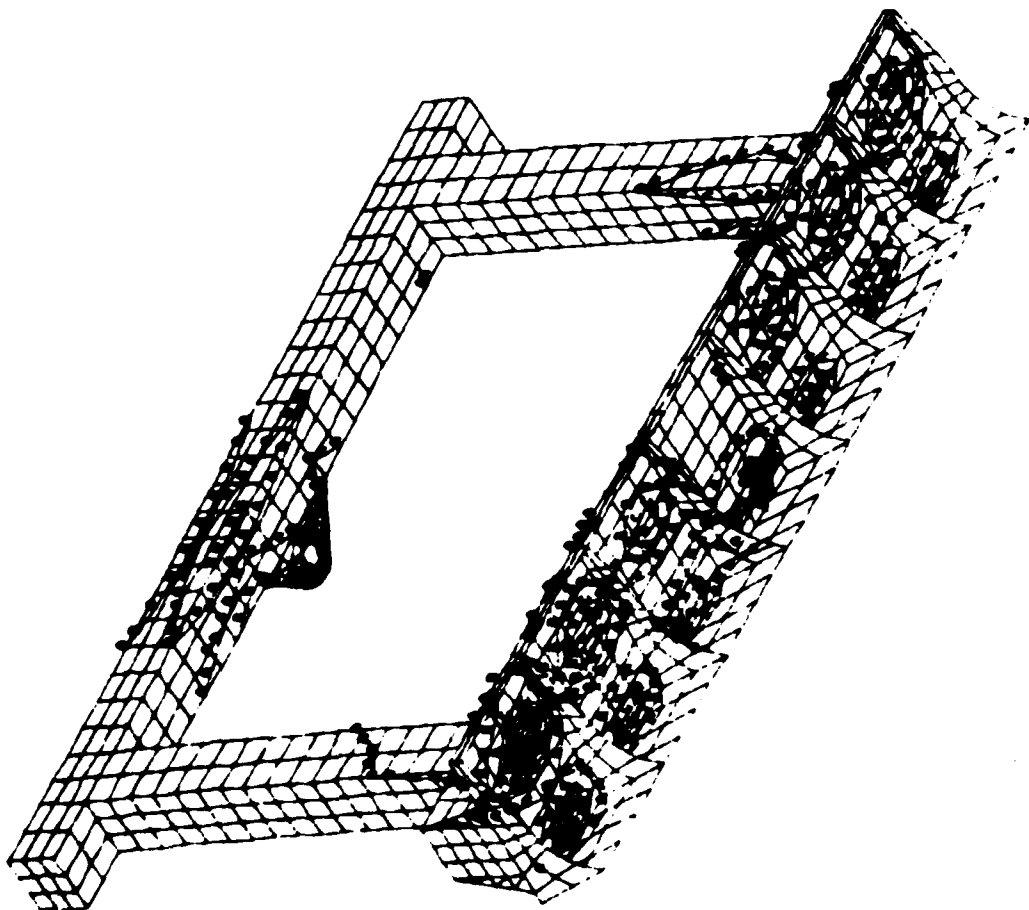
C. R. Ortloff

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 FEB 11 1967
 10:30:47
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 0700-1
 1700-1
 0100
 700
 20-1
 30-1
 20-1
 0107-00.0
 27-00.0
 27-00.0
 27-00.0
 27-00.0
 010000
 01-170000
 00-000
 0-00000
 0-01700
 0-17000
 0-100000
 0-100000
 0-100000
 0-100000





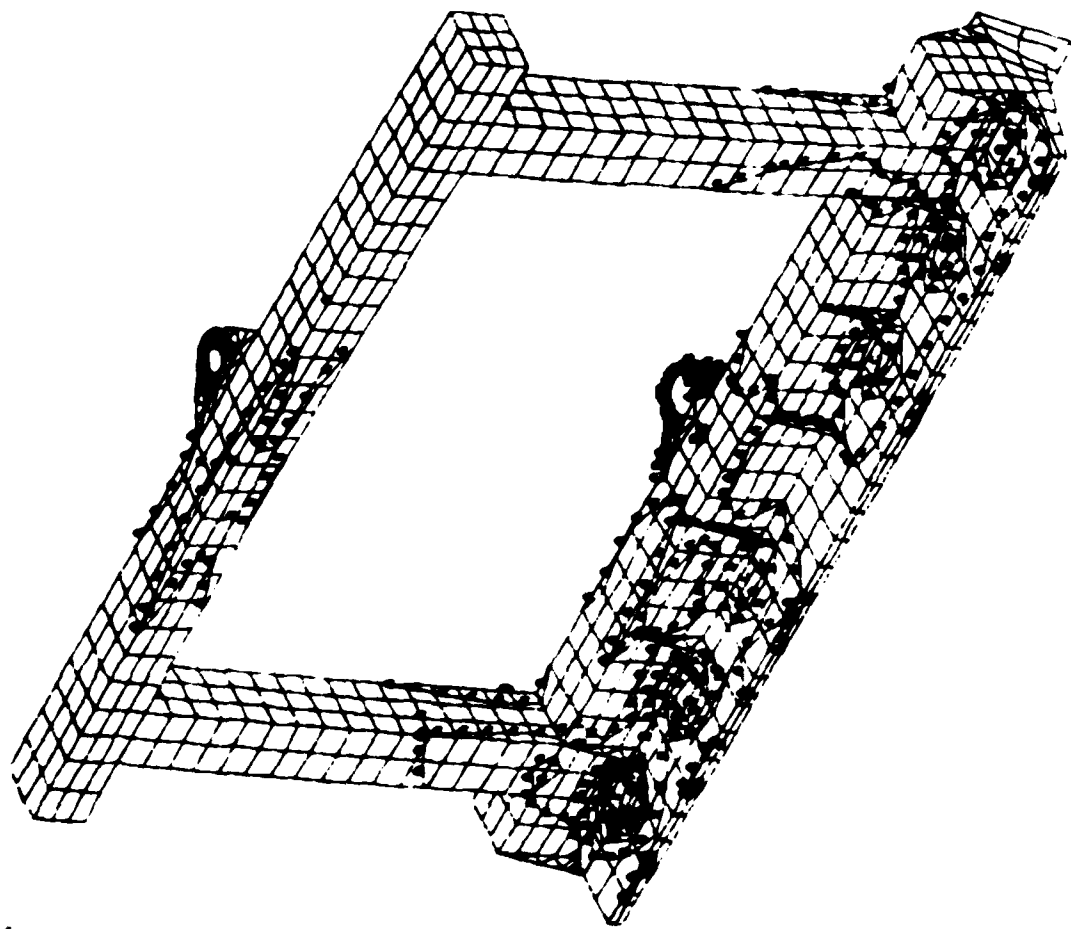
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 SEP 11 1987
 10:07:55
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 STEP=1
 ITEM=1
 SLOC
 BOTTOM
 MU=1
 VU=1
 20=1
 0101-00.2
 WF-04.2
 VF-06.4
 27-04.05
 MIDDLE
 MM-10104
 MM-556
 0-00453
 0-00352
 C-70261
 0-104150
 E-120040
 F-155040



POST1 - IMP.

22.5-72 CASE, LUMP, STATIC LOAD CASE

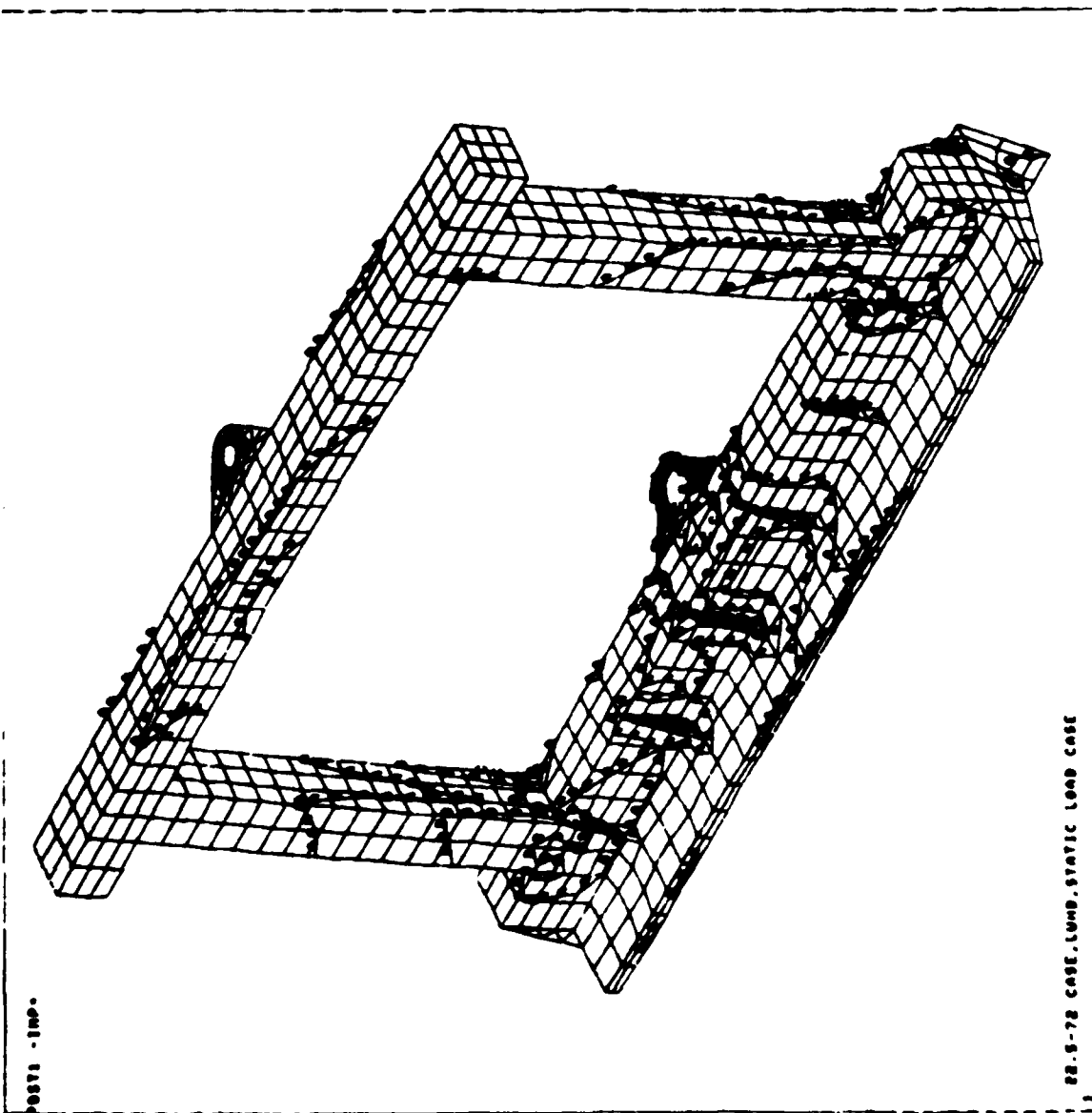
ABOVE 4.00
 FEB 11 1987
 11:04:30
 POST1 STRESS
 STEP=1
 ITER=1
 STRESS
 TOP
 20=1
 20=1
 20=1
 5157-55.8
 27-51.2
 27-27.7
 27-4.17
 MIDDLE
 22-170004
 22-464
 4-26004
 8-51726
 C-77358
 9-102000
 C-120622
 F-154854



POST1 -IMP-

82.5-72 CASE.LUMP,STATIC LOAD CASE

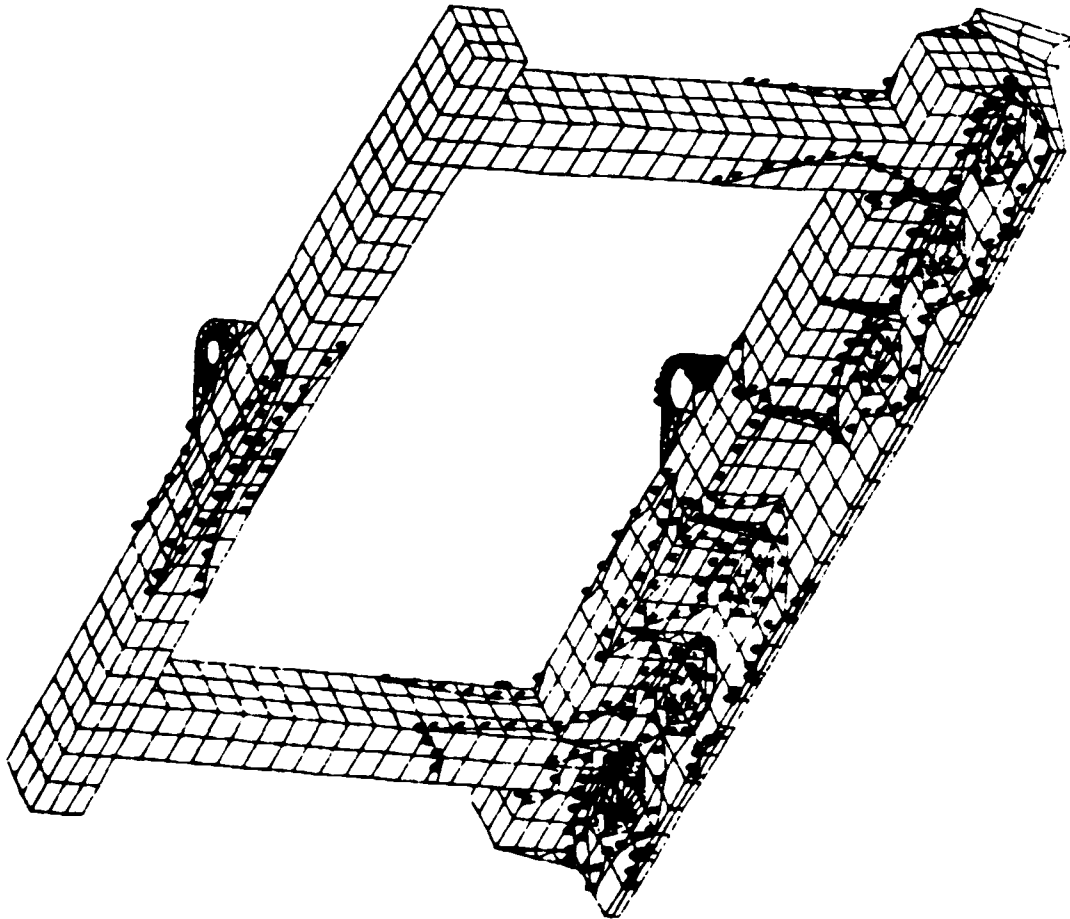
ANSYS 4.20
 FEB 11 1987
 11:18:00
 POST1 070005
 STEP=1
 LREQ=1
 SLOC
 MIDDLE
 10=1
 20=1
 3101-50.6
 37-51.8
 39-87.7
 39-4.17
 MIDDLE
 22-55004
 22-194
 4-13737
 8-27882
 8-40827
 9-54372
 9-87817
 9-81482



POST1 -IMP-

22.5-72 CASE, LUMB, STATIC LOAD CASE

ANSYS 4.80
 FEB 11 1987
 11:29:34
 POST1 STRESS
 STEP=1
 ITEM=1
 SLOC
 BOTTOM
 ZU=1
 VU=1
 ZU=1
 DIST=56.6
 XF=51.2
 YF=27.7
 ZF=4.17
 MIDDLE
 RM=181844
 MM=556
 A=28453
 B=52352
 C=78251
 D=104150
 E=120048
 F=155948



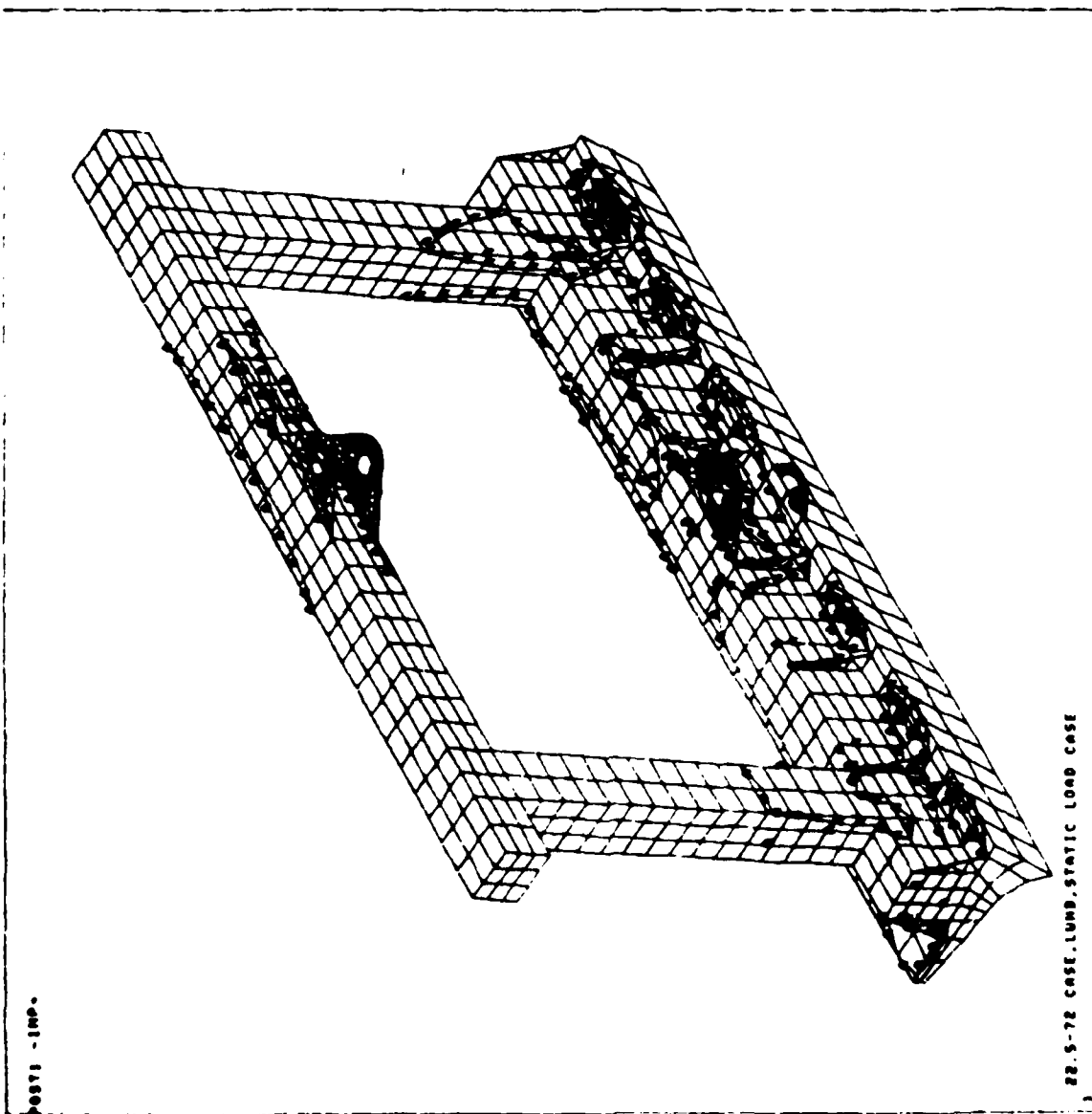
POST1 -IMP.

28.5-72 CASE, LUMP, STATIC LOAD CASE

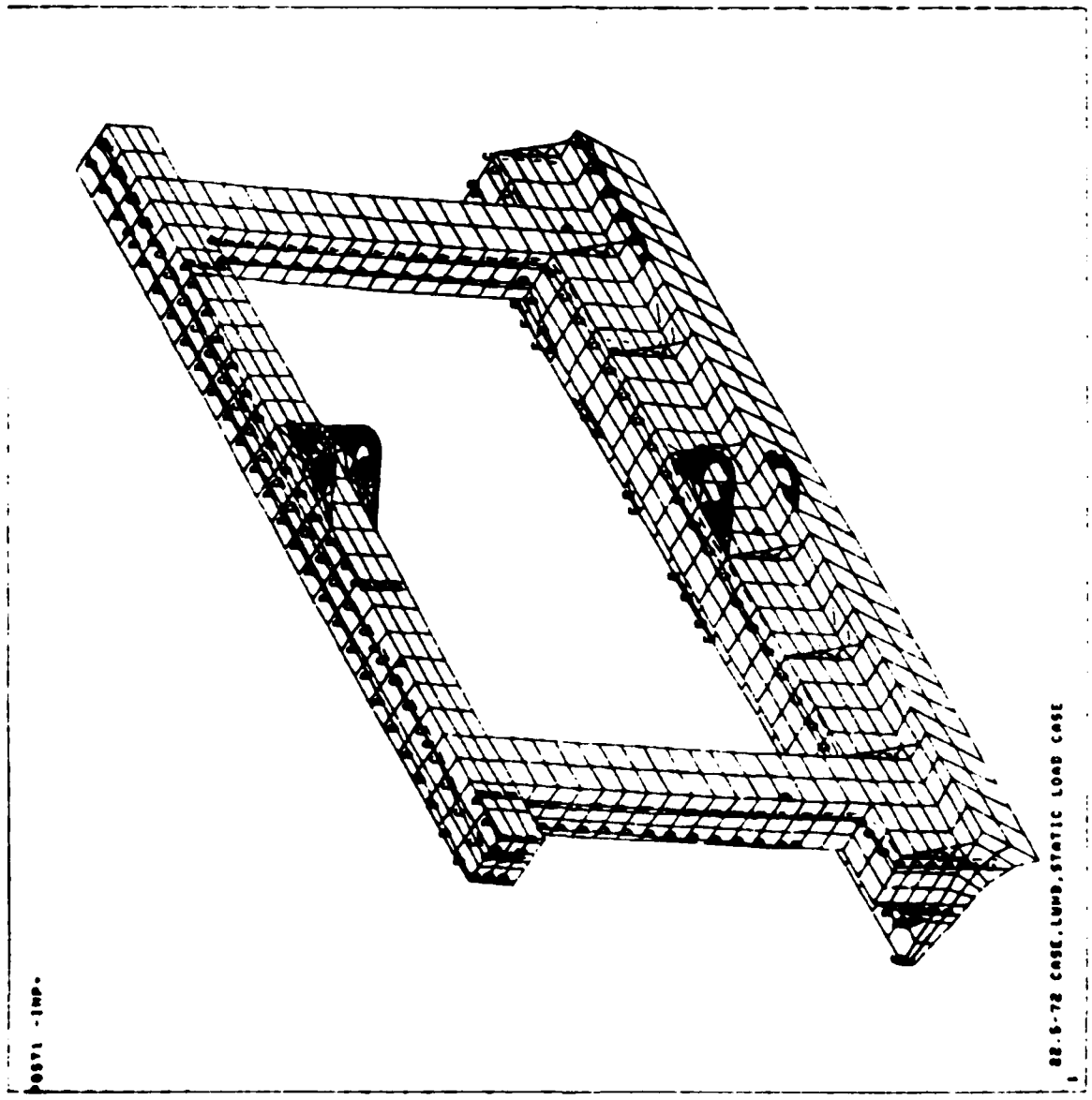
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ANSYS 4.20
FEB 11 1987
11:38:24
POST1, STRESS
STEP=1
ITER=1
SLOF
TOP
ZU=1
VU=1
ZU=-1
DISP=0.2
SF=54.2
VF=26.4
ZF=4.85
HIDDEN
RM=179804
MM=464
A=26094
B=51720
C=77350
D=102000
E=120022
F=154254

```



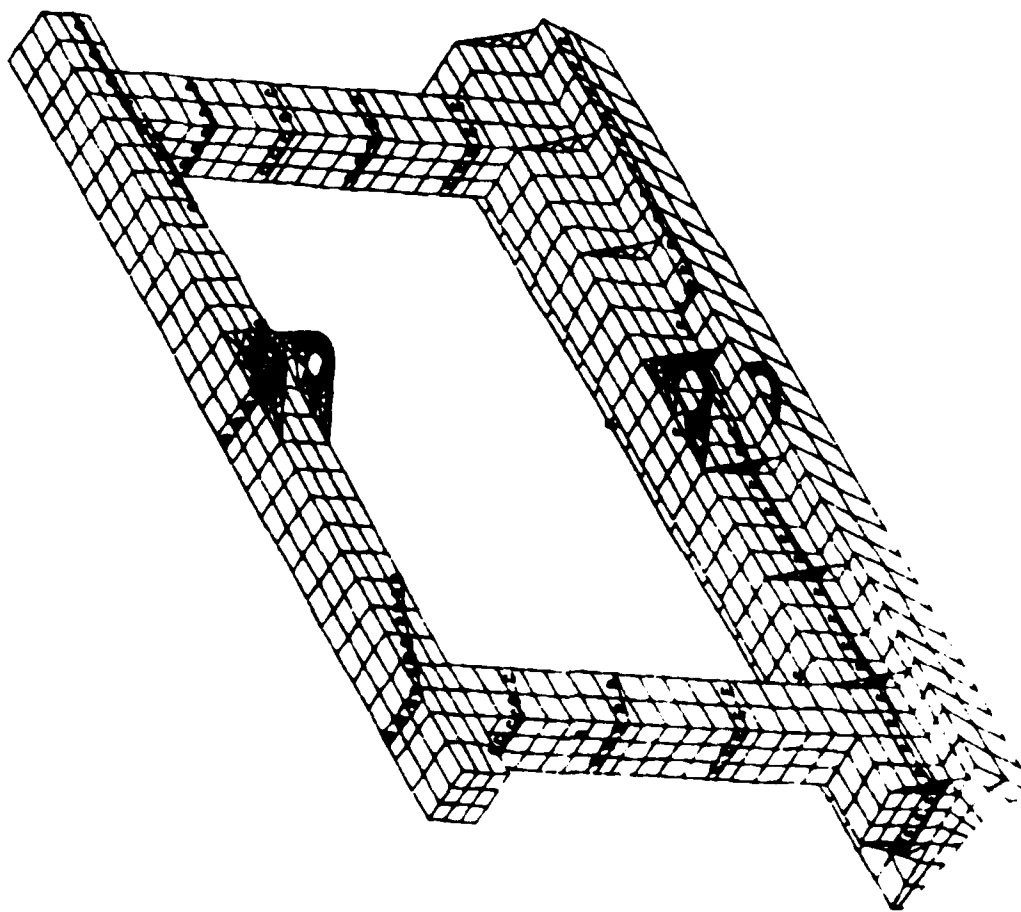
ANSYS 4.20
 FEB 11 1987
 12:27:57
 POST1 STRESS
 SECT=1
 YZ=1
 UV
 019PL MODEL
 NU=1
 VO=1
 ZV=1
 0197-68.8
 MF=64.2
 VF=20.4
 ZF=4.85
 MIDDLEM
 MX=.474
 MY=...308
 MZ=...817
 A=...0071
 C=...0000
 D=...143
 E=...863
 F=...383



```

00000 4.00
FEB 11 1987
12144147
POST1 STRESS
STEP=1
1788=1
U2
DISPL MODAL
NU=1
VU=1
20=-1
DISP=50.8
27=54.2
VF=20.4
27=4.05
MIDDM
MM=-8.00
A=-8.00
B=-8.10
C=-1.00
D=-1.10
E=-.90
F=-.10

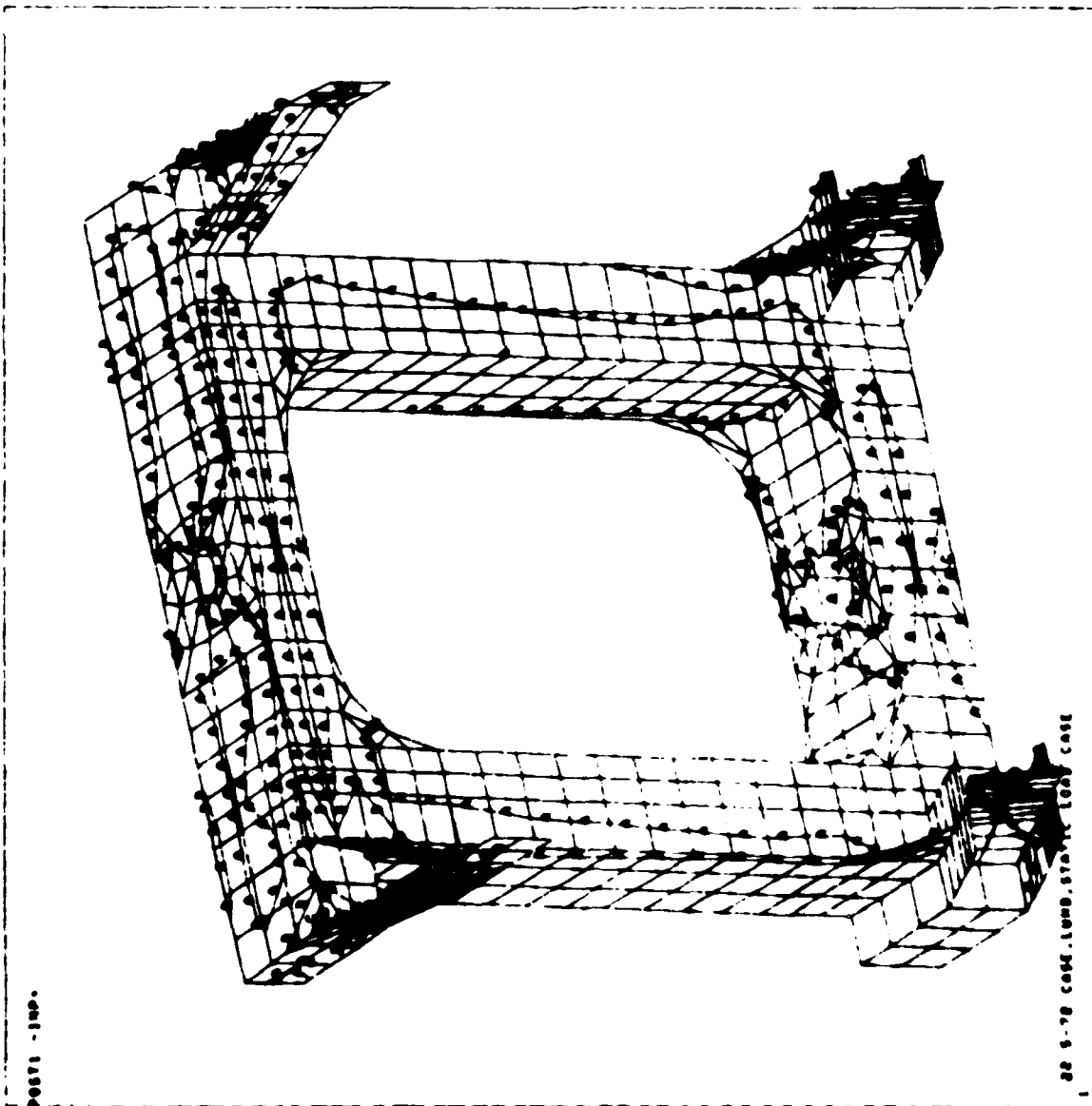
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POST1 -IMP.

22.8-72 CASE LUMP. STATIC LOAD CASE

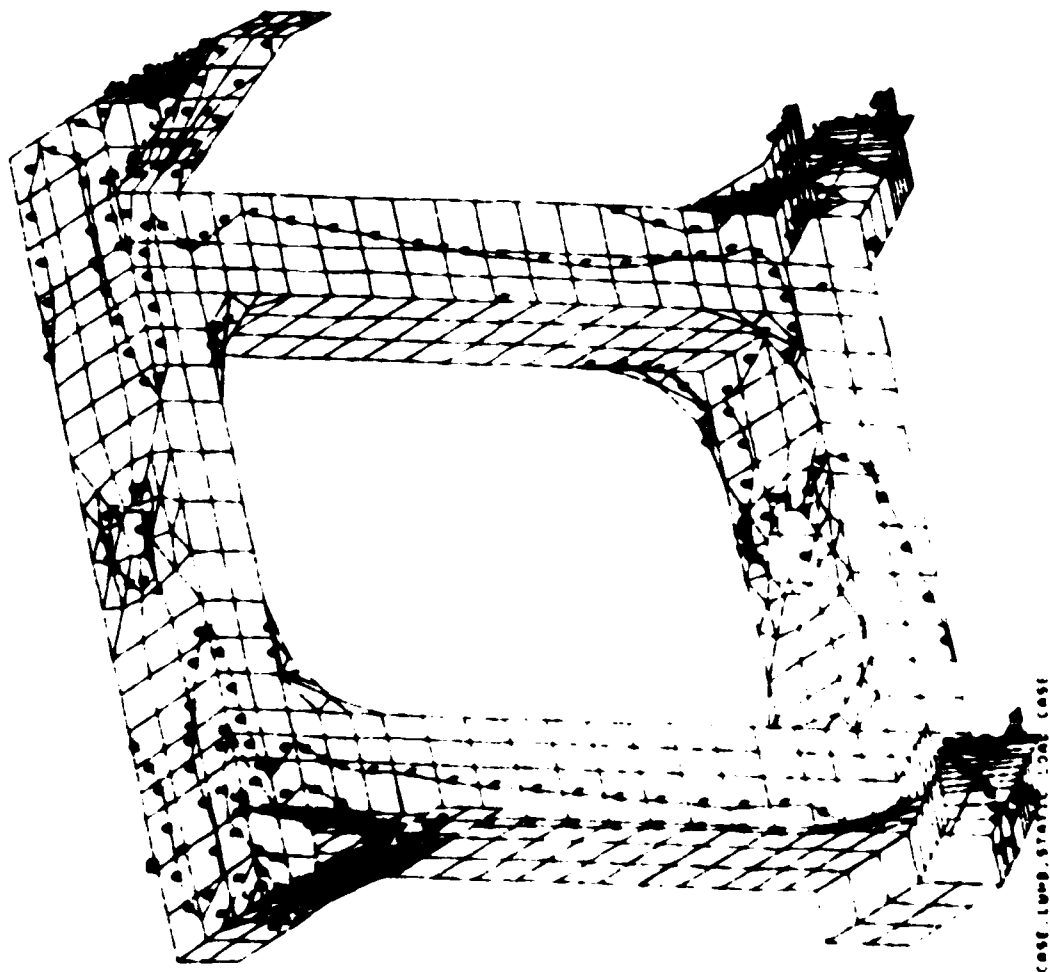
ANSYS 4.80
 FEB 11 1987
 11:47:46
 POST1 STRESS
 STRESS
 1768-1
 SLOC
 TOP
 MU-1
 VU-1
 20-1
 0197-00.7
 00-03.0
 00-36
 20-7.03
 41904m
 01-210302
 00-1070
 00-21020
 0-00007
 C-03306
 D-184106
 E-154004
 F-100023



```

ANAL 4.20
REV 11 1007
11-50-14
POST1 SYMB
STEP=1
1928-1
SIZE
MIDBLE
20-1
VAL
20-1
0187-20.7
10-53.8
17-36
27-17.83
MIDBLE
MID-210014
MID-1107
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MID-155270
MID-180000

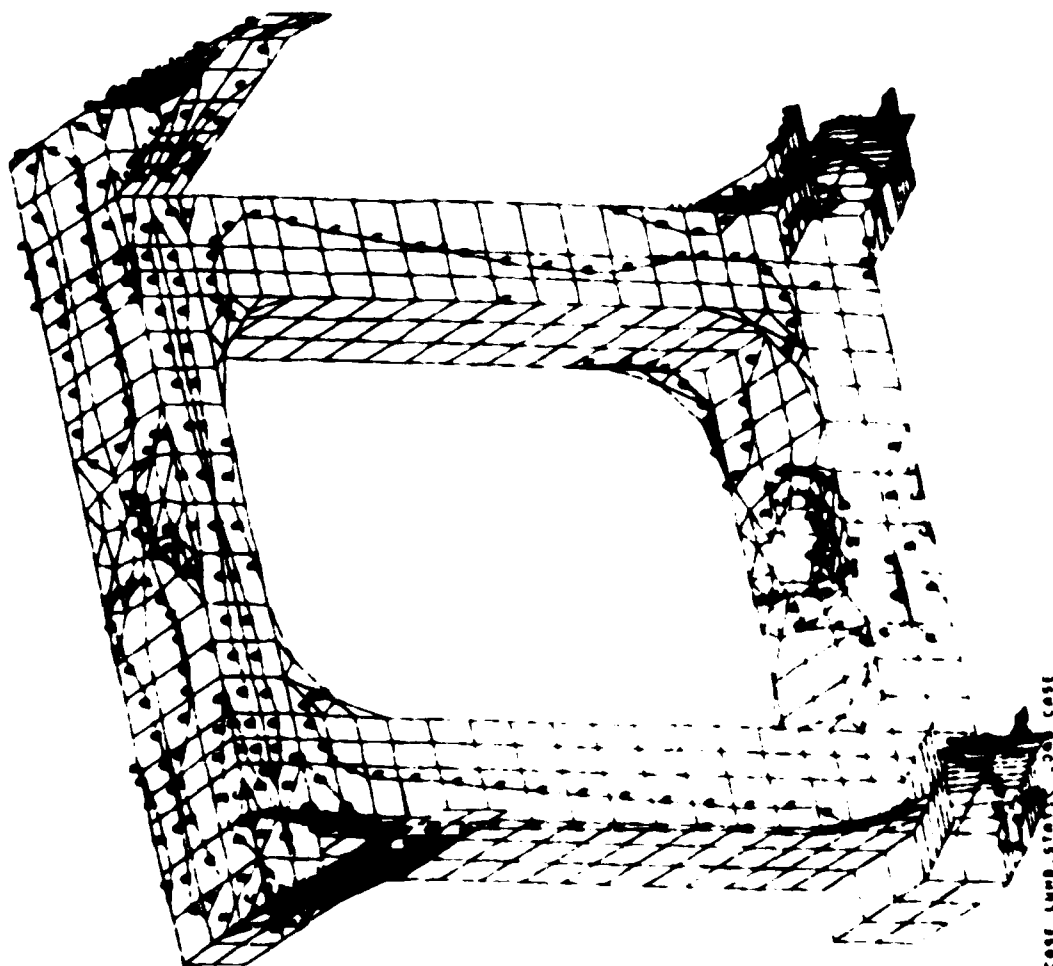
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POST1 -140.

22 5.72 CASE LUMP. STAT. CASE

00000 0.00
 000 11 1007
 12-02-100
 00001 000000
 0000-1
 1000-1
 0100
 00000
 00-1
 00-1
 20-1
 0100-00.7
 00-03.0
 00-25
 20-17.03
 010000
 00-010000
 00-1000
 00-2000
 0-03000
 0-04103
 0-100007
 0-100000
 0-100000



00001 - 1000.

22 8-72 CASE. LUMB. STAY. 30 CASE

100-100000

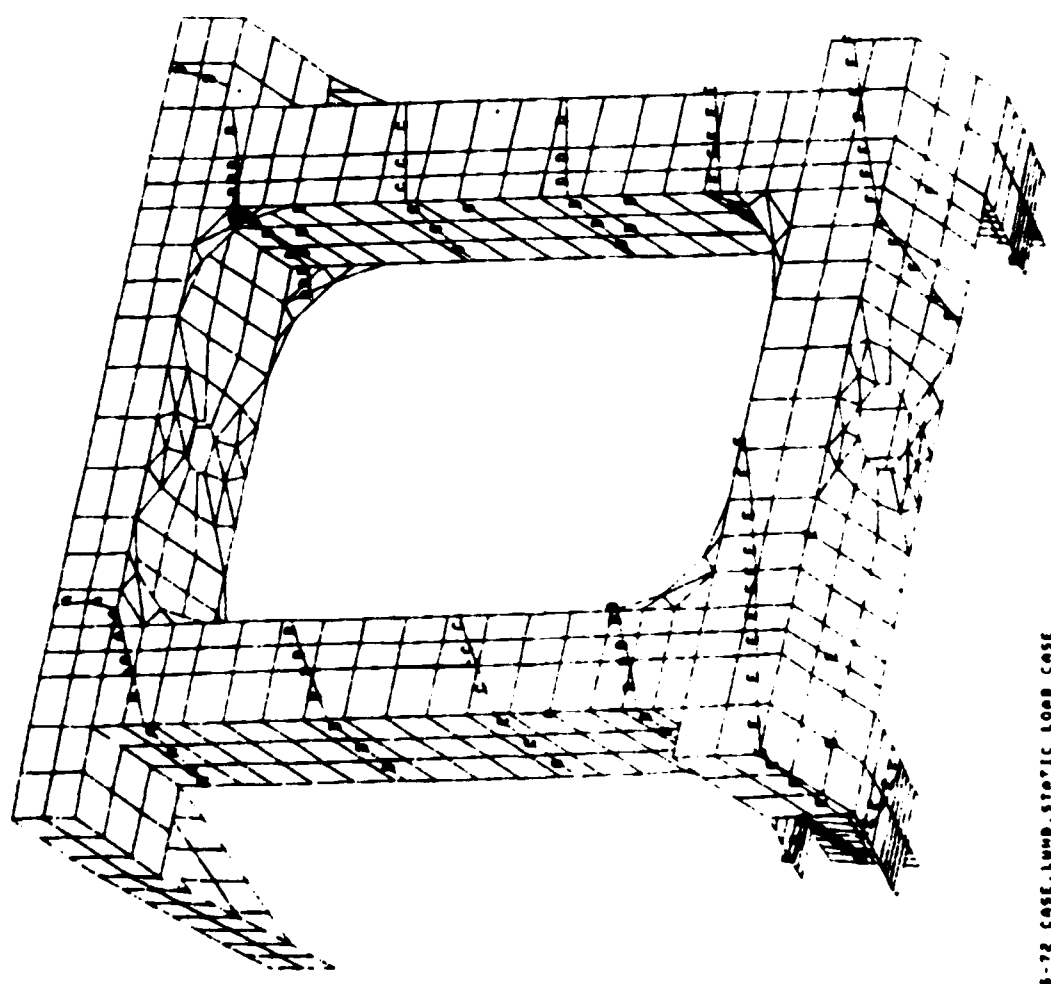
22 8 72 (001, 1640, 5707; 1 008 - 051

615


```

ANSYS 4.20
FEB 11 1987
12:20:25
POST1, STRESS
STEP=1
ITEM=1
U2
DISPL MODAL
MU=1
VU=1
ZU=1
DIST=29.7
NF=53.8
VF=35
ZF=7.83
HIDDEN
RM=0.0178
RN=3.11
A=8.88
B=8.28
C=1.82
D=1.22
E=0.816
F=0.316

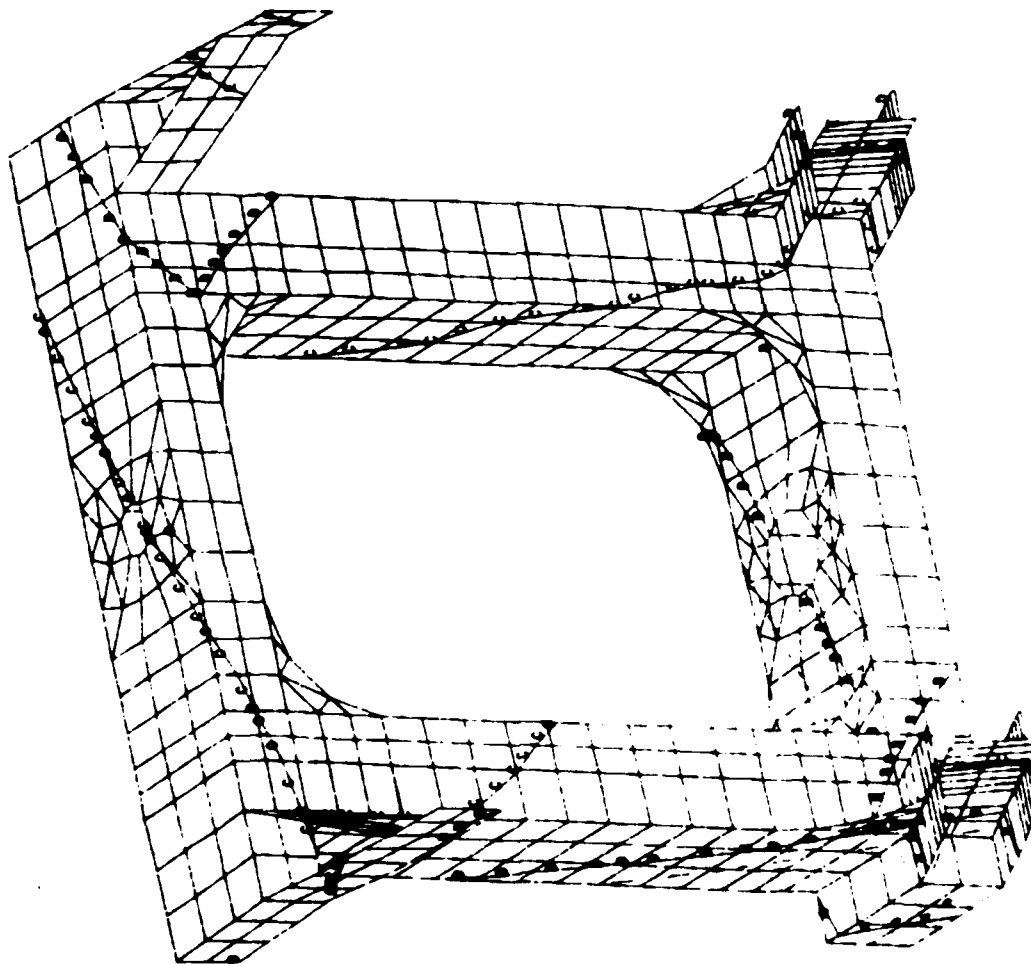
```



POST1 - 1200

22.5-72 CASE, LUMB, STATIC LOAD CASE

ANALYSIS 4.00
 CLO 11 1987
 12/23/01
 POST1 STRESS
 STEP=1
 1700=1
 UX
 DISPL NODAL
 NU=1
 VU=1
 20=1
 0157-80.7
 MF=83.9
 VF=35
 2F=7.93
 M1000M
 MX=.047
 MY=-.220
 MZ=-.141
 0-.050
 C-.350
 0-.450
 C-.050
 F-.050



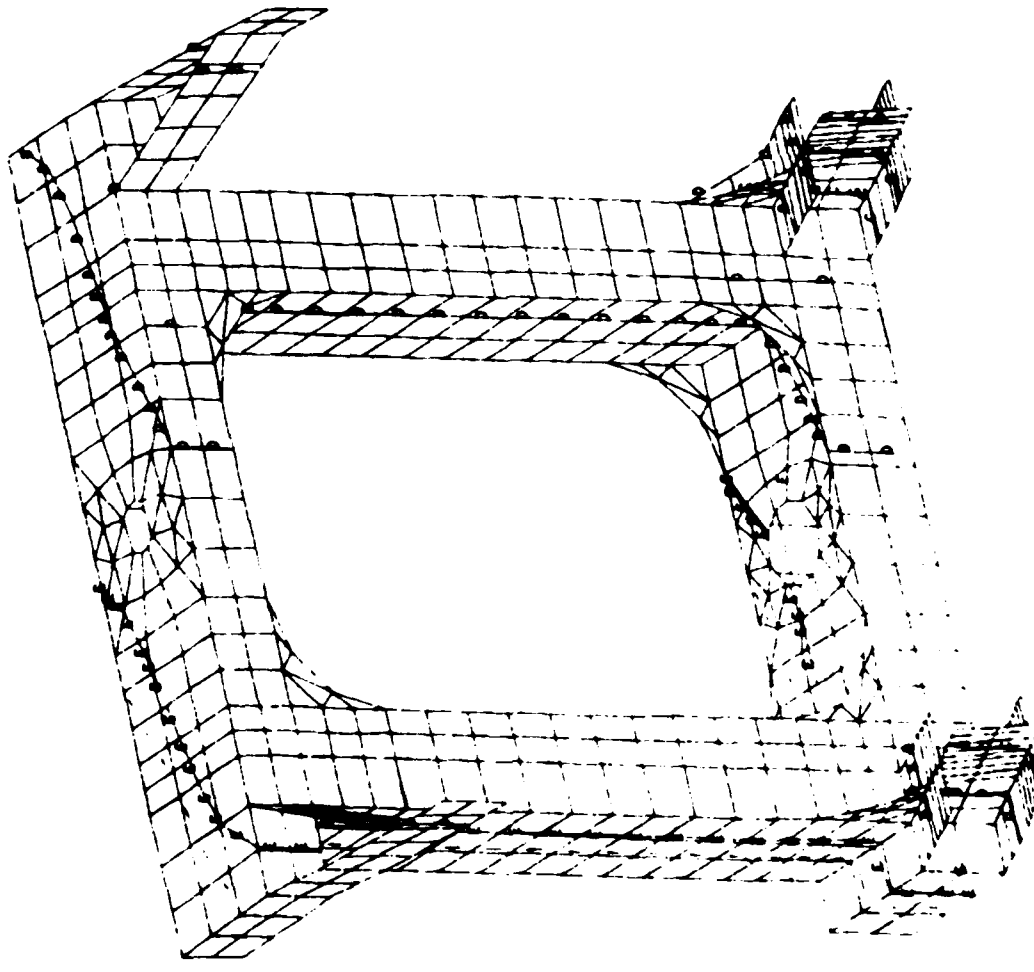
POST1 -IMP-

22.5-72 CASE, LUMB, STAYC, COMB CASE

```

ANSYS 4.20
PCB 11 1007
12:25:13
POST1 STRESS
STEP=1
ITER=1
UY
DISPL NODAL

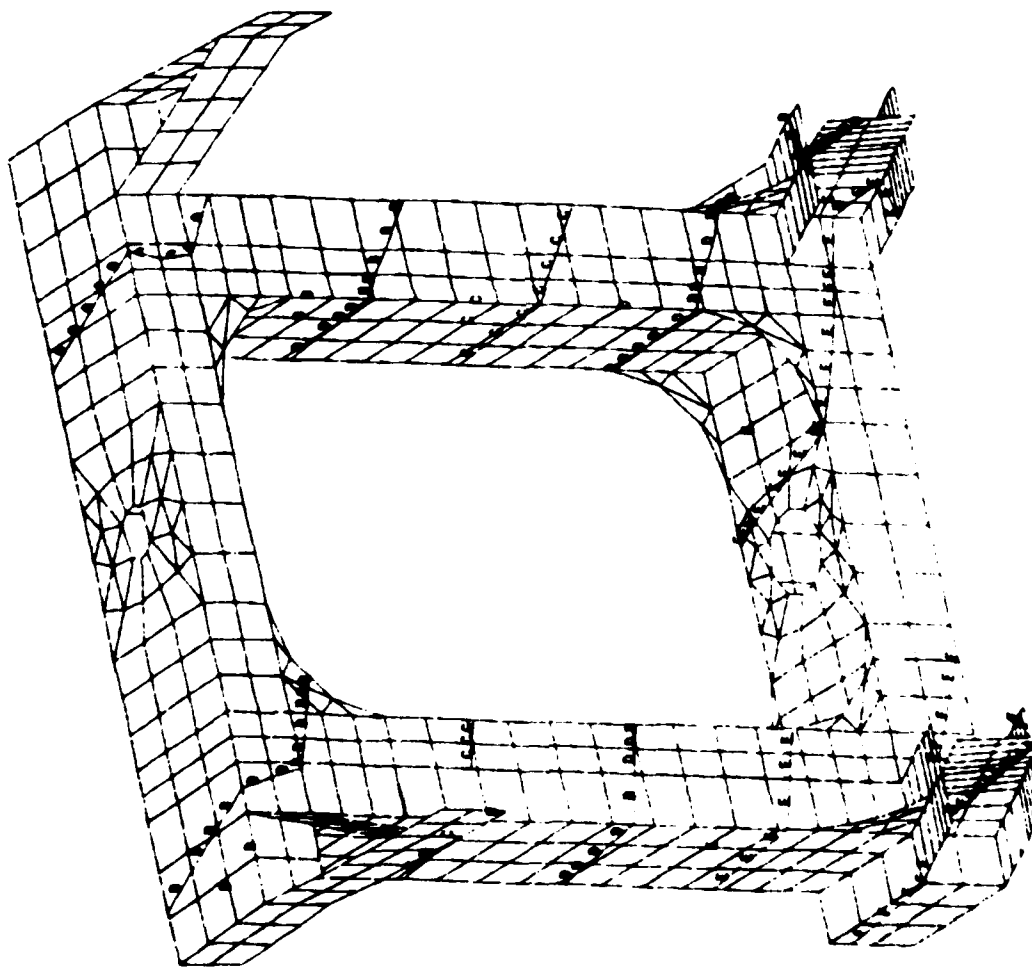
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POST1 -IMP.

22 5-72 CASE, LUND, STATIC LOAD CASE

ANSYS 4.20
 FEB 11 1987
 12:27:24
 POST1 STRESS
 STRESS
 1788.1
 1788.1
 UZ
 DISPL MODAL
 MU=1
 VU=1
 ZU=1
 0157-29.7
 MF-53.9
 VF-35
 ZF-7.93
 MIDDEM
 MM-0.178
 MM-3.11
 A-8.82
 B-8.32
 C-1.82
 D-1.32
 E-0.15
 F-0.315



POST1 -IMP.

28.5-72 CASE, LUMB, STATIC LONG CASE

END

10-87

DTIC